

In this Issue

FLYING BOMBERS ACROSS THE ATLANTIC

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AVIATION

The Oldest American Aeronautical Magazine



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Deliveries

On their delivery flights from San Diego to England, Consolidated Liberators cover 6,000 miles of land and open ocean. The speed and ease with which they are completing these flights demonstrate once again American leadership in long-range bombardment aircraft. Each Liberator gets its power from four dependable Pratt & Whitney Twin Wasp engines.

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From a scant 43,000 men in the summer of 1932, it now employs 122,000 men and in a few months will employ 207,000 men. From 13,700,000 square feet it is expanding to 20,600,000 square feet—an increase of 209%. That is a measure of the job being done.

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is America—building the job—and getting it done.

These pages tell the story of Martin's part of the job—a part that will make Martin the largest aircraft manufacturer in America. From 1,263,000 square feet, Martin will increase to 3,717,163 square feet at Middle River—and other plants operated by the Company will bring the grand total to 5,102,725 square feet—41% plant increase, operating twenty-four hours per day for the National Defense. That is the job being done quickly and efficiently at Martin's.

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Martin

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"Somebody said That's Couldn't Be Done!"

As Dave C. Davis wrote:

Somebody said that it couldn't be done,
But he with a double-edged
That thought it couldn't, but he would
do it.

Who wouldn't say so if he'd tried.
He's been fighting with the force of a pin
On the line. If he wanted to build it,
He started in and he built the thing
That couldn't be done, and he did it.



How Big Is 5,102,725 Square Feet? This floor space of these Martin Plants, if a one-story building, would cover 100 acres in New York City extending north and south from 234th to 236th Street, and east and west from 74th Avenue to Park Avenue. That's over 100 acres!



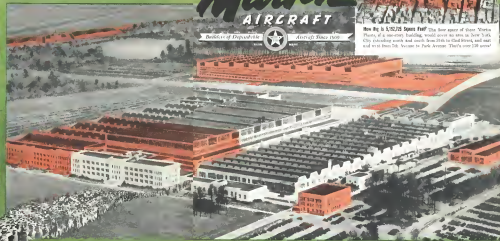
New Army Building—1,100,000 square feet of floor space in production of war materiel. A four-story building, this new Army plant is located in building, approximately 100' long, 100' wide and 10' high, the "Pond" in the "Pond" Airport, where the new Army building is being constructed. 1,100 feet in construction, the higher speeds and greater wingspan of modern bombers, which bring them down daily.



New Coast Plant—Expansion of existing plant, which will be completed by December. The new plant is located in building, approximately 100' long, 100' wide and 10' high, the "Pond" in the "Pond" Airport, where the new Army building is being constructed. 1,100 feet in construction, the higher speeds and greater wingspan of modern bombers, which bring them down daily.



12,000 Workers will be at work at the Martin Plant at Baltimore when the new plant is completed. In the photograph are Engineering Building, Administration Building, Personnel and Office Buildings, Machine Building, with addition completed, old and new Army Building, Shop and Drop Hammer Building, 500-acre Martin Airport and Hanger. Total floor space, 5,102,725 square feet.



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Martin Plant, more than doubled in seven months. That's a new addition since last September—expansion of the building is complete. In the photograph are Engineering Building, Administration Building, Personnel and Office Buildings, Machine Building, with addition completed, old and new Army Building, Shop and Drop Hammer Building, 500-acre Martin Airport and Hanger. Total floor space, 5,102,725 square feet.

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...the high standards the school has attained in the aircraft industry.

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The Martin "Flying Torpedo" designed for carrying heavy
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Fitted with four-bladed Curtiss Electric Propellers and many
other new features of design, the B-26 bombers of the U. S.
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Now BOEING is delivering to Pan American another fleet of these aerial ocean liners, this time to make possible daily-except-Sunday schedules across the Atlantic, and continued regular Clipper service across the Pacific. The new advanced Clippers, largest and most luxurious flying boats in the world, are being delivered considerably ahead of original schedules, while BOEING continues to devote its major effort to the quantity production of 4-engine Flying Fortress at an ever-increasing rate for the U. S. National Defense Program.

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REPUBLIC AVIATION CORPORATION
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Bombers Over the North Atlantic

▶ IT IS IRONICAL that, on the Armistice Day, the first formation of bombers for Britain to be flown over the non-stop North Atlantic since air travel safely, somewhere in the British Isles.

It makes you wonder, a little about what we have been celebrating on Nov. 11 for all these years since this first transfer of equipment but still these operations have become routine. As far as we now know, there have been no losses of planes or personnel on these frankly hazardous flights, without benefit of adequate radio, meteorological or navigational equipment. Some pilots have been lost on the return trip by land. And, contrary to some of the current hard headlines on the subject, these ferry trips have not been interrupted by German fighters. If they were, the unrelentingly North Atlantic weather would probably be depended upon to sweep fog or cloud cover for the bombers. The success of the North Atlantic ferry is a high tribute to American airplane engines, and equipment and to the pilots and crews who fly the ships across. It was even the straight news of one of these was giving most of the editorial facts and some of the fiction on the subject, late in page 30.

▶ AFTER YOU HAVE READ Captain Beulah's interesting account of the greatest ferry job in aviation history, don't get the idea that the Atlantic is as quiet as Wall St. on Sunday afternoon. Remember that one of the great strategic mistakes of the present war was that neither side had provided

with web enough long range bombers. As soon as it was realized that was, might not be merely local officers, everybody started to scramble for airplanes of the flying fortress type. Our Air Service and our manufacturers had known the need for this type and had gambled some private and public money on the idea. The guar-

ty would not be one of the best investments ever made. It is now held by the Germans and the British are in possession of airplanes of this type in reasonable numbers. And by their use they can carry the war deeper and more menacing. They return more RAF activity over Germany and more German activity in the Atlantic Ocean.



IT IS HIGHLY SIGNIFICANT that General H. H. Arnold has been in Europe for several days with the British. As Deputy Chief of Staff of the Army and as Chief of the Air Corps, his studies is of great importance. He landed via PAA's Duke Clipper.

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AIR RACES, stunts, mass parachute jumps—there are thrills galore at the National Air Carnival, Birmingham, Alabama, June 7th and 8th.

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The outstanding performance that has made Texaco preferred in the aviation industry has also made it preferred in the field elsewhere as well.

Texas Aviation Engineers gladly offer to make in the selection of Texaco Aviation Products, available in leading airports in the 48 States. Please check nearest Texaco distributing place, or write:

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★ More turbine unit Texaco Five-Chat Gasoline than any other brand.

★ More revenue airline miles within the U. S. are flown with Texaco than with any other brand.

★ More buses, more bus lines and more bus-miles are lubricated with Texaco than with any other brand.

While recent reports it appears that this firm has developed a new form of scaleless suspension, independent of the long range loader and the tank, 25 new submarines. This will have more from the construction in the near future.

★ **MID-ATLANTIC WARFARE** brings the battlefield a little closer and already we are beginning to hear each other. As maybe Haler will send a new ship out to go through the waters of London New York just in a gesture. Most of this talk comes from Jimmy Long (leader who don't leave the front). Until Harry Haler suggests the Duty, he is usual revenue. The simple facts that the preceding words are exactly. That means that we can always fly toward Europe with confidence. Until Haler suggests the Duty, he is usual revenue. The simple facts that the preceding words are exactly. That means that we can always fly toward Europe with confidence. Until Haler suggests the Duty, he is usual revenue. The simple facts that the preceding words are exactly. That means that we can always fly toward Europe with confidence.

★ **SEVERAL CERFIMONES** made airplane production industries. Since the delivery of the first No. 1000 in the Army and No. 1001 in the Navy in mid-March, DeLorean comments were held. Age 7 for North American's \$1,000,000, later at 1,000,000 sq ft outside Dallas (See page 21). Wright Associates, which is selling the land against the light alloy engine, has taken with a local new program, finally at Fairfax, N. J., having 1

★ **BUT DON'T LET IT LULL YOU** into a state of mind where you dream of security in which we Americans have to admit. It will always be an advantage for us to be in airplanes from in and out of the country, the landing will be to Dallas. Already we are building planes capable of flying the Atlantic non-stop with substantial loads of passengers and mail. Because of the great size we must always be planning in the development of flying equipment of great range. European designers are not sleeping meanwhile.

★ **WE CAN'T LEAVE THE WAR** without a couple of observations on the current situation. In recent years there was the highly effective use of aerial planes in the recent British naval victory in the Mediterranean. This type of airplane seems definitely to have proved on North America in the fact that the enormous British-Panzer team has demonstrated that it can work effectively in sea-land country. Apparently serious conditions to serious trouble.

★ **IT'S CHERRY BLOSSOM TIME** in Washington but the government offices of the Army and Navy and the boys at the Armed Division in OCS haven't had much time to see the blossoms. All who have in their office in the fact that the government, more square feet, and more horsepower, and how they are going to put more of all at home. The (they are just beginning to try some of the new) for their kind and sometimes discouraging work. The record is in terms of total monthly

production figures. This brought some figures for airplanes with over the 1,000 per month now. The Army Navy-Brick (and) the Navy in mid-January production figures at the time. Total production for all military and heavy commercial types was 1,256

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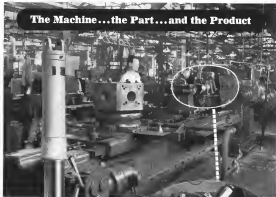
reporting at 20,000 sq ft, a change a day. First & Whitney paid two hundred out of the way at Rembrandt Field to make room for 200,000 sq ft of additional floor space bringing the total beyond 2,000,000 sq ft. Appleby would up the price for 17-43 and was with the same as a larger order for P-48 Interceptors. Volvo put into action an additional 750,000 sq ft at Nashville, Tenn. local 100,000 sq ft more at Buffalo, New York doubled its space by a total of 500,000 sq ft at Hawthorne, Calif.

★ **WEEKEND BLACKOUTS** in air defense plants are being according to O.P.M. Sunday 10 hours who reports that 88 percent of the production workers in the aviation industry are now working six days a week. 12 percent are working five days, and two plants employing a total of 48 aircraft workers are now working seven days a week.



"Wagon wheels his illustrated, around the holy country, will keep the entire history busy after the war ends."

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Designing and building the design of the Bendix Shock Strut for a new Light Airplane. The material of this strut is 1317 metal plate. The plate is 1/2 inch thick and 1/2 inch wide.



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The tremendous size of parts, and the accuracy and strength factors which must be maintained, have made necessary the use of very large and versatile machine tool equipment of the most modern type.

Bendix Pneumatic Shock Struts are in each instance, the carefully calculated answer to a set of conditions — for every airplane model creates its own distinct factors of weight, speed, impact shock — and flexible engineering trends ready to meet them.

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AIRPLANE WHEELS AND BRAKES • PNEUMATIC SHOCK STRUTS
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Side Slips

By Robert Coburn



Screenings all over the world seem to have made considerable progress in recent years in the development of synthetic and plastic materials. Natural combustion engines had been corn, leather, dresses from wool and those other garments were by the far too made out of silk, cotton, rubber from their own estate, and airplane body parts from breakfast food. It's going to be a disaster to build a chicken house out of wood, milk and roofing paper, if some plastics can be found to fill the next niche.

Not that we're ungrateful for the great work of those scientists and research laboratories. y'know. In fact we'd like to suggest to these most valuable research projects, as soon as they get around to it, as attempt to make ships, airplanes, tanks and guns

Steadily this action will greatly concern those who are really interested in the National Defense Program, as we suggest the appointment of a Defender of the Defense Program against

work like can operators and factory engineers who clearly resemble army engineers. In any case, however, results were accomplished; so mean where they have been constant.

Sooner, we think it was a fellow named Alvin, once wrote a little book about the box. It seems that after long years of war some supply young had decided there must be something better than hexagonal was cells for a hexagonal box. They tried squares, octagons, round, octagonal and many other shapes without success. Finally, after several generations of long hard experiments, they were very much up against it until some bright young boy hit on the idea of using hexagonal cells. These worked very well and he was hailed as an enormous technical genius, medals and honorary degrees.

All of which leads up to the fact that we're answering that the question is: are you a man or a woman? We've gone back to what controls in the cockpit, airplane law, air law, take-offs, air-craft, landing gear, square wing tips, and finally we shall soon see airplanes with twin pusher propellers driven by bicycle chains. Just as we won't be overbaking anything, we hope the NACA has got to be going forward the Wright Brothers' winging wings, forward control surfaces, and the pilot lying flat on the lower wing of a bi-plane. However, even if it were a big improvement we'd be going better the passengers sit on simple canvas seats along between two planes of bi-plane, as that might cause the elimination of the steering as a member of an aircraft's crew.

the Divisions of War Life. There's a real trick, probably coming at least 20 years from the first gas turbine engine.

It seems that the most of the current war we made some results in this space which we are now very happy to report, with apologies. We will save these eggs that the English women and girls were being trained as skilled mechanics in airplane plants as they could replace the men workers in the event of war. At the time we suggested of English girls were going to feel confident in flying airplanes held together with hair pins and bits of chewing gum.

Now, considering the outstanding fighting and bombing planes being turned out by these same skilled American workers, our reliance has it seriously off to them.

On the other hand, possibly we should make some reservations in the above, not being so full of ourselves of the facts of the case. Maybe some clever English designers worked up a line of planes and motor jets which look like hair pins—wonder what



out of the great over-supply of "bottle necks" in the country seem to have

So we see by this paper that wild life conservationists have become concerned over the possibility that new bombing and gunnery ranges, and the pollution of streams and coastal waters by sea area activities might be harmful to fish, birds and wild animals. Accordingly Mr. Deane has appointed a wild life expert to defend them against the Defense Program.



NORTH ATLANTIC FERRY

By Capt. V. Edward Smith

It is midnight and I am sitting in my "office" watching on an augeux tube and wondering why it takes the wing so long to come back up after each dip. My First Officer is there at my side, also with a tube in his mouth, but he isn't working. He is just told due to the continued effect of high altitude. My radio operator is in better shape but, since we are not permitted to use our radio equipment except in an extreme emergency, he spends most of his time sleeping and only the darkness

The night is as dark as the inside of your fist. The glow from the running lights only 20 ft. away is barely visible through a gelling rain. We are being bounced around rather heavily by the storm after climbing to 10,000 ft. in an effort to get rid of a load of ice. More than 2 miles below us the North Atlantic is swirling in chills of

up on the shoreward at the bottom of the ice, alongside the flat of victims created in the preceding salvos were. And it is these survivors that a first-time Atlantic ferry flight of bombers for Britain. Chances are you place in a formation of seven, all appeared new, even, and for himself in the storm.

If the above paragraph seems to be a little out of place, it is not. It is the men who made their first night formation flight, and according to "cater" delivery of bombs, it is a far cry from routine commercial flying. The men who made their first night flying a few Massachusetts men, not good, however, as most of the German ships have done. Trust of the owner is that on from Atlantic ferry flight has yet to be made. The men who made their first night flying are not the same as the men who made their first night flying.

are plenty of difficulties to be met on such a flight without worrying about enemy aircraft. Although I made several ferry flights in Great Britain, the possibility of interception was always fact on my list of worries.

Along with labor, there has been friction across by the aerial route, and the stream of fly-away deliveries grows larger each month. Although the actual number of planes flown is fluctuating, it has a small fraction of all of the green-work figures I see published, it is clearly considerable. And to the best of my knowledge every plane has arrived safely at its destination. Many flights of Sembers across the Atlantic, in mid-winter and over the great circle route, have now been demonstrated at a feasible delivery system. It is faster and safer than shipping them by rail, and avoids

thrive more on a diet of shopping for other needed war supplies. The main danger was part of any ferry week was always the return to Canada by boat. This danger and delay is soon to be eliminated, I understand, through operation of a flying boat service which will carry ferry plane pilots and passengers from the British Isles to Canada.

Of course the Atlantic had been down many times before the lumber terry system was started. The last of such nights had proven so long people were present was that some people had been sleeping in the "back porch" department. George and the North American continued. The trip had been made by all sorts of people in all kinds of responses. Nevertheless, the Atlantic is still a formidable "back porch," especially for those who are not used to the place, struggling by dead reckoning, without benefit of radio, and under wartime restrictions. But it is what we have to try to make that is the general neighborhood of Greenland and Iceland, at night, and with very little light, except reporting. Until last August, Dan, and the other men, and some Lockheed Hudsons arrived at Great Britain, the unsuitability of



Coryell et al. • RasY

Capt. V. Edward Smith is typical of such airline pilots who have chosen to serve in the military. He has completed 22 years of commercial flying without accident.

He says, "I've been a part of the war effort in the work of the military. I've been a part of the war effort in the work of the military. I've been a part of the war effort in the work of the military."

After eight years of service, including a combat tour in Vietnam, Smith is now a civilian pilot. He is now a civilian pilot. He is now a civilian pilot.

Smith has made a career of it. He is now a civilian pilot. He is now a civilian pilot. He is now a civilian pilot.

RENTON, WASH.

winter delivery flights to European harbors across the Atlantic will open its question. Repeated successful delivery flights during the winter months have proved the feasibility of the operation.

Further information on the Southern ferry operations has been unavailable to date due to war-time secrecy in which the crews are pledged. Most of the accounts of such flights have been national or highly imaginative. Since I am no longer connected with the ferry service, having retired in 1945, I am unable to give a reliable or verifiable account to give a factual report on the ferry service. This has been classified in such a way that no military information is revealed, even now. Therefore the reader will understand why no description is given of the airport or the flight itself. The ferry pilots are being delivered, on the route by which it was constructed.

When the rail went out last last summer for American pilots, to ferry US bombers to Brazil, I had just arrived back in this country from a period of airline service in China. After what I had been through in the Orient the North Atlantic scheme sounded

(There is more left)



READY-MADE AIRPORTS ONE ANSWER TO CONGESTION

If your community is short of airports, the author suggests
that you put floats on your plane and fly it off the water

By Captain Robert S. Fogg
Chief, Seaplane Unit, C.A.A.

THE thousands of rivers, lakes and bays in this country offer a natural and logical solution to the airport shortage that faces us today. Already 25 percent of the nation's flying clubs have been closed to private pilots and commercial operators. Each week the Army and Navy take over additional fields for purposes of national defense. In many communities the competition at airports soon to provide there is becoming a real problem. Water airports are one obvious solution.

While there are 2,600 man-made airports in the United States, there are almost countless thousands of bodies of water from which seaplanes and flying boats can be flown. It is amazing that commercial operators and private pilots faced with the dilemma of either giving up flying or facing a new airport, sometimes overlook or ignore the obvious solution in their problem by neglecting their localities with potential. Then they could either nearby lakes, rivers, and so on may make the very necessary Government restrictions instead of some remote, but not ideal, airport. Here is a means of decentralizing flying and reducing the growing congestion at our major airports.

Flying fields are often expensive both at first cost and in maintenance expense. Floating water areas offer an immediate solution for flying facilities in many communities which for financial or topographical reasons might not be able to construct an airport. Many operators who have Civilian Pilot Training stations are being forced to find new bases from which to operate. If they shift to a water base, the added expense of buying floats for their



These studies represent bodies of water where large flying boats could safely land and take off. There are thousands of additional lakes and rivers which small seaplanes might use. Map courtesy of Consolidated Aircraft Corporation.

ships is balanced by the extra \$30 per student paid by the government to non-private operators. In fresh water operations the maintenance of equipment is actually less than on gravel airports. Salt water operation with its associated corrosion difficulties can be combated by proper treatment of the structure and the generous use of a hose following each day operation.

The advantages of student training on seaplanes are at both obvious and unobvious. On a busy airport with its traffic rules and signals, plus the necessity of making a complete circuit of the field for each landing, a student does not get to average four landings and takeoffs per hour. In a seaplane, instructor and student can go to some remote sector, make successive straight-away landings and take-offs entirely free from mental and physical distractions.

Still more important, seaplane flying offers a means of lowering entry to the most pleasant and safer conditions known both to personnel and equipment, as in flying over water there is always an airport under you as well as suitable terrain or forced landings. The center of time, particularly with business and college students, is an important factor. To maintain only one of the many examples used by the writer is the case of a southern university where it is a 30 mile drive by automobile from the campus to the airport and return. Allowing two hours for driving through traffic and another hour upon arrival at the

airport takes quite a visible amount out of an eight-hour day. This particular airport is quite good at its long runways, giving that it is not as great as five, whereas five minutes from the campus is a beautiful protected water area measuring 10 miles by 100 miles, quite an airport if must be admitted, and free for the using.

More and more people are now beginning to realize the great comfort, convenience and safety of water flying

both for business and pleasure. It definitely has an appeal which attracts people who would not otherwise fly. Wisconsin forced landings from motor boats are now advantageously taken, consider the numbering thousands of sail on a flight from New York to Miami, for instance, where the inland waterway provides a continuous landing strip within easy gliding distance all the way. Or again take the 1800 mile flight from the Gulf of Mexico below New Orleans to the Canadian border at Minnesota. You don't usually need any maps as you fly along with constant points of land at 300 feet above the Mississippi River.

With all these apparent advantages then, what is the trouble—why not three more of this type of flying? There are several reasons, the natural and inherent four of the unknown, the lack of an understanding of its basic requirements, such as how to properly handle service and maintain a seaplane at its base, and where to stop for service along the chosen route. Primarily it is the lack of proper facilities that has retarded the growth of water flying, whereas, had a reasonable number of them were more built we would have more seaplanes and if there were more seaplanes we would have more bases. Obviously proper facilities bear a similar relationship to seaplanes as that of good roads to automobiles. It is disconcerting, to say the least, to fly 300 miles and then have to spend several

(This is page 142)



Capt. Bob Fogg, shown here while in a cleaning berth at Bush, Mich., has three of his seaplanes in water along all over the country. With U.S. Government landings of seaplanes camps have been built and improved under the supervision of the C.A.A. Seaplane Unit. Seaplane flying will give a head start away from traffic.

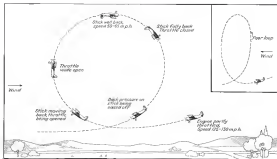


Fig. 1. A periodic loop. In the square is a pure loop. Note the difference in the wind direction.

Acrobatics Are Easy!

Part II

By Daniel J. Brinen, Jr.

This is a continuation of a series of articles which started in the April issue. The author has been a test pilot, commercial operator and instructor since 1913. In addition to having written numerous books on aviation subjects, he is the co-author of CAA Bulletin No. 12, "Civil Training Manual" and Bulletin No. 3, "Flight Instructor's Manual." He is now on active duty as a Lt. Colonel in the Navy.

ALTHOUGH the loop is perhaps the most difficult of all the acrobatic maneuvers it has always been considered by the general public as being thrilling and spectacular. Perhaps the word "acrobatic" should be somewhat qualified, for though little difficulty is perceived in making the ship climb into an inverted position and dive out of it, it is well-nigh impossible to perform such a stunt illustrated in Fig. 1 with the considerable skill and fort of the acrobat.

However, without bowing or parades, the nasal cord of a leopards in practicing the loop was to open wide the throat of the dear old GIN, did until the entrapment scene was occurring in preter brain has lost, nasal loop on the stick and loop for the best. Sometimes he made it around the loop, but he was not able to get out. The stick was pulled on his back and well-loved out of it. Here so, a good pilot could perform a pretty da loop with the under-powered and poorly-sown variable JN. In fact, some were capable of looping these ships with the engine still thrashed. Though, once with full power, considerable area was always. Many of the modern sweet this will gain several hundred feet of altitude per loop.

In the type of training ship about which the movie is written, it is desirable to dive until a speed of 125-150 mph has been reached. Some types of the monoplane class, such as the low-wing Panchard trainer, require somewhat lower speeds—140 to 145 mph. This dive is made with the throttle partly closed for two reasons. First, with throttle open the prop would

go up well past the desirable limit. Second, there would be no reserve power at the point where it is needed most. Only one control, the elevator, is used in the actual performance of the loop. The rudder and ailerons cease to play only to keep the wings level and to increase direction.

The poultry is a chicken. Before a head-on, preferably a road which dips directly into the wind, point the ship along it and raise the stick about 10 degrees. The wings will be blown to the throat so as to keep the r/y in below the previsible maximum level horizontal flight. (As a matter of fact, if the cruising r/y is not too exceeded, a steady satisfactory r/y may be maintained for 100 ft. The pilot should rather the figure mentioned above, check to see that the wings are level and that the down-stroke is light as aeratic along the road. Being the head-on, the r/y is not too exceeded. Also, be sure that it is brought straight back. Since the force required is appreciable, there is a tendency to pull the stick slightly toward the right and the r/y will be toward the right. This does not lie in the r/y.

As the insect passes the burrow on the way up, gradually open the throttle about, so much a rate that it is fully open when the stop is in an approximately vertical disk. As the pressure on the seat of your pants begins to lessen, come back more and more with the stick. In a correct loop there is no tendency to hang on the left. Do not make the mistake, however, of pulling the stick all the way back before the peak of the loop has been reached. There should be at least several inches of movement left for use on the way down.

As the ship approaches the vertical slink, you will instantly lose all sight of the ground. Hereafter you have been able to see the ground and the horizon throughout every maneuver and when it disappears and you have nothing but the sky before you, you will have a rather hot feeling. However if you have followed instructions carefully, kept the adrenalin in control and the rudder likewise, you will suddenly see the horizon again appearing over the top wing and parallel to it.

As the nose passes the horizon on the way down, ease the throttle closed and gradually pull the stick the rest of the way back. As you begin to approach level flight once more, you should feel a severe bump. This is your own airplane and if you do not feel it your loop has not been true. You should finish pointing directly along the cord in your several descents, and

raise the stick ahead to neutral as level flight is resumed, opening the throttle to the cruising position.

Now, let's consider some of the things you might have done wrong. If your wings are not parallel to the horizon when you first see it, you have to do the click move to one side or the other and you should make the move

remains substantially the same you are advised this may be confusing, but simply remember that the controls function, as far as you are concerned, just as they always did. If your best way to return the horizon to the right, the stick should be moved to the left to correct the condition. Obviously,

1989, in press, 1992).

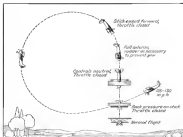
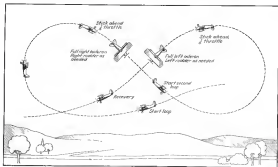
Fig. 9. *B.* Temp. with an integrated salt recovery

Fig. 3 The Culture shift



North American Aviation's new plant in Dallas

NORTH AMERICAN Dedicates Dallas Plant

New factory completed in record time.

DRAMATIC proof that the American aircraft industry can do its job at peacetime weapons for our national defense needs was given by North American Aviation, Inc., at Texas, with dedication Apr. 7 of the new 400,000 square foot factory at Hamley Field, Dallas. Representatives met in an attempt to describe the plant, one of the first of our expanded national defense program which is to go into operation.

The new plant is now the world's largest completely air-conditioned and completely lighted building. The most impressive demonstration was that, on the very day the plant was dedicated the first of the AT-6A advanced trainers were already completed and rolled out for delivery. And more than 1,800 employees were at work in the factory on the day of its formal opening, just a little more than four months after the first construction was started Nov. 25, 1948. This staff will be increased rapidly to a peak of approximately 12,000 making possible a production rate considerably in excess of two trainers a day. AT-6A for the Army, and SNJ-3 for the Navy.

The amazing story of how North American has been able to start delivery

of finished military airplanes from a spot which was virgin Texas ground just ten months before, goes back to the parent company in California, North American Aviation, Inc., and its production-minded president, J. H. "Jack" Knudsenberger.

Modern "Knack" Knudsenberger, a graduate of the Glenn L. Marcus and Douglas Aircraft Co. organizations, is always the first to discuss any pro-

ductivity in the production field. Yet Knudsenberger cannot escape the hands which have been banded on him by aviation experts. And, as "Knack" would be the first to point out, that reputation has been established by the able corps of assistants with which he is surrounded. For Knudsenberger's greatest genius is his ability to obtain men who have the "know how" of production. So many such men have con-

tributed to the phenomenal rise of North American Aviation that it is not possible to credit them all.

The parent plant of North American Aviation is just over five years old. Operations commenced at the lightwood plant on Jan. 14, 1946, with a staff of about 300 employees, less than one third the number employed in the Texas plant at its dedication. So rapidly did the production curve develop at the California plant that total personnel has expanded more than 3,000 percent in the two-year period. And this factory was proving its production curve in 1940, before the fall of France and subsequent pressure for quality output, by taking as many as 100 planes per day off the assembly line. Approximately 3,000 planes have now been built in the California plant, and operations have expanded to cover the field of trainers, gunners, single-engine light bombers and two-engine medium bombers, all of which are in production at the present time.

Out of the lightwood organization has come the nucleus of trained personnel for the Dallas plant. About 30 top executives, department heads, and supervisors, were sent to Dallas from California, and around these men the staff was built. A three-way job was done from the start, with prompt establishment of adjustment, procedure, conduct of an extensive training school in developing qualified workers and lead men, and immediate initiation of maintenance operations. All this was done in rented quarters while the factory was being erected, which explains how it is possible to deliver finished airplanes on the day the plant was dedicated. Equipment, parts, fixtures and supplies were shipped to Texas from lightwood, in order to get the new organization started.

The widespread recognition of the Dallas plant as the parent company's constant reminder of its steady production shift, rather than a complete factory. Most administrative, engineering, and tooling work is done in lightwood and left in Dallas. This provision will be continued as a matter of policy, leaving to Dallas the new job of building airplanes for the work provided. This new general procedure will be followed with the new Kansas Dry bomber assembly plant, now under construction. And the scheme can be further expanded if Dallas needs so require. It is this ability of our airframe plants to break and rebuild, with almost invisible repairs and efficiency which provides America with its standard defense asset.

In general layout and arrangement the Dallas plant approaches the ideal of production efficiency. The building is approximately square, all with being of one-story construction with light

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Pres. North American



J. A. Stuart
V. P. and Gen'l. Mgr.



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Director



David B. Smith
Director of the Plant



C. H. Davis
Director of the Plant



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Pres. North American



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V. P. and Gen'l. Mgr.



R. A. Lenth
Director



David B. Smith
Director of the Plant



Many visitors, including President Knudsenberger, gathered for the dedication of the Dallas plant. Mr. Knudsenberger was given a standing ovation as he was introduced.

exceptions. General arrangement is such as to provide a clear systematic flow of all parts and materials from the time they enter into the receiving department until the finished plane is wheeled out for its flight use. Storing and storage departments control completely every side of the building, while final assembly extends completely across the opposite side. In between are four lanes across the building from stockroom to final assembly of all such units such as fuselage wing, engine, etc. Progressive manufacturing operations, and subsequently work of like piece as early as possible in a series of straight-line movements across the plant. Arrangement of the plant and its equipment to make this possible is greatly

defeated by the fact that the building is completely air conditioned and artificially lighted. This makes it possible to place heat treat, chemical bath, grinding, and other operations which are normally segregated due to fumes, heat, etc., at their logical position with relation to production flow.

Provision of artificial fluorescent lighting, and air conditioning, also promotes efficiency in other ways. Reduction of regular lighting and improvement in vision is said to give increase in individual output up to 30 percent. And industrial savings result from elimination of winter heat loss through windows and ventilators.

Location of the plant at Dallas has strategic and economic benefits. The

(Times page 10)

Plastics in Aircraft

This article, and the two that immediately follow, provide a basic understanding of the materials and uses of plastics in aviation today

By J. Earl Simonds, *Plastics Consultant*

IN the field of aircraft design the term "weight penalty" may be heard repeatedly. An additional weight appears simply justified because of the addition of materials for aircraft due to weight of the finished part is usually of primary consideration. This holds true regardless of the nature of the part.

Engineers have shown designers how to balance properly the weight of finished parts with the safety factors adopted for both transport and combat aircraft. The proper balance between weight and evaluation of the properties of the materials, and the relation between certain of the properties and the known or estimated stresses to which the part may be exposed under variable service conditions.

The property of specific gravity or density of aircraft materials, while of great importance, is not the sole factor considered in choosing materials. For structural members a favorable ratio between specific gravity and certain other properties is more than the governing factor. This is best illustrated by the very extensive post-war usage of duralumin rather than stainless steel in the all metal plane design. Stainless steel is much stronger and heavier than duralumin, yet the ratio between certain physical properties and actual conditions of service favors duralumin and accounts for its widespread use.

At least two reasons exist for this. (1) (2), who have contributed much valuable data on plastics in aircraft appear to be in substantial agreement that the specific gravity and elastic modulus of plastics, and the bulk or thickness of the structural member, are in greater importance and that the strength of materials is of secondary importance.

Plastic materials are inherently light in weight, the specific gravity of the most extensively used types falling within the range of 1.05 to 1.30. While this property is always attractive to aircraft designers, often solvent properties peculiar to certain types of plastic materials have influenced the choice of

materials used, as will be apparent in the following discussion of several widely varying applications of plastics in aircraft.

Aircraft Propellers

With the trend toward more powerful engines, the weight of aircraft propellers has become a relatively real problem, according to data supplied by Mr. Fred E. Wink, chief engineer of the Engineering & Research Corp. (3). The difficulty, which is well understood by designers, is several points listed at the end of this article (4), (5), (6), (7), (8), (9), (10), (11), (12), (13), (14), (15), (16), (17), (18), (19), (20), (21), (22), (23), (24), (25), (26), (27), (28), (29), (30), (31), (32), (33), (34), (35), (36), (37), (38), (39), (40), (41), (42), (43), (44), (45), (46), (47), (48), (49), (50), (51), (52), (53), (54), (55), (56), (57), (58), (59), (60), (61), (62), (63), (64), (65), (66), (67), (68), (69), (70), (71), (72), (73), (74), (75), (76), (77), (78), (79), (80), (81), (82), (83), (84), (85), (86), (87), (88), (89), (90), (91), (92), (93), (94), (95), (96), (97), (98), (99), (100), (101), (102), (103), (104), (105), (106), (107), (108), (109), (110), (111), (112), (113), (114), (115), (116), (117), (118), (119), (120), (121), (122), (123), (124), (125), (126), (127), (128), (129), (130), (131), (132), (133), (134), (135), (136), (137), (138), (139), (140), (141), (142), (143), (144), (145), 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With the trend toward more powerful engines, the weight of aircraft propellers has become a relatively real problem, according to data supplied by Mr. Fred E. Wink, chief engineer of the Engineering & Research Corp. (3). The difficulty, which is well understood by designers, is several points listed at the end of this article (4), (5), (6), (7), (8), (9), (10), (11), (12), (13), (14), (15), (16), (17), (18), (19), (20), (21), (22), (23), (24), (25), (26), (27), (28), (29), (30), (31), (32), (33), (34), (35), (36), (37), (38), (39), (40), (41), (42), (43), (44), (45), (46), (47), (48), (49), (50), (51), (52), (53), (54), (55), (56), (57), (58), (59), (60), (61), (62), (63), (64), (65), (66), (67), (68), (69), (70), (71), (72), (73), (74), (75), (76), (77), (78), (79), (80), (81), (82), (83), (84), (85), (86), (87), (88), (89), (90), (91), (92), (93), (94), (95), (96), (97), (98), (99), (100), (101), (102), (103), (104), (105), (106), (107), (108), (109), (110), (111), (112), (113), (114), (115), (116), (117), (118), (119), (120), (121), (122), (123), (124), (125), (126), (127), (128), (129), (130), (131), (132), (133), (134), (135), (136), (137), (138), (139), (140), (141), (142), (143), (144), (145), 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Although the greatest advantage of these composite wood and plastic blades is probably their light weight, usual duralumin blades ordinarily being about 80 percent heavier, they have other attractive features also, such as their tendency to damp out vibrations, freedom from fatigue difficulties, and the ease and low cost with which new or experimental designs can be con-

structed. The ease of construction of experimental models because of increasing interest in the propeller area is becoming apparent.

The Schwann type of blade requires assembly of a main core of laminated spruce or other light weight wood, which merges into a rest or impregnated and compressed wood (the rest is called "coupling"). The coupling is threaded and screwed into a steel ferrule which supports the blade in the hub. The remainder of the blade is covered completely with a heavy coating of cross-hatched cellulose-acetate plastic sheet and the leading edge is covered with a lead strip of metal. The plastic covering thus protects the wood core against climatic changes and warping, as well as against contact with aircraft particles rain and sea spray.

The coupling is made up of sheets of 4 in. thick laminated spruce impregnated with a phenol-formaldehyde resin. A stack of these impregnated sheets is then subjected to heat and high pressure where it is reduced to one-half the original thickness and the grain forced to a hard and inflexible condition. Thus the finished product becomes a composite wood with the wood fibers impregnated and compressed in proportion to the actual stress. The actual weight of the coupling is about double that of the original hard wood, the specific gravity being about 1.30.

Sheets of laminated coupling are applied and glued to spruce boards to form the blade block shown in Fig. 2. The blade block is then composed of laminated spruce for the greater part of its length, laminated coupling at the rest end and at the leading edge where the spruce gradually merges into the coupling. After machining the coupling end, the blade is cut to form as shown in left-hand view of Fig. 1.

The blade on the right-hand is finished. The cutaway view in Fig. 3 shows the details of the complete blade assembly.

Before leaving the subject of propellers it seems appropriate to mention a new plastic application which may be

(Turn to page 120)



Fig. 1. Two views of a propeller blade. The blade on the left has just been cut in form. All the right is the finished blade.



Fig. 2. The laminated spruce sheets, lead strips and the plastic covering are applied and glued to spruce boards to form the blade block shown in Fig. 2.

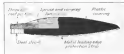


Fig. 2. A cutaway view of the blade block showing how the spruce coupling and the steel ferrule fit together. Left.



Fig. 4. A detailed view of a propeller assembly showing the plastic blade attached to the hub. Labels include: 'Plastic blade', 'Hub', 'Coupling', 'Spruce', and 'Lead strip'.



Fig. 5.

Molding Plastics

A discussion of molding methods being used in airplanes and accessories

By Robert Beest, President, Associated Aircraft Research Co.

Fig. 5. The Dacronoid fuselage is sealed with an metal reinforcement of one block. This section is sealed in two before fuselage head is ready.



Fig. 6. The Dacronoid head and pressure tank for molding into fused aircraft structure. This pressure tank is in the process of being installed in the Grand Rapids MCB, plant of the Borden Corp.



Fig. 7. The fuselage of the Titan trainer is made in two halves which are then pressed together in a pressure mold and sealed in the area shown above. On the left is a view which will be sealed at the same time



Fig. 8. Molded parts at the Borden airplane are placed in this pressure chamber. This shows the chamber and automatic temperature and pressure recording apparatus. Equipment is installed at the Borden Aircraft Corp., Ft.



Fig. 9. When the Dacronoid comes out of the mold it is so smooth as a white top. Former then and after are attached to one process. The Dacronoid is made by the "Vitol" process which is a method of molding under heat pressure and heat of being impregnated sheets of resin which have been placed on reinforced and sheet wood forms.

The photographs on the two previous pages will be of interest to readers of this article.

THE family of plastics is apparently unlimited in its size, with early day bringing new members within its fold. When one stops to consider that such materials as rayon and neoprene, although coming under a different branch, are still members of the same family line, an impression can be seen of how large the group is, even outside of the well-known plastics such as bakelite and melbaform. Then if one of these are reinforced with a cellulose fiber such as wood, paper, cloth, or with metal or cork, you obtain as variety of reinforced plastics, each one with a different characteristic and strength.

Of all these possible materials, only a few have been chosen for use in aircraft structures. Many of the plastics had to be discarded because of their poor resistance to humidity, heat, or work, or for their decreased changes in aging. However, such plastics as methyl methacrylate, acetate, butyrate, or polystyrene, though not used today, might be used as the future because some new addition to them may be found to give them greater resistance. As for the fiber used, nearly everything seems to have been tried. The great difficulty is to get either a thorough impregnation of the product into the fiber or a good bond between the fiber and plastic or between the fiber and more fiber. Metal would have been a very good filler, and was even used as far as Hearn's propeller blade is concerned with certain reinforcement on the leading edge, but the great difficulty is to get a thorough and comfortable bond between the metal and the plastic. Linen, silk, or canvas, though used in industry are not employed very much in aviation because they have no structural rigidity, and for molding they also require a good mold, or at least a steel mold, which is heavy and very expensive. In Europe these materials are used as a filler and with them complete parts for airplanes have been molded that have a tensile strength rated around 45,000 lb. per sq. in.

These uses have been discarded for

the propeller, at least temporarily, because of a cracking that takes place of one section of the blade in relation to another. Of course this might be overcome any time, with progress developing as rapidly as that has.

Then comes wood, which thus far is the plastic most employed. There are a good many reasons for the use of wood. It can be worked easily and with fairly cheap tools. Also wood is a cheap material, its supply is still ample and wood workers are very plentiful. The structural properties of wood cover a wide range, starting with balsa having a density of about 4 lb. per cu. ft. and going to lignum-vita with a density up to 60 lb. per cu. ft. Above all, wood can be fastened easily, not only by nails or screws, but also by adhesives. Wood can be given a smooth finish and by impregnation, and sometimes compression, it can be given properties that make it capable of competing with metal. Wood is very strong used in solid form because its strength depends on the direction of the grain. Also it is very difficult to get a piece of wood of uniform grain as it changes under conditions of moisture. It is for these reasons that plywood construction is usually favored. Plywood maintains the consistency of wood in shrink and swell, especially across the grain, but, as the shrinkage is not completely eliminated, all kinds of experiments are utilized to get a constant material.

Sheet-Ply Veneer Construction

To understand how this resin is applied, let us visualize the construction of a 3-ply veneer of 1/8 in. layer, grain lengthwise, the second at 90 deg. and the third parallel to the first. Between each of these veneers, either a film of resin named Tego, or a solution of water soluble or alcohol soluble plastic, or even urea or vinyl phenol fusible hydro, is applied. The whole is placed under a hydraulic press under a pressure of 300 to 350 lb. per sq. in., and heat up to 300 deg. If it is applied, the resin changes the structural formation of the resin to "polymerization," making the molecules of resin combine together to give a product having a molecular weight which may be several times that

of the parent substance and which gives an entirely different property to the final resulting product.

Resin in Common Use

The three most popular resins now used because of their characteristic ease of workability and strength of change, are phenol, urea, and vinyl. Phenol, hydro, and vinyl. While phenol is easily used for an impregnation characteristic and its resistance to moisture, urea is now a great competitor because of its lower heat resistant in hot-water is consumed. Vinyl which has just been introduced, is not yet widely used.

Any of these resins are available in water or alcohol soluble powder or in water solution as film, and they can be used either wet or dry. Though the setting is always better with heat, they all can be changed with the addition of a hardener now nearly sold or even completely cold setting; the only drawback being that of a hardener is added the resin sets after a few hours contact with the air, leaving very little time for any assembly. For this reason most manufacturers use a water base kind of heating from 140 deg. up to 200 deg. If with little or pressure heat. All these resins require some kind of filler which increases their strength and their strength in the same way. When phenol is used as a bond for a panel, the panel immediately becomes much more water resistant, it can be boiled or dried, soaked in water for years, or buried in the sea. The wood will give before the heat. Urea will withstand all but the boiling test.

Vinyl, though now very much improved, especially under water or outside is considered, is excessively used in safety glass and is only now finding its way into aviation use.

There are several good reasons for the use of the resin film. It is at least four or five times faster, very little as no moisture is added in the press, and a thorough control of the moisture content can be made when curing. Also when the panel comes out of the mold it can be used immediately with no air or kids drying.

You may ask why hydrolysis are not made of bakelite material film that need (Time in page 124)



PLEXIGLAS

PLEXIGLAS is one of the most useful plastics being used in the aviation industry today. Because of the high speeds of our fighting airplanes and the great stresses to which they are subjected, transparent covering is necessary for cockpits, and for the various gunners' and bombardiers' stations throughout the ship. In the last war, wooden panes and the vacuum-process of bombards' glass in the open. Now the use of the cockpit crew must be isolated from the elements and yet they must have large, transparent eyes through which to see the enemy.

In the field of synthetic resins, the acrylics, in which Plexiglas belongs, are outstanding in colossal transparency, stability against weathering and aging, and resistance against chemicals. These resins are derived by chemical synthesis from such raw materials as coal, petroleum, air and water. In conversion, they range from soft, tacky resin liquids to hard, tough thermoplastic solids. Variations are produced by controlling manufacturing conditions rather than by adding plasticizers. No fillers are added to the acrylic plastics.

In the Martin B-26, Plexiglas is also used for the tail gunner's windows, rear gunner's turret, and again, hatch and for the windows. From the rear position the tail gunner can view multiple machine guns in a wide angle.

Left: The case of a Martin B-26 bomber, showing the completed Plexiglas section, the forming of which is pictured and described on these two pages.

John A. Ford



Sheets of Plexiglas are heated in an oven to 250 deg. F. Then the pliable sheets are covered and draped over a deflavored form. When it starts to harden, the vacuum of the form. Because it is then free held the weight of glass and yet is strong and permanently transparent. Plexiglas is used for gunners' observation hatches, side windows and cockpit enclosures in all types of fighting planes. Photographs by Eikon & Eikon



The next step is to clamp the Plexiglas firmly to the form. The wood clamping frame runs entirely around the form. While this tension is applied, the same crew of workers move on to another form to repeat the process. A great deal of care is used to make sure that no specks of dirt get into the warm plastic, or that the bottom does not prevent, or other would affect the accuracy of section against or otherwise.

EYES FOR OUR BOMBERS



Workers in the Eikon & Eikon factory are installing the window of Plexiglas sheets that are to be cut. The work is cutting around a wood and metal pattern. The sheet will then go to the power saw for cutting to exact size.



Before all this work can be done, the necessary specifications are finished. The two halves are mounted together along the heavy line shown in this photograph. As acrylic resin forms the bond. Edges of the Plexiglas are easily drilled so that fasteners can be added to both the inner openings and to the outer periphery by which the unit is attached to the fuselage.



This is the same sheet that is shown in the two lower pictures on the opposite page. The original sheet used to be removed and this is done with a hand-sawing hand saw. Unlike most glass, the product of chemical research, can be scored, drilled and finished with ordinary woodworking tools. It will withstand a heat, does have without breaking but it is not bulletproof. The workers have picked up some reflections due to the nature of the sheet but when drilled and polished. Plexiglas is even more transparent than the best plate glass. While these critical sheets are formed by the manufacturers in his own factory, the Air Corps has found that maintenance and replacement are a relatively easy matter because the material is so easy to handle.



As important steps before that installation is cleaning, polishing and buffing. These workers use steel wool sections with cleaning rags and power buffers. Note that the wooden tables are covered up so that the Plexiglas sections are held firmly in place. Table legs have a circular cap so that men polishing the tables may stand comfortably at their work.

5 GUN TURRET

As defensive armament must be flexible, turret mounting of multiple guns is centering in bomber, patrol and military transport planes. Here a new design five gun battery is described.

By Louis Bruchiss

The multi-gun turret mount has been described, and the theory on which it is based, was developed by the writer a few years ago in Europe, and patented there. Its construction and development were carried through to an adaptation of the variable fire pattern theory to anti-aircraft use, which was approved, after careful consideration, by one of the leading Goodyear armament manufacturers. The aircraft armament picture has changed rapidly in recent months, however, and this multi-gun turret should prove of definite interest to armament designers, inasmuch as it may be expected to be an improvement over existing types of aircraft fire.

ACTUALLY, multi-gun turrets are nothing new. Gun batteries up to the number of four machine guns have been variously mounted in turrets, with the entire battery operated either flexibly, or else (more commonly and decidedly) as an integral part of the turret, with the latter power driven by hydraulic or electric means. But the guns of a multiple mount have the same limit; they all fire at a fixed point, determining inherent dispersion and shooting difference. There is only one advantage in this conservation of fire, and that is in fire power effectiveness. When one bullet might hit, a concentration would enable several to score at approximately the same point, or even enough to possibly do more damage.



The Author

Factors not under his control, but no such advantage. While following an enemy target with his turret gun, the aircraft gunner has no means of knowing at what moment his own pilot will change the course of his airplane, which would upset any aiming and firing he may be doing. The greatest obstacle in aircraft gunnery is the lack of time in which a target may be engaged and held. All other difficulties are minor compared with this one. The time element is so important that it has dictated airplane flexible armament design as an overruling element, and the conditions are so changeable, and requirements have become so exacting, that present day practice may be obsolete tomorrow.

The best proof that most of the highly accurate sighting and aiming instruments available, operating in conjunction with guns in power driven turrets, are hope with the high speed airplane is the fact that turret armament, and turrets only, form the principal source of target and fire control for the aircraft gunner. With airplanes varying in speed from 400 miles per hour in combat maneuvers, and with the gunner's own plane flying at non-parallel speeds in constantly changing positions, the gunner is faced with an increasingly perplexing problem.

Turret mounting of multiple guns is more and more being observed in multi-engine ships at the bomber, patrol and

light to a target. Five power conservation is highly desirable as pilot gunnery, where the pilot can aim his fuselage and wing mounted armament directly at the target by maneuvering his own plane. The aircraft gunner, operating against a maneuvering, or variable target, is at a disadvantage.

Factors not under his control, but no such advantage. While following an enemy target with his turret gun, the aircraft gunner has no means of knowing at what moment his own pilot will change the course of his airplane, which would upset any aiming and firing he may be doing. The greatest obstacle in aircraft gunnery is the lack of time in which a target may be engaged and held. All other difficulties are minor compared with this one. The time element is so important that it has dictated airplane flexible armament design as an overruling element, and the conditions are so changeable, and requirements have become so exacting, that present day practice may be obsolete tomorrow.

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Turret mounting of multiple guns is more and more being observed in multi-engine ships at the bomber, patrol and

Fig. 1. The five gun turret with "A" the release for rotation, "B" the central control hydraulic control "C", and the rest "D" which operates with the direction of the gun.

military transport types. All these types of ships are forced to arm themselves heavily, inasmuch as the borders cannot carry fixed guns in hangars or weight for defensive purposes. Defensive armament must always be flexible. The heavy long range bombers will therefore always have the disadvantage of less maneuverability and speed compared to enemy fighter craft. Their defensive equipment of armor plating and flexible gun installation is limited only by considerations of weight and space. Present day large scale bombing is being done at night, for only then is there reasonable safety from enemy fighter pursuit ships. With improved methods being devised to employ patrol planes at night, large aircraft will shortly be susceptible to constant attack when over enemy territory. The loss of such large airplanes with its personnel represents a blow to our power, and every possible means of improving defensive armament must therefore be considered and adopted, if at all feasible.

A gun battery for aircraft turret mounting must necessarily be restricted to machine guns, and when above two in number, to light machine guns of 30 caliber. On very large bombers, it might be possible to employ three 30 caliber machine guns in a single turret and even incorporate the variable fire pattern idea. Usually, the light machine guns have the advantage of a very high firing rate, combined with low weight of gun and ammunition, thus allowing fast and easy adjustment of the individual guns of the mount. Actually, full armament positions of large aircraft call for gun portions of varying types, i. e., the multiple machine gun portions being followed by other portions carrying a single or two 30 mm. cannon to cope with conventional enemy planes.

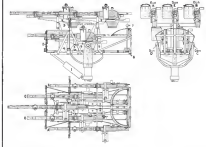


Fig. 2. Improved design of a 5-gun turret mount. Incorporating variable fire pattern mount. 1. Hydraulic cylinder for elevating and depressing gun mount. 2. Control leverwork for lower gun "D" and "E" and support for center gun "B". 3. Support for lower gun "C" and "D", 4. Support for upper gun "A" and "E". 5. Control lever handles for rotating gun and battery. 6. Control leverwork. 7. Control lever for rotating battery mounted on the central hand ship which are connected in electric trigger firing device.

The technique of firing from an aircraft turret at airplane targets is simple enough, but varies in accordance with the weapon used, and the position of the turret. Primarily, it is only a question of waiting until an enemy plane comes within the effective range of the gun, rapidly calculating the lead to be taken and pressing the trigger or trigger control button. The whole turret should show whether the line of fire is correct. If incorrect, an adjustment may be attempted if there is still a fraction of time.

(To be continued)

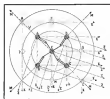


Fig. 3. Diagram illustrating the five gun turret with controlled dispersion system. The line outer gun could be spread out radially only from center gun "B".

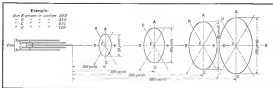


Fig. 4. Diagram illustrating the five gun turret with controlled dispersion system. The line outer gun could be spread out radially only from center gun "B".

When gun "A" is fixed and gun "B" is centrally fixed, curves in center "B, C, D" laterally.

PRODUCING 'CHUTES

Here is what is behind the scenes in building aviation's number one life saver.



One of the first steps in parachute production is cutting out the material for canopy and pulleys. The men above are using an electric cutter to cut away layers of cloth as one feeds in very carefully laid out patterns.

In setting the canopies (below) the parachute types of machines used are looms, trimmable machines, trimmable machines, and other machines for making the necessary stitches. The work room must be kept clean and dry at all times.



THE technique of parachute production has been developed by the British Empire, in a period of many years. This technique is also applied to the manufacturing of clothing, research in the majority of the steps actually are done on sewing and stitching machines. Although the machines are similar, the quality of

the work on the case of parachutes can be nothing less than the very best. Extremely accurate machines with experienced operators are necessary as it might almost be said a parachute is as strong as its weakest stitch. Every step must be repeated as there is no room for error. The following photographs illustrate production methods.

In the picture below can be seen a line of stitching machines for producing patch harnesses. The production line moves from the front to the back and on the left are the completed lines being inspected and prepared for reassembly in the assembly department. In the two lower sections is set in proper length and must be checked. The line working has a length strength of 2,000 lb. and the length 2,000 lb.



The trimmable sewing machines (below) used in setting the canopies are set up to cut the canopy and trim it to the exact size. After the canopy is cut the necessary inspection is made as this is an important step.



Over 30 sewing machines (below) are in constant use in producing the canopy for the parachute. This operation is repeated until the canopy has been cut to the exact size.

The steel frames (left) for the parachute pulleys are made by specially designed lathes. The operator at the frame right is handling the steel wire in the lathe while the one at the left is making the necessary corrections.



The completed frame is of an extremely strong design. A pilot is shown with a steel frame and steel pulley harness. The cable around the frame is attached to the canopy and held in the canopy cover which is on the bottom. When the canopy is pulled the cable is forced over by a spring steel mechanism which moves out a pilot. When the cable is pulled out the cable moves bearing operation.

200 is an elastic material. In order that all lines be exactly the same, the cable around the frame is pulled under tension and held in that position where the lines are cut they will all be of uniform length.

A Method for Solving SHEAR LAG PROBLEMS

By Eric Reissner

Massachusetts Institute of Technology



Dr. Eric Reissner

DESIGN experience has shown that the actual stresses in all-metal wings are sometimes up to 50% higher than the calculations according to the customary bi-linear moment diagrams. The reason for this can be roughly described by saying that the wing spans do not succeed in obtaining the cooperation of the metal skin to the extent the elementary theory of bending requires. After this efficiency-reducing effect would be absent if the skin material did not delaminate under the influence of shear, its degradation as "shear lag effect" is appropriate.

The purpose of this paper is to present an analytical method for the determination of shear lag in monocoque construction, which is both simpler and more general than methods previously given. The simplification lies in the fact that the use of linear trigonometric series is avoided and the method is more general insofar as it is applicable for tapered as well as isosceles beams.

The problem and its solution may be explained by considering a typical structure such as a cantilever beam with constant doubly asymmetric rectangular cross-section under the action of a given distribution of loads and with that of bending moments (Fig. 1).

The elementary theory assumes, without regard to shear lag, at every known station a uniform shear-stress distribution of bending stresses in the sheet and a linear distribution of rib web stresses. This leads to the formula

$$\sigma_x = \frac{M}{I} y \quad (1)$$

so that the shear stress would be given by

$$\tau_{xy} = \frac{K}{I} y \quad (2)$$

Now the spurious variation of σ_x would produce shear stresses in the sheet of magnitude

$$\tau_{xy} = -y \frac{d\sigma_x}{dx} \quad (3)$$

It is when one considers the deformation of the sheet under the influence of these stresses that the phenomenon of shear lag becomes understandable.

The bending stress σ_x would produce an extension

$$\frac{\Delta u}{u} = \frac{\sigma_x}{E} \quad (4)$$

The shear stress would produce a shear strain

$$\frac{\Delta u}{u} = \frac{\gamma}{2} = -\frac{1}{2} \frac{d\sigma_x}{dx} \quad (5)$$

where as both formulas the quantity u stands for the spurious displacement of the elements of the sheet.

Equation (4) would indicate that u is independent of the transverse coordinate y so that successive sections of the sheet remain straight. Equation (5) would indicate that every originally straight transverse section would be warped into parabolic shape.

To ensure that contradiction it is necessary to modify the simple equations (2) and (3) for the stresses which are used in the elementary theory and to find expressions in better agreement with what actually happens in the structure.

There is experimental evidence that the actual distribution of bending stresses in the sheet is such that it can be closely approximated by the following expression, which is the simplest possible modification of σ_x to agree by Eq. (2),

$$\sigma_x = \sigma_0 - \left[n - 2 \left(\frac{y}{b} \right)^2 \right] \tau \quad (6)$$

In this formula σ_0 is defined as in (2) and τ is a constant term due to shear lag. The quantity n is defined as

$$n = \frac{1 + 3.0(\tau u_0 / \sigma_0 b)}{1 + 0.4(\tau u_0 / \sigma_0 b)} \quad (7)$$

and is determined by the condition that the stress τ does not give rise to an additional moment about the central axis of the beam.

An expression for the shear stress τ , corresponding to σ_x , is obtained by means of the equations of equilibrium for the elements of the sheet:

$$\text{From } \frac{\partial \sigma_x}{\partial x} + \frac{\partial \tau}{\partial y} = 0 \quad (8)$$

there follows

$$\tau_{xy} = -y \left\{ \frac{d\sigma_x}{dx} - \left[n - 2 \left(\frac{y}{b} \right)^2 \right] \frac{d\tau}{dx} \right\} \quad (9)$$

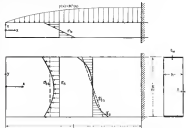
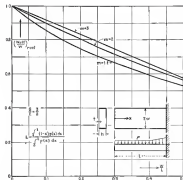


Fig. 1. A cantilever beam with constant doubly asymmetric cross-section under the action of a given distribution of loads and with that of bending moments.



where f is a single quadrant function of its argument.

To make W a parameter, for given σ_0 , by an appropriate choice of τ is a problem of the calculus of variations. Its solution is expressed in terms of a differential equation and boundary conditions for f .

If one defines two dimensionless parameters by

$$W = \frac{3(bn - 2)}{2(n - 2) - 3} \tau, \quad \rho = 5n \sqrt{\left(\frac{y}{b} \right)^2 - 1} \quad (10)$$

the equations which determine f can be written in the form

$$\frac{d^2 f}{d\rho^2} - \frac{W}{2} f = \rho \frac{df}{d\rho} \quad (11)$$

$$f(0) = 0, \quad \left(\frac{df}{d\rho} - \rho \frac{df}{d\rho} \right)_{\rho=1} = 2$$

(From Eq. (10))

Fig. 2. Stress pattern resulting from plotting equation (11) with values of $n = 2, 0.5, -1$ and 0.1 . (Reissner)

Though it is not possible to attain the stress-strain relations exactly with these expressions for σ_x and τ , it is possible to satisfy these much more nearly as in Eqs. (4) and (5), by an appropriate choice of the function $f(\rho)$.

To determine K , the least work method is employed. Knowing that the exact solution of the stress problem minimizes the internal work W of the structure, one demands of f that it makes W as small as possible.

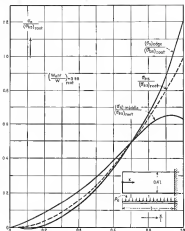
The expression for the internal work has the form

$$W = \int_0^L \int_{-b/2}^{b/2} \left(\frac{\sigma_x^2}{2} + \frac{\tau^2}{2} \right) dy dx \\ + \frac{1}{2} \int_0^L \int_{-b/2}^{b/2} \left(\frac{\partial \sigma_x}{\partial x} \right)^2 dy dx \quad (12)$$

where the first integral represents the work stored in the cross section and the second integral the work stored in the side walls. Eq. (12) contains the assumption that the contribution of the transverse shear normal stresses to the internal work can be neglected.

If one introduces into W Eqs. (6) and (7) for the stresses and carries out the integration with respect to y and x one obtains W in terms of a single integral

$$W = \int_0^L f \left(\sigma_0^2 \frac{d\tau}{dx} \right) dx \quad (13)$$



Where we stand in Aviation

In its relations with aviation, Goodyear occupies the same place it has long held in the automobile and motor truck industries — that of a major supplier of parts.

From the earliest days of the motorcar we have been manufacturers of tires, tubes, fan belts, radiator hose, motor mounts and numerous other rubber accessories — and of essential metal parts like rims designed out of our tire experience.

With equal faith in the future of aviation we began, thirty years ago, to apply our specialized knowledge of transport problems to the development of better products for all types of aircraft.

In close cooperation with the aviation industry we have perfected improved tires, tubes and other rubber needs — become large manufacturers of related accessories like wheels and brakes — acquired extensive facilities for handling alloy-castings used in aeronautical construction.

Continuing this policy, Goodyear today is erecting a great new factory devoted to the exclusive production of airplane subassemblies and parts.

This plant, to be operated by our subsidiary Goodyear Aircraft Corporation, will soon make available to the aviation industry our long engineering experience in fabricating metal parts for aircraft.

To overhauled manufacturers this will mean a dependable mass-production source of supply for wings, nacelles, floats, tail and other control surfaces such as we

are now building on subcontract for Glenn L. Martin Company, Consolidated Aircraft Corporation, Curtiss-Wright Corporation and Grumman Engineering Aircraft Corporation.

These new facilities will round out Goodyear's position as the nation's largest manufacturer of aviation tires, tubes, wheels, brakes and more than 30 other closely related rubber and metal parts.

We believe that our many notable contributions to aviation, including among numerous others the

Airwheel and Hydraulic Disc Brake, are the best guarantee of our ability to serve the aircraft industry advantageously in the future as a primary supplier of highest quality parts and accessories.

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Chairman of the Board
The Goodyear Tire & Rubber Company

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GOODYEAR

ON YOUR NEW SHIP SPECIFY GOODYEAR AIRPLANE TIRES, TUBES, WHEELS AND BRAKES

Detroit Gets New Airport

The story of how the Detroit Airport Committee overcame all obstacles in getting a much-needed new field is one that may help other communities finding themselves in need of a new airport.



Col. E. S. Evans

DETROIT is headed on its way toward getting an airport commensurate with that city's importance as a population and production center.

Plans for a new airport supplementing the existing municipal fields in the Detroit area are now so far advanced that only details remain to be worked out before the start of construction. This new airport, known as the "Class Four" airport, along with LaGuardia Field and the Cleveland airport—the only two airports in the country to hold this maximum rating of the Federal Government.

In the selection and acquisition of land for the new site, an Airport Committee headed by Col. Edward S. Evans, president of Evans Products Co., ran into so much more than a normal outpouring of political jockeying, attempted hold-ups by landowners, and opposition from communities bordering on the proposed site that a committee of their experience might prove of value to the other communities contemplating a similar expansion of their air facilities.

Detroit's major problem was overcoming The City Airport at Greater MI and Greater East-land as early as 1,200 operations of private and transport planes a day. Built in 1930, it was then regarded as among the best airfield airports in the country. But it was built for lighters, slower craft, and for a far less rapidly developing industry that had created two years later. Computer errors closed it among the most inadequate airports in any major city in America, and no provision had been made to allow for possible expansion.

Prohibitive obstacles to expansion in a 230-ft. gas storage tank situated by a local company on property adjacent to the City Airport. By a variety of luck and good navigation, the tank has never yet been struck. But Captain Eddie Rosenbader worked in an address at Detroit but agreed that unless the tank were removed a pilot would eventually do it himself. The major reason of the

fact is that it sits up stiff and impenetrable transverse in a high wind. Parity as a result, the GAA work program findings with a cloud ceiling below 600 ft.

Persevering American Justice Frank Murphy of the United States Supreme Court, when Mayor of Detroit, resolved the growing problem and named an Airport Committee, headed by E. S. Evans, of the Fisher family, with Colonel Evans as vice chairman. During several succeeding city elections, this committee failed to get administration support. Fisher died, and Colonel Evans became chairman.

Then last year, wealthy and well-meaning Edward J. Jeffries, Jr., moved into the Murphy's chair determined to do something about the City Airport. He joined in Colonel Evans the man he wanted—a straight-driving, enthusiastic and persuasive committee head who had the additional recommendation of a fine business and civic reputation. Despite the vast amount of his time and energy, which he takes up with the direction of the affairs of his company, Evans has never turned a deaf ear to any worthy civic project.

Colonel Evans, it will be recalled, created headlines the world over as

1928 when he guided the globe at his own expense in a specially chartered plane, ship and train in 28 days (a record which stood until a few years ago) to prove the feasibility and possibilities of commercial aviation. He also managed the St. Hubert polar expedition in 1929, brought gliding to America, still awards the Edward S. Evans gliding trophy each year and has helped promote and direct nearly a dozen airplane manufacturing and operating companies which are now internationally known.

Mayor Jeffries gave his committee practically a free hand, subject only to the limitations of an unincorporated administration. Colonel Evans immediately appeared unobscured to investigate the feasibility of removing the gas storage tank, improving the present City Airport and building a suitable location for another port.

Colonel Evans has constantly maintained that Detroit needs two airports: one for public transportation lines, another for schools and student training, a third for manufacturing and testing, and a fourth for private firms. He has just as consistently maintained that Detroit, playing an increasingly important role in the production of aircraft parts

(Turn to page 18)



Partial view of the present Detroit Municipal airport showing the hazardous 160-ft gas tank adjacent to the field. This tank has been an obstacle to pilots for years.

Drag of Riveted Wings

With so much emphasis placed on speed in present designs even rivet drag is an important factor. The following is the result of some NACA tests.

By Charles Payton Aubry, Boeing Aircraft Company

This discussion is intended to provide a practical insight of the excess power required in the drag of riveted metal wings with surface lips, being particular in small areas of high speeds. The entire types of wing surface construction and disposition the excess power required is of such an amount that this construction is hardly as satisfactory. Charts for a range of wing areas Reynolds Numbers and rivet disposition are plotted to indicate a means of estimating the drag in terms of the power loss. These power losses have been plotted with the inclusion of an 10 per centative efficiency.

THE NACA tests conducted by Hoof on wing surface irregularities considered the effects of rivet heads and surface lips on wing drag. These tests were principally on an NACA 20012 airfoil of 5 ft chord. The drag coefficient increments of the tests were employed in the power factors for the curves as shown in Figs. 2, 3 and 4.

The rivets and lips considered herein are those shown in Fig. 1. The results of C_d in this chart. The increments are based on these rivets for a 5-ft chord. The tests also consisted of runs on a like airfoil and rivet arrangement, of which the chord was 2 ft and rivets were two-thirds the size of the 5 ft rivets. Drag coefficients of the two rivets being increased by precisely equal amounts as equal Reynolds Numbers, although the Mach numbers were considerably apart. The tests are likewise applicable on the basis of Reynolds Numbers alone.

Characteristic laminar of the rivet and lips, in these tests, were given further for the power charts herein.

Spanwise pitch of the rivets for the following power curves is 0.025 of the chord as λ in for the 5-ft chord. The tests showed that rivet drag reduction was negligible up to the point of doubling the λ in pitch with the forward rivet row at 64 of the chord from the leading edge. This was probably due to dominance of the boundary layer, the drag increments being only slightly reduced by the slanting in rivet pitch. With the forward row at 25 of the

chord from the leading edge the rivet drag varied proportionately to the number of rivets as the pitch varied. The tests also indicated the effect of various positions of the forward rows of rivets on the drag increments. Sheet thickness of lips as dealt with herein was 0.03 in. Drag increments as applicable are taken at values where $C_d = 15$.

Power increments at Two Wing Types at 600 m.p.h.

Figs. 2 and 3 present the values of the drag increments for rivets with horizontal rivets and plain lips. Fig. 3 gives the values for counter-sunk rivets and pegged lips with the head flush of the airfoil sheet clad with a smooth surface film. This lip is concept to be representative of a lip having no drag, or that the lip drag has been successfully eliminated. The pegged lips at the plain lip portions without a film at the airfoil head, however, has a zero drag increment at an R of 7,000,000 and an maximum increment at an R of 13,000,000. Such a wing at 573 ft/s at 400 mph would produce a power increment of 45 at an R of 15,000,000. The drag under consideration in this wing is due to counter-sunk rivets, which is caused by an eddy flow around the rivet head, in the short, the result of pushing. Fig. 2 is not typical of a wing of extensive drag increment, possibly being as low as any average wing having plain lips and horizontal rivets, with the exception of any outward difference of position of the forward rivet rows and lips or effective rivet pitch increase that might be made. Sheets conforming fairly close to reality, without any excessive or unrealistic variations.

The rivets are in 12 rows at each surface of the airfoil. The plain lips are 4 in number on each surface. The

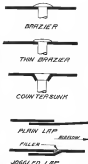


Fig. 1. Illustration of several types of rivets and lips used in estimating sheet metal air drag.

number or position of the jagged-filled laps has no bearing on the results, since they are assumed to offer no drag. The position of chest roots on each surface of each wing are at 94, 86, 72, 58, 38, 26, 44, 52, 66, 68, 76, 84 and 92 of the chord from the leading edge. The position of the plain lap centers of each surface are 96, 82, 36, 52, 68, and 64 of the chord from the leading edge.

The charts of Fig. 1 and Fig. 2 are for various Reynolds' Numbers which will give usable power values for the range of wing areas given, the 480 mph speed, average chords for the area, given and an average value of $\frac{L}{C}$.

The drag coefficient increase caused by the test plotted against Reynolds' Numbers were extended from the maximum of 16,000,000 to 20,000,000 to include the increase for the latter value.

In accordance with the power charts Fig. 1 and Fig. 2, a wing with boundary layer roots and plain all-filing tips of the number and disposition given, the power increase may be found to be about 22 percent by the use of jagged-filled roots and the jagged-filled lap or its equivalent.

Chart of Forward Root to Power Increase

Fig. 4 is illustrative of the power increments for a wing of 280 sq ft at 300 mph, with three types of roots. The abscissa of the chart is the most forward root. Root A of the first row are at the positions previously stated with the exception of the root forward of the position in question, which are non-existent. The jagged-filled lap or its equivalent is assumed. Root B forward of the second wing boundary layer transition point cause this point to move forward resulting in an increase of the turbulent boundary layer extension and a corresponding decrease in the laminar layer. Jagged point drag varied similar to that of roots. In Fig. 4 the roots forward of the 30 chord point constitute about 75 percent of the total power value when the forward root was at the 64 chord point.

Reduction of the power increment by moving the transition point of the roots forward of the fourth wing boundary point may not be appreciably accomplished unless the goal is of some worthwhile consequence above 925 of the chord, as previously stated from the selected tests.

Conclusion

A typical high-speed single-engine pure-jet airplane incorporating the type of roots and laps of Fig. 2 would be referred to yield as much as 26 percent

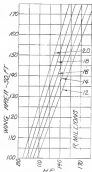


Fig. 1 Elimination of several types of roots and joints used in boundary layer root for aircraft.

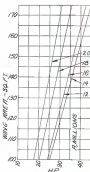


Fig. 2 Curves of horsepower plotted against wing area for 50 to 170 sq ft wings for various Reynolds' Numbers.

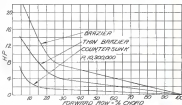


Fig. 4 The curves are for various types of roots at a constant Reynolds' Number of 16,000,000.

of its total power in the root and lap drag of the wing. This could be reduced to about 5 percent or less by the use of jagged-filled laps and counter-sunk roots and elimination of roots forward of the 30 chord point would further reduce the power increment to about 1 percent or less. The estimate derived at high speeds would be complex elimination of root and lap protrusions or indentations. Preferred

or smooth leading edges up to the 30 chord point and any practicable amount above are of much value.

References

1. "Wind Tunneling," The Effects of some Cases on any Airplane in Wind Tunnel, NACA Technical Note 1011, 1942.
2. "The Effect of the Shape of the Wing on the Drag," NACA Technical Note 1011, 1942.
3. "The Effect of the Shape of the Wing on the Drag," NACA Technical Note 1011, 1942.
4. "The Effect of the Shape of the Wing on the Drag," NACA Technical Note 1011, 1942.



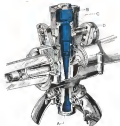
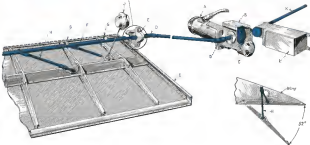


At the right is shown the main linkage gear retaining shaft used on the Vought 12 Series airplanes. The shaft is slotted through the tabs at the top with the gear quadrant alloy to retain the wheel which is mounted to the rest of the bottom.



The flap on an early Metal Lathboard Treatment is shown below. In blue is shown the power mechanism which supplies power for raising and lowering the flap. "A" is an electric motor which is connected to the transmission at "B" into the reduction gears "C". From this transmits

also and through the reduction gears "D" power is supplied to the thrust bearing assembly "E". This unit is also shown below, illustrating the thrust ball bearing "F" and the use of the gears "H". From the thrust bearing assembly the shaft assembly "I" can be moved longitudinally in either direction, being guided by rollers "G". Depending upon whether the flap is to be raised or lowered. The link "K" shows in the large detailed drawing connects this shaft assembly with the flap "L". "J" is the flap position indicator for recording on the instrument panel and is connected to the left flap only. Similar power assembly for the right flap is connected through "M". "N" is a limit switch for controlling the amount of flap travel.



Above is shown a runway section of the 1st assembly and intermediate of the 2nd assembly, showing the 1st assembly and intermediate of the 2nd assembly. This assembly is made by the 1st assembly under license from the Hamilton Propeller Manufacturing Company of the United Aircraft Corporation. The assembly mechanism is shown in the two pin positions for lateral and chordal. (1) from the engine laboratory system which is the rest of the power assembly "A" and follows the path shown in blue through "C" into the cylinder "B" which covers the cylinder externally, extending the blades and holding them in low pitch. When the air pressure is released the counterweights "D" are left to bring the cylinder back to its normal position and the blades then take the position of maximum pitch.



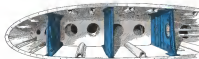
Shows above is a typical built-up rib as used in the Seabee 204 Clapnet. The cross members are U sections with aluminum being used for the top and bottom of the rib.



Shows above is one of the main structure members of the Seabee 204 Clapnet. "E" is the section for the pressure resisting structure the length of the flap. "F" are square tubes which are attached to the E-section spars. "G" is a typical fitting to which the main structure members are attached. In one of the main spars of this built-up rib there is a great deal of the load from the bottom end of the flap when loading. Above in blue are the pins of this built-up rib designed for taking load of the load.



The wings of the Vought Sikorsky VS-30 are built monocoque under low boom construction. Shows in blue are the aluminum ribs at the low boom construction of the wings which are attached at regular intervals along the top and bottom. The main boom is located at 17% of the chord and the rear boom at 47%. This with an upper and lower metal skin from the construction of a bridge box which carries all flight and load loads. The front and rear booms are made of extruded aluminum alloy flanges maintained in final shape in which is created a pin-joint rib with various stiffeners. Each shape of the boom is made up of a single extrusion and there are no laps or butt-joints in which extrusion runs start. In between also are carefully located to give the structure ample strength. The upper and lower skin extruding member of low section allows reaction system, resistance added to the skin to give the surface high rigidity and strength.





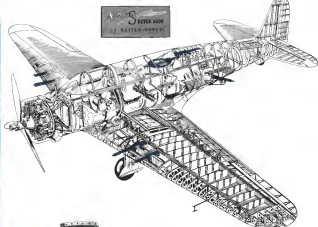
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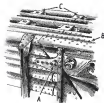


The structure of the Vickers V-100 is shown in blue. This ship is an earlier design of the V-100 series but nevertheless is well equipped with modern guns. The new observation plane can use either the top machine gun mounted on the control ship or lay on its stomach on the bottom of the ship and fire through the hull.



The attachment fitting for the Vickers Model K-10 is shown in blue. This fitting is shown at the left, indicated in blue is a typical open welded stainless steel fitting. The main pipe is attached to each of these sides by two bolts, one of which can be used on top and one on bottom. Above the upper and lower sections of the wings can be seen openings through which bolts are inserted to mount motor wiring panels with the control system. A section of the leading gear is shown below.

At the right is an enlarged drawing of the attachment of the upper section of the wing "C" of the Vickers Model K-10 with the main lower pipe "A". This view is at the same position that is shown from the underneath side shown in the drawing above. The leading gear attachment fitting is open welded to the front of the main pipe at "A" and above can be seen additional open welds for mounting the winged reinforcement underneath the upper wing surface with a concealed bolt type at "B". "C" are attachment fitting for attaching the upper surface of the wing.



Fleet's Trainer and Transport

Two Canadian models, an advanced trainer and a commercial air transport.

CANADA, being the primary ground for much of RAF pilot personnel, production of training planes is to be expedited, but when vehicle plant space, steel, cast and materials are devoted to commercial transports in wartime it speaks for the importance with which air transport is regarded.

Fleet Aircraft, Ltd., of Ft. Erie, Ontario, offers two outstanding models as the 3-place Model 60 Advanced Trainer and the two-seater Transport.

The trainer is an advanced step having semi-outboard wings with central bracing. Powered by a 180 hp Jacobs engine, maximum speed is 181 mph (sea level) and it cruises at 159 mph with a service ceiling of 16,000 ft, the wings climb at the rate of 1,250 ft per min and has a range of 910 mi with 80 gal auxiliary tank, range is 1,820 mi.

Fracture structure is semi-monocoque, and to facilitate maintenance it is constructed with a bolted joint at the second bulkhead aft of rear pilot's seat and a second bolted joint just forward of the horizontal tail plane. These joints may be actuated so that their strength will exceed that of a normal riveted joint, permitting replacement as any one of these portions of fuselage which may be damaged.

Wings are in two sections, a right and a left, which are bolted together in the mid-section. Each wing has a single boom of high strength aluminum alloy with internal bracing of soft castor cross-section in take off and the leading section. Metal ribs are riveted to main boom and extend aft to the tip and section ribs between. All drag bracing is taken in the leading edge.

The engine is mounted by struts and bolts through rubber bushings held in steel fittings on engine mount ring. Mount is a welded structure of 4130N

steel tubing bolted to the fuselage at three points on face of fire wall.

A unique system of oil temperature control consists of radiator area built into the oil tank. Upon leaving the tank on its way to the engine the oil passes through this heat exchanger which is supplied with hot or cold air under pressure through a valve valve controlled from the pilot's cockpit.

Cadaver type landing gear is provided consisting of individual shock absorbing legs mounted in main wing structure at the stress point just forward of the main beam. These are readily detachable from the wing as a tank being held in place by a bolt pattern at the lower wing surface and a single bolt at the upper surface. Right and left members of the landing gear are identical and interchangeable. Sufficient strength is provided to accommodate the use of steel, and all metal portions may be attached to main wing beam and landing gear by means of monocoque perforated bolt integral with the beam.

Specifications and performance data on Jacobs powered Fleet Advanced Trainer are:

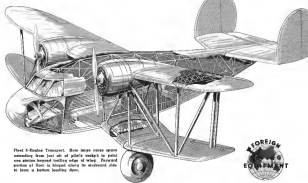
Gross Weight	5,000 lb.
Wing Empty Weight	3,457 lb.
Empty Load	207 lb.
Power Loading	11.8 lb./hp.
Wing Loading	28.5 lb./sq. ft.
Max. Speed (S.L.)	181 mph.
Cruise Speed (S.L.)	159 mph.
Landing Speed (S.L.)	58 mph.
Rate of Climb (S.L.)	1,250 ft./min.
Service Ceiling	16,000 ft.
Range (normal)	910 mi.
Range (Full Fuel Tank)	1,820 mi.

Fleet Model 18C Transport

Available in either landplane or sea plane models, the two-seater Fleet Transport is a conventional light plane having no stagger and little overhang;

Powered by Jacobs engines, each 180 hp, top speed is 158 mph, in the landplane model, or 141 mph, in the seaplane. Cruising speed (landplane) is 132 mph, landing speed 62 mph, and take-off is 60 ft or under full load of 6,326 lb is possible in 34 sec. Maximum rate of climb, 1,000 ft per min., and service ceiling, 15,000 ft. With one engine operating the ceiling is 4,200 ft. Extra fuel tanks of 70 gal capacity increase normal cruising range from 600 to 950 mi.

Especially designed to accommodate all types of freight, the cargo floor is a single sheet of 1757 aluminum alloy, .064 in. thick, supported by very closely spaced transverse beams. These beams are riveted to lower edge of the side wall structure which carries the floor loads into the main fuselage structure through large aluminum alloy cross



Fleet 3-Engine Transport. Note steps across span extending from just aft of glider cockpit to point one station forward leading edge of wing. Forward portion of tail is hinged along its attachment side to form a bottom loading door.



and bulkheads welded into the vertical fuselage members. Cabin side wall facing is covered with aluminum alloy plate .064 in. thick to a point about 12 in. above the floor, and from there to a point at the bottom of the cabin windows the walls are covered with corrugated aluminum alloy sheet .032 in. thick.

A distributed load of 2,000 lb. can be carried on the central portion of the cabin floor at full flight and landing factors. Forward and aft portions of the floor are designed to carry a distributed load of 1,000 lb. each as the main runway. All door sills are not stressed to sustain a load of 2000 lb with a factor of two concentrated at the center of the sill.

The wing wings, which form the main frame structure to a point outboard of

engine nacelles and landing gear fittings, are of aluminum alloy 1757 upper wing is bolted integral with the engine nacelle. The lower wing wings are of the same general type of construction having considerably heavy front and rear beams to sustain high bending loads imposed by gusts or landing gear. The three bracing the lower main wing is taken by top ribs of the main tail tank and a bolted into the wing structure on its low sides by closely spaced No. 12 steel bolts. The main tail tank is riveted to this upper surface with heat-treated rivets and the same sealed with Thiokol compound flanges in the main tail tank are closely spaced, making the entire assembly extremely rigid.

All engine controls are of special design having sliding members all the

flexible cable or stiff wire operating through fiber-lined aluminum flaked tubing without interference. This is to eliminate the possibility of trouble due to lubricants compounding at unusually low temperatures. Throttles are operated through a flexible aircraft cable working in parts. All other engine controls are operated by bellows copper hard draws, lead treated wire 0.02 in. diameter sliding in fiber lined flaked tubing.

The fuel system is separate and complete for each engine and the two systems are only interconnected through the hand pump. Cold weather starting is greatly facilitated by the positively reversible flow of preheating fuel. The preheating system is connected directly to pressure side of fuel fuel pump, and by closing delivery valve to the fuel tanks and opening primer valves, a continuous flow of preheating fuel may be delivered to the engine when hand pump is operated.

The landing gear is a welded steel structure has treated to 130,000 lb. per sq. in. Main shock absorbing member is, tapered to the front spar of lower wing unit. Wheels or disc are mounted on the lower end of this member, and the knee and all loads are taken by a single tapered member running from the side sill at the rear spar of lower wing unit. Span, 45 ft., length, 36 ft., height, 13 ft 6 in., wgt empty, 4800 lb., normal load, 9720 lb.

The Fleet Transport is fitted with glider landing doors which could be retracted while landing gear is in place.



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DUAL INSTALLATION IN CAA DC-3. Shown above is installation in the Civil Aeronautics Administration's Douglas DC-3. With the Dual Automatic Compass, bearings to two known points are automatically indicated on the Automatic Indicator to enable the pilot to know at all times the relation of the direction of flight to two known fixed points. The CAA installation also includes two Bendix RTA-1 Communication Units, Bendix RAS-32 Master Control Unit, Bendix MM-8 Equipment Mounting Rack, Jack Boxes and other accessories Bendix equipment.

DUAL-BEARING DUAL-ALERTNESS INDICATOR. (Right) is instantly read, and is automatically corrected for quadrantal error. The movable scale can be set in accordance with the aircraft's magnetic, or true heading, and variations in course are instantly shown by the relation of the true heading to two direction indicators.



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two of which comprise each Dual Installation. Each Receiver has three bands, two for Direction-Finding use, and the other band on one receiver may be used for high frequency communication purposes, and the other for dual Airport Control Frequency. Five standard models offer frequency coverage to meet every need.

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General Aircraft "Owl" biplane trainer. Notice the distinctive step for making the cockpit.

OWLET TRICYCLE TRAINER

The British have adopted this popular light plane for RAF instruction on tricycle landing gear ships and night flying.

By Paul H. Wilkinson *Canadian—Dial Aviation*

TRICYCLE fighters and bombers require tricycle trainers. General Aircraft's "Owl" described in this article is Great Britain's answer to the tricycle trainer problem. Not only is the "Owl" used for day training, but it also is used for night training. Furthermore, the safety characteristics of all metal low-wing airplanes are such that it is unusually suitable for blind approach extraction work. Royal Air Force pilots and cabin crew now gain experience on tricycle trainers ready to take over American-built tricycle airplanes.

General Aircraft claimed that the early stages of training are accelerated greatly with tricycle trainers because cadets have more confidence in making landings and therefore take shorter. Undoubtedly there is a good deal of truth in the British contention and the idea warrants serious consideration by those in charge of aircraft development in the United States. Here we are building vast quantities of tricycle fighters, bombers and transport planes, but we have no auxiliary tricycle trainers. This condition should be remedied without delay so that our cadets can gain experience on airplanes, low-wing tricycle trainers before proceeding to expensive, high-powered tricycle airplanes.



The main landing gear unit (A) is attached to the rear spar (B) by six bolts (C). (D) is the axle to the mainline wheel and (E) is the hub and the whole is at the point where the outer wing panel is bolted to the outer spar section.

The "Owl" is not the first tricycle airplane to be developed by General Aircraft Ltd. A civil version known as the "Cygnet" has been in production for some time. While the "Cygnet" does its share of work with its low-side seating in a truly streamlined cabin, the "Owl" has tandem seating in open cockpits in the approved military manner. Both airplanes are powered with a 150 hp. four-cylinder, six-cyl. air-cooled Cirrus Major built by the engine division of Blackburn Aircraft Ltd. (See *Aeronautics*, March 1941, p. 11.)

The fuselage of the "Owl" follows a

carved military pattern in that it is constructed in three sections to facilitate production. The nose section consists of an engine mount of welded chrome molybdenum steel joined forward with an aluminum casting which houses the propeller and the engine. The center section consists of an open metal box-like structure with members at the corners and a removable wood deck. The seats and controls are mounted on two longitudinal beams in this section in such a way that the seats can be slid out if necessary. The plywood covering over the beams also can be removed for inspecting the controls. The rear section of the fuselage consists of a monocoque structure with forged steel metal bulkheads and stressed aluminum skin.

The main wing is built in three sections, the center section being integral with the outer section at the fuselage. The outer section of the wing has two London type made of 8 R.55 aluminum alloy riveted to sheet metal ribs and stiffened on one side with riveted down sections. The ribs are of the sheet metal type flanged top and bottom and ribbed with riveted angle pieces. The wings taper in chord and thickness and are covered with sheet aluminum. The ailerons and the aileron trailing edge flaps also are



is all-metal construction with sheet aluminum covering.

The tail unit has twin cinders and fins and is of cantilever construction. Two metal spars are used for the stabilizer and it has a stressed metal cover cap. The rudders are fitted with cam balance similar to those used on the ailerons. A trimming tab is provided for the elevator.

The landing gear consists of three General Aircraft standard telescopic legs with interchangeable Dunlop wheels. The two rear legs are anchored to the extremities of the rear spar of the outer wing section and their wheels are equipped with shock absorbers. The front leg is steerable and rotates on bearings, attached to the forward bulkhead just behind the engine.

Each wheel has columns and rollers in its axle. The rubber tire also supports the steerable nose wheel. The aileron trailing edge flaps are operated by engine-driven vacuum pump. The two seats actually are of the ladder type for use with oxygen gasolators but they can be fitted with cushions if required. The seats are adjustable fore and aft and are mounted on four ball supports. A baggage locker is located behind the rear seat under the fuselage. Navigation lights and a landing light in the wing are standard equipment.

Two fuel tanks each containing 15 U.S. gal. of gasoline are carried on the "Owl's" one in each wing stub. The tanks fed by gravity to a standardized pump located forward the fuselage from which the fuel is drawn to the carburetor by means of an engine-driven pump. This system eliminates the possibility of air locks and tends to reduce the hazard in the event of engine stoppage.

The Cirrus Major engine which powers the "Owl" is a compact yet light weight power plant with a maximum output of 150 hp. at 2,400 rpm. It drives a Plessey two-bladed metal propeller 7 ft. in diameter. When loaded with 31 U.S. gal. of 70-octane gasoline and 26 U.S. gal. of lubricating oil, the "Owl" has a flight range of 410 miles at a cruising speed of 110 mph.

The weight empty of the "Owl" varies according to the purpose for which it is to be used. As a day trainer its weight empty is 1,263 lb., while with its night flying equipment its weight empty is increased to 1,641 lb. Its useful load is 737 lb. which

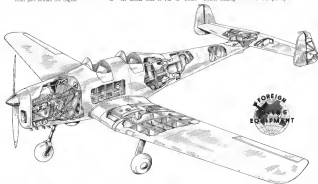
includes an allowance of 340 lb. for the crew and 40 lb. for their parachutes. Thus, its normal gross weight is 2,300 lb. but when it is used for aerobatics this has to be reduced to 2,000 lb.

A run of approximately 450 ft. is required for take-off and a 30-ft. obstacle can be cleared at a distance of 1,425 ft. from the point of departure. The stall speed is 77.5 ft. per minute and the service ceiling is 15,000 ft. When loaded with flaps and landing the "Owl" can be brought to a stop within 450 ft.

The civil version of the "Owl" known as the "Cygnet" is of cantilever construction and in fact the entire tail unit of the two airplanes are interchangeable. The "Cygnet" has a weight empty of 1,473 lb. and a useful load of 725 lb. including 135 lb. of baggage, making its gross weight 2,900 lb. Its maximum speed is 135 m.p.h., it cruises at 115 m.p.h. and it has a flight range of 430 miles. Its span is 34 ft. 6 in., its length is 23 ft. 3 in., its height is 7 ft. 0 in. and its wing area is 179 sq. ft. These dimensions are approximately the same for the "Owl."

Specifications of "Owl" Trainer (on day trainer)

Weight empty	1,263 lb.
Useful load	737 lb.
Gross weight	2,300 lb.
Maximum speed	125 m.p.h.
Cruising speed	110 m.p.h.
Landing speed	60 m.p.h.
Flight range	410 miles
Service ceiling	15,000 ft.
Wing loading	12.8 lb. per sq. ft.
Power loading	14.1 lb. per hp.





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The Blackburn Botha I

A British general purpose bomber used for land or marine operations in England's growing offensive against Germany

As a general purpose torpedo bomber the Blackburn Botha I is doing its bit in carrying the war back to Hitler.

Two Bristol "Pomero" Star 3A sleeve valve engines are carried in nacelles below the high wing. The engines are fitted with leading edge exhaust collector rings, and are enclosed in deep chord fairings with controllable cooling grills. Three blast, constant speed, retractable propellers are fitted. Blast portions of the nacelles house the retractable landing gear and are detachable.

Of all-metal construction, wire fabric covering on nose control surfaces, the Botha carries a crew of five. A peculiarity is that the nose of the ship is transparent plastic only on the starboard side, which is the bomb-aimer's position.

The fuselage is of monocoque, both raised construction throughout. A

power operated multi-arm turret is mounted in the rear of the main plane trailing edge. Extending along the starboard side of the cabin, a gangway permits intercommunication between forward and rear positions.

Composed of three principal sections, the center section of the wing houses the main fuel and oil tanks besides carrying the engine nacelles forward below the high wing. The outer sections of the wings taper sharply inward to the rounded tips and bear a pronounced dihedral angle. Balanced vacuum control using a portion of the trailing edge of the outer section, and hydraulically operated flaps along the trailing edge on the center section.

Both forward and aft engines are enclosed in cowls of streamlined metal construction. Landing and elevator air intake covered and are belted by a combination of metal bands and horn

bolts. Tapering ribs are fixed.

The landing gear consists of separate retractable oleo-pneumatic legs, a tail wheel and the engine nacelles. When the wheels are raised hydraulically, doors, mechanically operated close beneath the struts, giving a smooth under surface to the nacelles. Retraction is effected by hydraulic ram of the "rot cracker" type with roller struts. Landing wheels are fitted with hydraulically operated brakes. The tail wheel is mounted on an oleo-pneumatic shock absorber unit, the wheel having corner action with friction dampers.

For icing tests, equipment of the Botha includes a cold tank, ducting with induction coils, heaters, fans, etc., in addition to complete radio equipment and electrical gear.

Overall dimensions: Span, 58' 0" length, 51' 0" height, 14' 0" height, 14' 0" height.





Rearwin's Tandem Instrument Trainer

A NEW two-place tandem trainer Model 112T, marks the re-entry by the Rearwin Company into the stadium field. Designed especially for instrument training, this model is a brand new version of the Fleetwings Chaudier, the structure and aerodynamic features of which have been retained throughout. Also, the same reins are used throughout the control system, which employs ball and needle bearings as all points.

The trainer is actually divided into two separate cabins, one in front of the other. Each compartment has its own door on the right hand side of the plane, permitting the entire width of the cabin, 43 inches, to be used as instrument space. A useful feature is the private operator who does not have a full-time assigned schedule for instrument training is that this cabin may be used as a primary trainer simply by removing the rear instrument panel.

Particular attention has been paid to maintenance in the design of the trainer, especially in regard to the primary structure, flight controls, engine controls and engine installation. A new rifle two-piece cowling is employed which hinges at the top and makes the entire engine compartment easily accessible. Except for a small gill on either side of the firewall at the rear of the cowling, there is no unobscuring of any kind. The cowling is retained in a low and air direction by clamps which fit over lips cast integral with the rocker box caps.

A vacuum pump installation is provided to drive the full sets of dual flying instruments, which include Sponer directional gyro and artificial horizon. Primary flight instruments are mounted in shock mounted panels retained by Lind type frangs. Provision has also

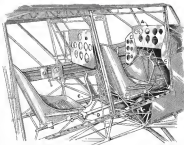
been made for many additional instruments in their correct spaces.

The list of this trainer items will be equipped with Lear transmitter and receiver, whip antenna, trailing antenna, DIF loop, wind driven governor, battery, etc. Intended as a trainer for the student, as well as with private operators, the instrument and general features of this airplane were developed through the cooperation of the Bureau Engineering Department with the American Airways.

Specifications and performance data of the Rearwin's Trainer are:



Wing Span	34'14 1/2"
Length Overall	21'5"
Height Overall	7'3 1/2"
Gross Weight	1900 lb.
Weight Empty	1340 lb.
Useful Load	560 lb.
Wing Area (line Airfoil)	241.7 sq. ft.
Wing Loading	11.23 lb./sq. ft.
Power Loading	15.8 hp./sq. ft.
Maximum Speed	125 m.p.h.
Cruising Speed	120 m.p.h.
Landing Speed	90 m.p.h.
Cruising Range	600 miles
Rate of Climb	400 ft./min.
Service Ceiling	14,000 ft.



The trainer is divided into two separate cabins and with both doors on the right hand side the complete 43 inch width of the cabin can be used for the instrument panels "A" and "B". Dual controls are also installed at "C" and "D".



FLEETWINGS PRIMARY-ADVANCED TRAINER BUILT TO MEET LOAD FACTORS

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FLEETWINGS TANDER TRAINER BUILDING STAMPEDED YIELD FOR ALL TRAINERS

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TO LOCKHEED, the greatest engineering achievement is not to have built airplanes that have set the pace in aviation's development...but to have built airplanes that have stayed modern through years of progress.

The Electra was Lockheed's first all-metal transport...one of the earliest two-engine airplanes to fly the airways. After it came the "12"...offering new economy and speed for private individuals and business corporations. Soon the "14" was developed...inaugurating wing slots and the effective Fowler Flap. And then Lockheed produced the Lodestar which again shattered all transcontinental speed records for airliners.

It is a tribute to Lockheed engineering that the basic design and inherent sturdiness of these commercial transports was so adequate that several nations are now using adaptations of them for military purposes. The "212" with rotating and retractable gun turret and other armament, serves in the colonial defense of the East Indies...and the "Hudson Bomber," has an amazing record of dependability in the Battle of Britain.

It is a further tribute to Lockheed engineering that when the United States needed a fighting airplane of higher efficiency for defense, the "Lightning" (P-38) interceptor pursuit was designed. World wide attention has been focused on its outstanding records for speed and the new tactical effectiveness that is offered by its performance.

Lockheed's new process—Electrolytic Transfers—accurately reproduces master templates from sheet metal. It consists of two steps: (1) Sealing of drawings on specially coated metal sheets. (2) Electrolytic Transfer of drawings with all the pertaining detailed information to metal sheets from which templates are cut.

Repeatability of reproduction, absolute accuracy, low cost, and economy of operation are advantages gained by Electrolytic Transfers. The two metal plates are held in firm contact in this special press during the process—Any number of duplications can be made. Time involved for making the complete reproduction is less than 30 minutes.



LOOK TO *Lockheed* FOR LEADERSHIP

LOCKHEED AIRCRAFT CORPORATION • BIRDBAY, CALIFORNIA, U.S.A.



ALUMINUM, DEFENSE, AND YOU



2

ANOTHER MONTH

DEFENSE IS GETTING its aluminum, but priorities are in effect, and many regular users of aluminum are having to do without.

YOU, SIR, may be one of those who have to wait. It is a hardship. It is awkward. Customer by customer, we are untimately and swiftly aware of the deflection of plans caused by this temporary shortage of metal.

BUT YOUR ALUMINUM is on the way. It is a promise.

IN MARCH we produced more than 44,000,000 pounds of new metal. That is 63% more than in the average month of 1939. Numerous new plants, already completed, made this possible.

STILL MORE producing units are coming in as fast as brick and steel and equipment can be put into place. We are getting superb co-operation from suppliers. A capacity of 60,000,000 pounds a month is definitely programmed, by day and date, at this writing.

PRAY FOR RAIN. Good precipitation assures the water power that we need to keep breaking production records. Production of aluminum depends on getting the power.

TWO POUNDS OF ALUMINA are needed for each pound of aluminum. We are jumping Mobile, Ala., alumina refining facilities from a million to 2,300,000 pounds a day. That requires almost other equipment, 64 precipitating tanks, 34 feet in diameter, standing 86 feet high. They would hold all the wheat grown in Wisconsin.

FORGING EXPANSION is an example of swift increase in fabrication capacity. On the first of January, 1940, we had 47 hammers, presses, and upstokers. Today 139. Increase 134%. We want delivery on 25 more units, which will make a total of 136, an increase of almost 280%. Future weight of the hammers alone is over three times that of January 1, 1940.

ROLLING CAPACITY for wire, rod, bar, and shapes has been increased 1 1/2 times. Expansion in sheet rolling capacity was reported to you last month.

COMPARED TO the fifty-some years it took the use of aluminum to reach 1939 levels, you might say that the industry is having to move than duplicate itself over the week end.

THESE PARAGRAPHS are factual evidence of our determination that no one shall have to forego the things aluminum does best one minute longer than we can help.

IT IS A PROMISE.

ALUMINUM COMPANY OF AMERICA



The Babcock Lightplane

THESE Babcock Aircraft Corp. of De Land, Florida, has announced its two-seat mid-wing cabin monoplane, the Babcock LC-13 A, which, with a landing speed of 42 m.p.h., should prove well adapted to small airports.

Powered with a Glenn L. Martin 120 hp engine, the Babcock has a maximum speed of 150 m.p.h., and cruises at 135 m.p.h. It climbs at the rate of 1,600 ft. per minute and has a service ceiling of 20,000 ft. Weighing 980 lb., this plane carries a useful load of 820 lb., including fuel and oil which is sufficient to give a cruising range of 450 miles. Performance may be expected to be increased with installation of a six-cylinder Franklin rated at 138 hp at 2,580 r.p.m., which the company is using in tests and with which equipment they complete an ATC.

Fuelage and tail group are of welded steel tube construction; wings



are fabric covered with wood, steel bound, box spars, and semirigid stressed ribs. The entering edge, nose ribs and covering back to the front spar are aluminum.

One of the outstanding features of the Babcock is that the area of vision is greatly increased by four large windows in the roof of the cabin, affording a clear view above and to the rear.

Specifications

Performance	
Maximum speed	150 m.p.h.
Cruising speed	135 m.p.h.
Landing speed	42 m.p.h.
Cruising range	450 miles
Rate of climb	1,600 ft. per min.
Service ceiling	20,000 ft.

Area	
Wing (including ailerons)	130.44 sq ft.
Ailerons	14.96 sq ft.
Horizontal stabilizer	12.72 sq ft.
Elevator	10.08 sq ft.
Rudder	5.26 sq ft.
Fus.	5.93 sq ft.

Dimensions	
Span	30 ft. 9 1/2 in.
Length	26 ft. 2 1/2 in.
Height	6 ft. 11 in.
Landing gear width	60 in.

Weights	
Gross	1,000 lb.
Empty	880 lb.
Fuel (Three tanks)	134 lb.
Oil	25 lb.
Engineer	70 lb.
Pilot	170 lb.
Passenger	170 lb.

Drains	
Gross L. Martin 223 in.-line 125 hp	
at 2,600 r.p.m.	



Aviation In Transition

By Selig Altschul

ALTHOUGH the 1948-49 aviation industry report for the year has not yet been published, in addition to being a complete documentation of the previous year's activities, it also reflects the rapidly changing status of the industry.

A number of general conclusions are readily discernible from the information at hand. All reports comment upon the unprecedented increase in wartime orders, consequent tremendous expansion in aircraft production, and other facilities. All also have noted a rather staggering of the industry's aggregate balance sheet. And here is a significant fact: The total costs of eight major aircraft companies—representing the bulk of the industry—totalled about \$650 million as of December 31, 1948. Of this amount, approximately \$373 million, or 55 percent, represented advances received by the companies. Such cash advances, made by the government and other customers for the most part, are allocated for materials and labor costs going into new plants. Yet to be reflected as its fullest extent in the Government financial report for new plants and facilities.

Continued awards for all government financed plant expansion, as all announced orders from July 1940 through February 1941 totalled \$1,350,615,280. In addition, letters of intent covering plant facilities estimated to cost \$294 million have been received. The British government is financing plant additions to the extent of \$171 million. Private capital provided only \$269 million. All told, \$2,136,000,000 in the overall cost of defense plant financing in the United States under government supervision as of the end of February. While no detailed breakdown as to industries is available, it is noticeable that the aircraft group has an important stake in this expansion program. One of the largest defense plant contracts involved \$57 million for the Wright Aeronautical Corporation.

Yet only two figures indicate the extent of plant additions yet to be made in the aircraft and other aerospace industries, but they may well describe entire firms concerned as to the post-war status of the aviation group.

The drastic reduction that followed the first World War is held up by many as the inevitable pattern that is bound to be duplicated in the construction of postwar facilities. The last wartime expansion, that the bulk of plant equip-

ment in the 1917-18 period was quickly reduced, followed in aviation circles when aircraft were sold on the day when the contracts were over. Expedients for new plants in our present rearmament effort are, for the most part, being borne by the Government. In fact, only about 15 percent of the expenditures represent private capital. This is an aspect of distinction and should carry particular weight as it pertains to the aircraft group. The larger aircraft plants will be entirely financed by the Government, with the aircraft companies bound to operate them on a fee basis. The Emergency Plant Facilities contract had a wide application in the aircraft industry. This plan permits the company involved to have its complete construction completed over a five year period at government expense. At the end of five years, the company has the option of acquiring the property or permitting the title to be assumed by the Government. Recently, the Defense Plant Corporation, an RFC subsidiary, has assumed the burden of financing plant expansion. Similar in many respects to the LTF wartime arrangement, Defense Plant Corporation holds title to the plant during the five year period. The

current builder rents the facilities at 10 to 15 percent plus upkeep and taxes. What it can be seen that the tremendous aircraft expansion is not being assisted by private capital. In fact, the industry may find itself in the favorable position of determining the extent of government financed plants it may decide to acquire based on the demands for aircraft that may exist in the post-war period.

While it is true the aircraft industry will have many problems to contend with in the transition from a rearmament program to a peacetime economy, overexpansion of plant facilities by private capital should not be a matter of undue concern.

The question of major aircraft companies, are so some state selling at less than their times reported earnings. This is merely a reflection of the widely asked question, "Is aviation really a war industry?" Under normal circumstances, the rate of return for aircraft manufacturers would be for the general run of stocks to sell at 18 times earnings, with growth equities related to sell at 20 times earnings. While it may appear difficult to reconcile prevailing conditions with this general rule, the answer is fairly simple. In 1945 and prior, as investors were willing to pay a premium for the normal peacetime development of a company which could be expected to realize a normal profit on increased business. But there are no ordinary reasons. Another answer to the market anomaly pertaining to the aircrafts, may lie in the fact that capital—naturally timid—usually gives one looking in wartime. With the emphasis upon "valuing first," rising earnings power tends to be equated at an discounts.

(Continued on page 112)

Annual Stock Summary

	for	for	for	for
	1948	1947	1946	1945
April 4, 1948	26.53	21.92	22.0	
March 28, 1948	26.12	20.46	22.0	
March 21, 1948	26.10	20.46	22.0	
March 14, 1948	26.10	20.46	22.0	
March 7, 1948	26.10	20.46	22.0	

Profit Margins for Major Aircraft Builders for Year Ended December 31, 1948

	Net Profit	Net Profit	Net Profit	Net Profit	Net Profit
	1948	1947	1946	1945	1944
Boeing	\$28,281	\$22,7	2.9%	2.9%	\$5,84
Curtis-Wright	1,030	1,481	26.2	18.1	3.38
Lockheed	136,170	12,147	28.6	12.4	16.1
North American	80,911	10,822	17.8	18.0	4.51
Republic	44,937	3,168	12.5	7.1	3.17
Wright	30,864	3,621	20.7	17.7	4.94
Wright Aeronautical	80,865	7,086	22.2	18.1	2.96
United Aircraft	127,447	12,146	28.4	19.2	4.94

* For Year Ended September 30, 1948.
† 1947, 1946, 1945, 1944.

The Finest that Money Can Buy



BONNEY Soft Face Hammers for Aviation Assembly

6 Types — 10 Sizes



HERE is a complete line of soft face hammers in all the most needed types that can be used for shaping and forming light sheet metals and the assembly of the most delicate and finely finished parts without marring or damaging them in any way. Bonney Soft Face Hammers are made in six types . . . 10 sizes . . . ranging from 1½ lb. to 25 lb. head weights . . . designed especially for the aviation industry.

Tips are replaceable . . . made of a tough, amber-colored, cellulose composition that will not chatter or chip. They are securely fastened to the tip shafts anchored in the steel center head of the hammer. Tips are interchangeable on seven of the ten sizes. They are easily replaced by turning loose with a pipe wrench and pressing on the new tip.

Handles are of first quality hickory, shaped to afford a firm, comfortable grip and are wedged securely in the steel center head. Handles and steel center head are attractively finished.

Bonney Soft Face Hammers are quality tools in every way, made to give long life under the most severe conditions and reasonably priced.

Your local Bonney Jobber carries Bonney Soft Face Hammers in stock—or write for Bulletin 99 and catalog covering the complete line of Bonney Sockets with handles and attachments, wrenches of all types, punches, chisels, files, screw drivers, files, feeler gauges, etc.

BONNEY FORGE & TOOL WORKS ALLENTOWN, PA.

In Canada—Grey-Bonney Tool Co., Ltd., Toronto
Export Office—38 Pearl St., New York, N. Y.
Stocked by Leading Jobbers Everywhere

profit controls these lamp-ignit all-control variables. The mechanism, highly sensitive in operation, is mounted near the control box but in a separate unit. The drag on the wire fully extended is about 6 pounds at normal cruising speed. The entire assembly weighs nine pounds.

An item of particular interest to the low-power field is a loading unit for use with a fixed antenna, which combines the function of a loading coil and a support insulator. Labeled type AR-6, this unit has a variable inductance range sufficient to load an antenna as short as 12 feet to the 3000 kc frequency. The unit is mounted at the far end of the antenna, where it serves also as an insulator attached to the fuselage and assembly in the plane. The coil is fixed with an adjustable slider by which the inductance may be varied to suit the transmitter. The loading is achieved by the inductor (sliding) length of the antenna, being the point of maximum current cut further into the exposed length of the antenna proper than the con-

siderable increasing the radiation efficiency of the wire. The slider can be used to compare favorably with those obtained with trailing antennas of moderate length.

The new, clean exterior of the Radio Division of Air Associates, Inc., P. O. Canby, Idaho, after many years was associated with the aircraft radio division of the RCA Manufacturing Company, serving as Radio Division in charge of domestic aircraft transmitters in that company until his transfer to Air Associates early this year.

Battery-Operated Transmitter-Receiver

A new type battery, portable, radio telephone unit has been introduced by the Electronic Specialty Co., Glendale, Calif., manufacturer of aircraft electrical receivers. Weighing 11½ lb. and 7 in. high, the Ranger unit carries complete receiving and transmitting equipment, dry-battery power pack and a two-way hand speaker which is automatically cut in when the headphones are released or when the unit is used as boom or other in-ear headset receiver. Average transmitting range is claimed to be 30 miles, with operation over greater distances frequently reported.

The transmitter is crystal-controlled on 3485 kc, and is put in operation with a push button control on the microphone. Fluorescent pe tubes are used, and a relay operated by the push button control cuts off the receiver filament during the transmitting period.

Flexible Receiver Now Western Electric Release

A new aircraft receiver, especially designed to offer high performance in installations where instrument-panel space is at a premium, is the 3A4 receiver being built by Western Electric Company, Kearny, New Jersey. The receiver's panel view space is accomplished by mounting the controls on the instrument board to the antenna, r-f, and tuning controls. The intermediate frequency amplifiers, detector, and audio equipment and power supply are mounted together in a subcarrier, larger size which may be placed anywhere within ten feet of the tuning unit.

The r-f unit has provisions for two crystal-controlled oscillator frequencies, and three continuously tunable bands covering 250-625 kc, 3000-7200 kc, and 6250-12,200 kc. The panel contains the tuning dial (with "semi-a" spots for logging the crystal frequencies) the volume control and on-off

switch, band-switch, beat-frequency oscillator switch and power switch. The sensitivity is high, 5 microvolts input for 30 milliwatts output, average, the power drain is 1.5 amperes at 24 volts (12 volt operation optional), and the receiver consumes 700 milliwatts, is high for a communication-type receiver. The power supply is of the dynamotor type. The amplifier circuits of the receiver are arranged for modulated, microphone operation if desired. The incoming signals and microphone current may be heard simultaneously by lowering the volume level. The receiver assembly, with cables and two units, weighs 16 pounds 3 ounces.



Amplifier models TR-1 and TR-2 have the same panel.

Airguide Transmitter-Receiver Features Compact Assembly

A compact transmitter-receiver combination occupying only 247 cubic inches of mounting space has recently been announced by Airguide, Inc., Edgewater, N. Y. The dynamotor model, TR-1, with three-quarters of a watt transmitter output, weighs 15 pounds complete with battery. The dynamotor supply model, TR-2, has a power output of 12 watts, weighs 16½ pounds. Each model comes in a case 7 by 14 by 14 inches with side panel mounting, suitable for supporting the equipment directly from the instrument board of the ship. The transmitter frequency range 2600 to 7300 kc, includes the harmonically related assignments 3175 and 6350 kc. The receiver covers 260 to 400 kc and 200 to 1250 kc, employs four tubes in the battery model, five in the dynamotor model. A dynamotor loop is available for the receiver section as is a trailing antenna kit for the transmitter. The battery-operated transmitter uses four tubes, the dynamotor type five tubes. Both models have simplified interphone connections and volume to monitor the output. An independent lamp indicator shows current. A built-in antenna loading coil, controlled from the front panel, permits the use of a variety of effective antenna lengths.



MR. PRESIDENT: We give you "Speed and Speed Now"

For America's Security and all we will do to the maximum the proper means of getting the job done. At 100 in volume's greater battery of light, power, which we will from complete parts of all types and sizes.



With a constant pressure of 5000 rpm, this engine can produce 14000 rpm over 15 hours. Designed as Douglas, this engine is so we make better through, and the smooth delivery. It is the only engine that can be used in the most difficult conditions.



Mass Production is the key to the technique of manufacturing assembly as performed by Douglas Aircraft Co., Inc., where "quality is essential" is a motto. The "quality" of Douglas Aircraft Co. is a motto. The "quality" of Douglas Aircraft Co. is a motto. The "quality" of Douglas Aircraft Co. is a motto.

"All Aboard" Here is Douglas plant layout, showing the entire production line, from the engine to the final assembly. The "quality" of Douglas Aircraft Co. is a motto. The "quality" of Douglas Aircraft Co. is a motto. The "quality" of Douglas Aircraft Co. is a motto.

"4,000,000 BUCKS" will be given Douglas wings in 1945, according to reports from the prime contractor. This is a very big figure, and it is a very big figure. The "quality" of Douglas Aircraft Co. is a motto. The "quality" of Douglas Aircraft Co. is a motto. The "quality" of Douglas Aircraft Co. is a motto.



Production "Speed and Speed Now" is enhanced by the use of new types of tools, in addition to speed-up improvements in the design and the layout of the plant. The "quality" of Douglas Aircraft Co. is a motto. The "quality" of Douglas Aircraft Co. is a motto. The "quality" of Douglas Aircraft Co. is a motto.



Douglas

FEEL ABOUT THE WORLD. LET IT BE IN YOUR HANDS. AIR SERVICE.

Mr. Associates, Inc. AR-6 loading unit for use with a fixed antenna.



E. P. Swick, new Chief Engineer of Air Associates Radio Division.

Production Swings Ahead

The production curve for Sperry Gyro-Horizon and Directional Gyros shows at a glance how Sperry is responding to the heavy demands for these important instruments. The ever increasing requirements of the national defense make it imperative that this curve swing upward faster than ever before.

Sperry is now producing Gyro-Horizons and Directional Gyros at the rate of more than 5,000 units per month—more in one month than the entire output of 1935.

SPERRY GYROSCOPE COMPANY, INCORPORATED
PRODUCTS NEW YORK

ACCELERATED PRODUCTION PLANNING

By the middle of this year Hamilton Standard Propeller production is expected to be ten times what it was in 1938. This will have been made possible by the production planning described in this article.

By Arvid Nelson, Factory Manager, Hamilton Standard Propellers Division, United Aircraft Corp.

The following material has been gathered from a talk given by Arvid Nelson, the superintendent of the Army Ordnance Department, and the American Society of Mechanical Engineers at a meeting held recently at Cleveland, Ohio. This summarizes material given by the Hamilton Standard Propeller Division of the United Aircraft Corporation in carrying out their remarkable accelerated production schedule.

HAMILTON STANDARD propellers, in common with the rest of the aeronautical industry, has had and is having an problem of capacity, materials, and priorities, and, to date, has been able to stem and, at times, even anticipate greatly accelerated schedules. To do this long term planning was necessary which presented a tremendous problem and a new line of thinking.

In order to have an accurate measure of the effort involved in the required task, Hamilton Standard has for some time used a sort of production which they call, in their terms, an equivalent factory profile. This, of course, does not necessarily mean that the physical task involved is an actual factory profile. It is, however, the most nearly accurate measure which they have today to determine the requirements of men, hours, floor space, dollar investment, and a number of production and management problems. From Fig. 1, showing accelerated production, an indication is given of the magnitude of the expansion which the plant has undergone and that which is hoped to be accomplished. This chart shows in 1938. From 1935 to 1940, a fairly constant, normal growth was maintained, but in the last half of 1939 the growth changes from the normal business growth to a rapidly accelerated production. There are four basic differences between the normal growth and the accelerated type of expansion



Fig. 1

The first difference is a very rapid rate of increase at the beginning of the expansion period. The second difference is that comparatively short period of peak production. The third difference is that the volume at the peak is extremely high in proportion to the normal base. Finally, at the end of the period, which is not shown on the chart, the rate of decrease will probably be very rapid. These four differences in the two types of expansion must be considered in any planning for accelerated production.

(Continued on page 186)

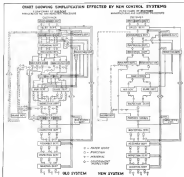


Fig. 2

BUYER'S LOG BOOK

What's New in Accessories, Materials, Supplies, and Equipment

A convenient hydraulic lift handling truck is offered by Lamm Iron Works, Green, N. Y., which is equipped with a switch for jacking down on and off the table. Truck illustrated is of 1,000 lb. capacity with 24" x 36 in. table having a lowered height of 12 in. and an elevated height of 51 in. In its standard position, the table extends about 1 ft. beyond the edge of the wheels so that the table may be put close to the press when taking out or putting in dies. Table is arranged to revolve with steps at each 90 deg. and is hydraulically elevated with a two-speed hand pump. A turn-around pump can be furnished with other capacities and specifications.—*AVIATION*, May, 1947



Lamm Die Handling Truck



P&H Propaganda Computer

A new plastic known as Resinex has been developed by The Resinex Co., Los Angeles, Calif. for the express purpose of eliminating high die costs without sacrificing speed of production. It is thought that wide applications in aircraft design can be made, including use in knobs, handles, fasteners, aerial clamps, aerial seat-ups, etc. Resinex is a liquid plastic which may be poured into the mold but which hardens quickly under heat and pressure.—*AVIATION*, May, 1947

To help solve assembly and quality control problems, E. E. P. Instrument Co., San Antonio, Tex., has developed a new instrument, the Propaganda Computer, Type E. Said to be as simple to use as to be applied with equal ease by either experienced or inexperienced pilot or instructor, the Computer has been flight tested by over 7,000 for use at the air. Reason ground speed is read directly by hand scales without any calculations. Map and watch are the only supplementary equipment needed. Only one movement of the Computer is necessary: rotation is then read directly on the one scale. All factors of wind velocity, drift, weight and ground speed are reduced to a simple factor of two, under constant conditions. Made of durable nickel-silver instrument can be carried in your pocket. Sold exclusively by Air Associates, Resinex, N. J.—*AVIATION*, May, 1947

Substantial simplification in stockroom and production line handling is envisioned as a result of revision made by Adel Precision Products, Burbank, Calif. All Adel chips now use aluminum handling material, eliminating necessity for 75-A coating previously required. Also eliminated is color differentiation of beams, insuring accurate speed to accurately dimensioned loaded chip. Black engraving is now used because of its superior quality over colored engraving. Adel will stamp all production chips with designation, "Max. Load" to differ certain new chip from original design black enameled stamped super chips manufactured by company.—*AVIATION*, May, 1947



Sundstrand Tooling No. 20

Designed for high efficiency milling of small parts, Sundstrand Machine Tool, Rockford, Ill., announces a smaller Rapidlat, No. 80. Of unit type construction and automatic lubrication, this greater feed machine has a hydraulically actuated table having a maximum stroke of 8 in. and rigid traverse, rate of 400 in. per min. Feed and traverse strokes can be regulated. Equipped with a 4 hp. specific motor, an outstanding feature is the wide range of spindle speeds possible with the high ratio belt. Between high and low speeds ratio is 425 to 1, making possible the machining of practically all types of materials without sacrifice in production or finish for lack of speed range.—*AVIATION*, May, 1947

Ignition markers suitable for aircraft engines, have been patented by Irvinson Farnish & Farnish Co., Irvington, N. J. These improved piston markers are said to meet the rigid requirements of both British and American aviation standards. Consisting of lengths of steel and cast aluminum-extruded heat-treated tubing, on which variable identification is screen printed, each specially formulated ink, the marker is furnished in standardized lengths which are slipped over the piston to be identified. Two diameters (nominal 1.0, .853 or .294 in.) are available with either of two heights or widths, and in two standard lengths.—*AVIATION*, May, 1947



Irvinson Ignition Markers



WHAT IS THE FUTURE OF AVIATION?

THINK YOU hear it said that aviation is "the industry of the future". You may ask why this is so . . . and what will become of the present great activity in aviation when the war is over.

Because my associates and I are engaged in preparing young men for worthwhile positions in the mechanical and technical phases of aviation, I am going to try to answer these questions, through a series of advertisements in this space. While I have been active in aviation for the past 28 years, I would not assume the role of authority—but will draw freely on the accredited views of others.

In appraising the future of aviation, it is well to ponder Captain Rickenbacker's bold and logical concept of the air as "this great universal ocean, making every city, no matter where it is on the face of the globe, a port of entry for aviation . . . by air you have one industry that can reach more places than the others put together".

The war will not change this fact. As the ocean of the air becomes the high road, a new era looms. One good that may come out of the anguish and horror of war is the hastening of the Air Age. Let us prepare for it.

C. S. Jones
—President—

Academy of Aeronautics, LaGuardia Field, New York
Cory Jones School of Aeronautics, Newark, New Jersey
Complete technical courses in aeronautics

TEXT FROM A SERIES OF ADVERTISEMENTS APPEARING IN TIME AND FUTURE

FEDERAL

Aircraft BEARINGS

QUALITY and PRECISION ball bearings are essential to the National Defense Program. The FEDERAL plant — working to capacity — is producing its full quota of fine ball bearings to meet the constantly increasing demands of the aircraft industry.



THE FEDERAL BEARINGS CO., INC.

Makers of Fine Ball Bearings

FOUGH-KEEFSE, N. Y.

Detroit Office: 2140 Park Tower • Cleveland Office: 462 Swindell Building
Chicago Office: 122 S. Wabash Ave. • Los Angeles Office: 6412 Wilshire Blvd.



Jerkor Aviaton Chronograph

A highly useful aviation chronograph, the Jerkor No. 660 waterproof, dust-proof and shockproof 37-jeweled time piece, made by the Jerkor Jewel Co., New York, N. Y., incorporates many features of value to the pilot. Embodying latest developments of Swiss watch construction, this non-magnetic watch has an hour running hand by means of which the pilot is able to know instantly the actual gas or flight time remaining. A sub-dialer indicates time into speed and speed into time, and a 12 hr. register, subdivided into 1-hr. periods, tabulates immediately the exact time consumed in flight without any calculations. A device for timing turns is another valuable feature and is calculated on the standard rate area of 360 deg. in 2 min. Scale is plainly marked up to 180 deg. for a 12 deg. turn every 3 sec. covering the complete revolution of the wing band in 1 min.—*Aviation*, May, 1941



Televis Poul Vibrometer

No more pulping, "contact," "switch off," "handle closed," etc., will be necessary for you lightning counts, now that Aero Federal Industries, Pittsburgh, Pa., is getting on the Aero Starter. Weighing less than 5 lb., the starter is used to do the work of 300 lb., and so simple that a 12 year old child can operate it with safety. A single pull of a convenient handle on the instrument panel starts the motor.—*Aviation*, May, 1941

Remarkable precision tests should be found in the alcohol battery for the Vibrometer developed by Televis Products, Inc., Chicago, Ill. As the name implies, the Vibrometer measures vibrations with extraordinary accuracy from 0001 in. to 1 in. in four ranges, to an accuracy of 3 percent. The modulus range is only 19 lbs. complete and is fully portable. The vibration pick-up is a lead foot which may be inserted into narrow and confined places.—*Aviation*, May, 1941



Genie '95' Gosh Compound

A complete decarbonizing process in package form is offered by The Corcoran Corp., Malden, Mass. "Hydro-Seal" for safety. Gosh Compound 95-8 is said to have been in service test for more than a year following the Military Development stage and to make a perfect cleaner for decarbonizing outboard and fuel pump parts, slushpan pistons, diesel engine pistons, etc. Six or more ready-to-use "Hydro-Seal" Carbon Gum Digestive kits make up a convenient retail wholesale stock order.—*Aviation*, May, 1941

A multipurpose unit for testing aircraft generators, vacuum pumps, hydrovac pumps, alternators, fuel pumps, etc., the U.S. Vandine Aero Test-Stand determines performance of accessories at all speeds, and safety factors at speeds far in excess of engine speeds. A product of U.S. Electrical Motors, Inc., with plants in Bedford, Conn., and Los Angeles, Calif., the Test-Stand is made in both upright and horizontal types up to 25 hp. Speeds of 715-5500 r.p.m. are obtainable for use on all commercial planes and routine tests on military aircraft, or 1,800-14,000 r.p.m. for research into higher capacities and speeds for Army and Navy work. Speed changes as small as 1 r.p.m. can be effected and a positive-drive magnetic generator and speed indicator shows the exact speed of the Test-Stand.—*Aviation*, May, 1941



W. B. Clark Aero Toolhead



Ray Repair kit with Handy & Harman "Easy-Fix"

A fast repair on a special top saved a mid-west plant from a serious production stoppage. Easy-Fix silver alloy, produced by Handy & Harman, 52 Fulton St., New York, N. Y., was used in less than a 12 in. top shattered in service and which could not be replaced by a new one in six weeks. Broken in an different way, the top repairers were back in service in 4 hrs. and three months later was 100 in. alt. Actual bluing time for this exceptional repair job was but 1 hr., using only 3 in. of 4 in. diameter Easy-Fix wire.—*Aviation*, May, 1941



Brush D-3 Surface Analyzer

Surface irregularities of less than one micron/inch in finished metals, glass, plastics, etc., may be accurately recorded with the Brush D-3 Surface Analyzer made by the Brush Development Co., Cleveland, Ohio. These records show the amplitude of irregularities in actual and not RMS values of irregularities, also the number of irregularities in a certain area, indicating whether above or below the bearing surface. Another lead camera of pickup arm and its drive are mounted on an adjustable stand. Pickup arm contains a piezo-electric crystal actuated by diamond tracer point. Drive unit provides actuator necessary for tracer point to explore the surface under test. Calibrating amplifier magnifies output of the pickup, and the direct writing oscillograph makes an actual record of the magnified surface irregularities on a continuously moving graph paper—*AVIATION*, May, 1947



Colson Broaching Machine

Broaching machines are now available in a complete line of standard sizes ranging from 1 to 20 inch capacity and from 24 to 60 inch strokes. Colson Broach Co., Detroit, Mich., who also supply larger machines on order, offer pull-down type broaching machines with completely automatic handling of the broach, through use of a hydraulic handling mechanism located at top of the column. Pull-down type broaching machines are especially advantageous under conditions where damage may result from the dropping of finished parts, as in manufacture of precision parts for aircraft engines. The line of Pull-down machines follows regular Colson milled steel construction. Length of stroke is adjusted by means of stops on the column. Manual operation is standard and the drive is hydraulic, fully enclosed. Spindle are variable, available 30 feet per minute drive to a full 60 feet per minute on the return stroke—*AVIATION*, May, 1947



B-W Package Unit

Skilled circuit production—in steel toward which every manufacturer is working—may be greatly facilitated by an ingenious "Package Unit" put out by R-B-W Mfg. Co., Division of Rayco Wire Corp., Laguna, Calif. An old, established firm, R-B-W will do a special job of supplying complete electrical equipment for units according to particular operating and space requirements at a low price. With the Package Unit, simple installation is a job in memory. A complete unit, polished to the last detail is carefully packed in an individual box, shipped to your warehouse and distributed to the production line as required, ready for final assembly. As manufacturing specialists, producing a variety of electric switches, relays, magnetic controls, wiring harnesses, etc., R-B-W is constantly engaged in the improvement and development of its products to insure the simplest design and most rugged construction—*AVIATION*, May, 1947

One of the most interesting and promising steps in wire, say disk recently came from DuPont Engineering Co., Newark, N. J.—about its multiple electric spot welder. From two to 12 spots can be made simultaneously, depending on the nature of the job, and adjustments can be made to make circular, straight, square, rectangular, or any odd-shaped weld. Machine is primarily intended for spot welding of light sheet metal work where a large range of adjustments for various spots and shapes are to be considered. Precision dielectric are used with suitable air operation equipment and have individual adjustment to maintain an even pressure on all spot welds being made—*AVIATION*, May, 1947



DuPont Multiple Spot Welder



TAKE THE Shortest DISTANCE BETWEEN THESE TWO POINTS With BW Prints!

BRUNING "TS" PRINTER AND 154 BW DEVELOPING MACHINE

Three new machines will greatly help high efficiency print production on your press plant in operations today. In one hour you can develop 1500 prints, then you can develop 1500 prints in 15 BW Prints per hour and 1500 prints in 15 BW Prints per hour. Prints up to 42" in width and length can be handled. "How High Do Prints Really Develop?" The Bruning "TS" Printer and the Model 154 Developing Machine will give you the answer. Order Bruning prints and third order. Send for your 1947

YOU take the shortest possible course between plan and product when you use Bruning BW (black line) Prints, instead of blue prints.

The reasons are simple. BW Prints are produced in seconds—rather than the minutes that blue prints require. BW Prints require two simple steps—exposure and development—while blue prints require five. BW Prints need no washing and drying. BW Prints permit big volume production of

prints cut to the exact size of your tracings—no unrolling and cutting of roll stock to get the prints you want. No wonder BW Prints smooth out "kinks" in production all along the line!

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Hand-Supplied Radio



Hand-Supplied Radio



Hand-Supplied Radio



Hand-Supplied Radio

An unusual hydraulic radio is the self-contained unit brought out by The Mosch Co., Boston Harbor, Mass. With hot wire in its antenna and all batteries enclosed in a sliding drawer inside the cabinet, the unit is previously mounted in instrument panel with brackets supplied by the same factory. Of special interest is the handy instant switch from beam frequency to control tower which eliminates need for changing tuning dial. Complete antenna lead is covered—suitable beam and control tower—with operation good in any climate. Weighing 4 lb. with batteries, set includes radio frequency and circuit in RCA brand tape-recorder—*Aviation*, May, 1947

Often while flying you want to take a picture of your new airport, or of a close new job looking along just under some wing tip, or maybe just to satisfy a hidden artistic urge aroused by the scenery. Skyline Camera Co., at its new home in Cleveland Falls, Ohio, has designed a Superior Camera, which fills a definite need in camera equipment. Useful standard 35-mm. in diameter, the camera, ready for action, weighs only 4½ lb. with 7½ in. overall length, two full corrected-contrast 14½ lens, and a shutter speed from 1/1000 to 1/200 of a second. A brilliant direct-view finder gives a large upright image. Complete with film, range finder, photo electric exposure meter, flash synchronizer and carrying case—*Aviation*, May, 1947.

New "Flat-Lite" Eye Shield too use on all grinders is announced by Shelden Electric Tool Division, New Britain, Conn. Fitted with two bayonet type light bulbs it throws light directly on grinding wheel and work. Design and lighting arrangement provides 36 general work positions, adjustable in 15° increments. Eye shield is adjustable up or down, but cannot be moved to non-grinding position. No 600 "Flat-Lite" Eye Shield comes complete with two light bulbs, bolts and instructions for mounting and connecting—*Aviation*, May, 1947.

Having the remarkable property of containing flammable gases, the Lithium Atmosphere Furnace, developed by The Lithium Corp., Newark, N. J., heats both alloy and carbon steels, and iron, without decarburization, carburization, or scaling. A ceramic gas, contained within the furnace, excludes vapor which enters continuously from the cartridge refill and behind the work being heated. Danger from byproducts and explosive gases is eliminated from Lithium Furnaces which are fully automatic, with no controls, pumps or moving parts. Extremely simple, the furnace is a completely self-contained unit with low operating and maintenance costs. Temperature range—1250 deg. F. max., 1800 deg. max. continuous. Dimensions, 48 in. high, 38 in. wide, 48 in. long. Shipping weight 4250 lb.—*Aviation*, May, 1947.

Greater convenience in mounting rear heads from aerial cables, can be had with a new line of load carrying apparatus introduced by Air Education, New York, N. Y. Consisting of two inches and three eighths, the new Aeron apparatus is also used for mounting ropes and for regular and other special applications. New models, styles 3198 and 3981, are of straight head type, each 21 in. in length. Both have custom made heads and suitable steel ropes, mounting system can be installed by either lever or trigger, and type selected can be placed on top, on either side, or on the bottom of hook to suit the operator. New spring clips are known as style 187 bent to 75 deg., style 181 bent to 90 deg. and style 181 which is a straight up F in long—*Aviation*, May, 1947.



Air Education Cables



**Ever had
a worry
like this one?**

TODAY is Tuesday. The Committee meets Thursday.

Just as sure as grass is green, Johnson's going to bring his crackpot idea with him. And, just as sure as 2 + 2 = 4, the Old Man will fall for it. Unless somebody bats it down!

The somebody will have to be George. And George knows it.

In other words, this one is a business worry ... and you've had lots of 'em. It's simply a known fact that most any man's business has a way of getting under most any man's skin.

... which turns up one of the best known of all publishing facts:

To make a magazine strong, you have to make it important to its readers. To make it important, you have to mold it to some absorbing reader-interest. You have to build a combination of Audience and Magazine that unites like two

perfect gears ... like the movements of a fine watch ... like Management and Business Week.

Having done that, you'll have a magazine that's different from its fellow publications. You'll have a magazine with more strength, more power than those built along more general lines. A magazine whose editorial and advertising pages both deal with the same thing ... that are absorbing interest of its readers.

And ... chances are ... your magazine will then carry more advertising directed to your kind of reader than any other magazine of any kind. Advertising-wise, it will hold fast places like the thoroughbred it is ... like the champion of 'em all ...

Like Business Week!

EP—Five free months of *Management and Business Week* with a gift of 50 advertising pages. Nearly five times that of any other general business magazine!

Business Week
ACTIVE MANAGEMENT'S MAGAZINE

AVIATION, May, 1947

For the Defense Program

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to aircraft navigation, know that they now depend upon Mallory for parts which reflect the most advanced design and practical efficiency.

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Standardization Result Of Lead-Lease Buying

Washington (Aviation News)—Standardize procedure for buying airplanes for the British is still being worked out, but general outlines are developing. The immediate job is listing of new contracts to manufacturers for planes to be delivered under the \$7,000,000,000 British aid appropriation. These orders in the main cover production now under way or starting U. S. contracts, deliveries on which will be deferred.

Once it is decided what job a given manufacturer is to do, working out of materials is turned over to Army or Navy, depending upon which service regularly deals with that firm. Contract terms are to be substantially the same as used on ordinary U. S. military procurement.

Standardization as between British and American equipment is a major objective. No sudden operating losses in the American are planned, however, except perhaps in cases where the user ships, with greater differences, is being delivered to the type of service. A transfer situation is presented where a firm is building one plane for the U. S. and another, with the same tactical purpose and perhaps a basically similar design, is being built for the British.

Examples are Douglas's two attack bombers—the British TB-7 and the U. S. A-24—and Martin's two medium bombers—the B-26 and the B-27, and the B-29. In cases such as these it is planned gradually to isolate the two models through a series of design changes, rather than to disrupt production of one or both completely by a sudden standardization. It is also clear that this government way tends to select one American model which only the British have been buying last for the RAF.

Standardization is also being worked out for the RAF.

Standardization is also being worked out for the RAF.

Desk All-Week Trainer

Edward H. Desk, president of Desk Aircraft Co., Hercules Beach, Calif., has announced inauguration of an intensive program to develop and manufacture aircraft and aircraft parts of value to help meet the shortage of training machines in addition to production of numerous small parts. The Desk Aircraft Co. is already active in the aircraft industry.

Standardization is also being worked out for the RAF.



SELF-SEALING FUEL TANKS are being put into the new Douglas bombers. Knowledge of that is a frequent factor, which is material improvement in quality. Between is a layer of rubber composition which seals to clear a bullet hole.

Mooney is Provided For Detroit Bombers

Knowledge of that is a frequent factor, which is material improvement in quality. Between is a layer of rubber composition which seals to clear a bullet hole.

Standardization is also being worked out for the RAF.

Standardization is also being worked out for the RAF.

Regional Bargaining Pushed For Air Labor

Washington (Aviation News)—Apart from the CIO executive drive, the big push now is almost entirely in the region of the question of regional bargaining through employment stabilization committees of the sort which OPA's wartime program is setting up in the aircraft industry. If these regional agreements work out well, they may be a good deal of help in the war production effort.

Standardization is also being worked out for the RAF.

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Standardization is also being worked out for the RAF.

Standardization is also being worked out for the RAF.

Mara Federal Money For Plane Plants

Washington (Aviation News)—Federal money for the construction of aircraft plants is to be made available to the states through the War Relocation Authority. The money will be used to help the states in the construction of aircraft plants.

Standardization is also being worked out for the RAF.

Standardization is also being worked out for the RAF.

New Aircraft Parts Subcontractor

Standardization is also being worked out for the RAF.

Standardization is also being worked out for the RAF.

Wright at New High

Standardization is also being worked out for the RAF.

Standardization is also being worked out for the RAF.

Standardization is also being worked out for the RAF.

Regional Bargaining Pushed For Air Labor

Standardization is also being worked out for the RAF.

Standardization is also being worked out for the RAF.

Latin America Suppresses Axis Airlines

The past weeks have not been very happy for Germany and Italian airlines in Latin America, but only if the South American neighbors follow the United States' example in suppressing Axis flightlines. Last Peru went one step further and took over the whole Luftwaffe fleet, which included at least two fighters, and all ground equipment.

In addition, Pan American Grace Airways took one of the first concrete steps to combat further operations of Axis-owned or controlled airlines in South America, by starting a non-weekly air service in Rio de Janeiro between two important island cities, in direct competition with the German "Lufthansa".

On the other hand, the competition by German airlines in this hemisphere is made easier by their slipping through the blockade with new equipment, in this way they have received at least two modern transporters, new personnel and a considerable amount of spare equipment.

The Dutch "Lucht" will be given one more charter by Brazil, an official announcement said. The airline, which maintained the last air connection between the Axis and Latin America, paid a large fee for commercial flights on Brazilian territory and was informed that the next one would result in immediate expatriation of all remaining personnel and equipment.

A new service between New York and Toronto will be inaugurated by Trans Canada Airlines, pending CAA approval.

British National Airways. The Air subsidiary will open a line from Charleston and Hampton to Colombia and Delhi in British India.

The past months saw a beginning in a network of airfields between Canada and Alaska, for just one of Canada and the United States, and the functioning of hemisphere defense.

American Export Airlines will probably receive \$100,000 for three proposed mid-air trans-Atlantic service, pending congressional approval.

Swedish ABA Results For 1940

Despite the war, operations during 1940 of the Swedish ABA appear to have been satisfactory. Connections to new routes maintained between Stockholm, Malmö and Copenhagen, Berlin, Stockholm, Oslo, Helsinki, Riga and Moscow, and have been flown practically without interruption.

Flights for 1940, compared to 1939, follow passengers 32,000 (29,000), seat occupancy 74 (64) per cent, miles flown 1,100,000 (1,000,000), mail 505,000 lb. (475,000 lb.) and freight 1,100,000 lb. (975,000 lb.).

De Schedule

"British carrier air" is the slogan that is heard so often in the present days, and so far as England's few flag airlines are concerned, they are correct on. Little has been published about the British flag airlines since the war began, and it is not generally known that more than 70 per cent of all passenger mileage is being flown on regular schedules by British Overseas Airways. This company, organized from the Royal Canadian Airways and British Airways in a government incorporation plan, now is the prime representative of England in Africa, Asia and Australia, and is doing an excellent job in these distant times.

About 10,000 miles per day are flown by BOAC's airlines, with passengers, mail and freight, and though there are Government requirements for spare because of the many varied passengers and cargo that have to be carried, a respectable number of passengers are brought to this destination with the same old efficiency that was so noteworthy before the war. Actually, there have been no accidents during wartime operations, though of course the "Clyde" was lost in the hurricane in Farnham last year, and the "Concor" made an unscheduled landing on the African jungle, but it was repaired on the spot and flown out by Capt. Rogers, one of England's air transport pilots.

On the whole, all routes are flown with the well-known short hours, but BOAC's equipment is notably international, and consists of Lockheed 14's and Lockheed, Junkers and Focke-Wulf, besides English types such as the Empire landplane, Short boats and De Havilland Handicrafts.

During the summer months of the past years, and continuing through the winter season, a route to Bermuda was flown by Imperial Airways and Pan American Airways. With the seasonal suspension in 1939, there were plans to send over a service and better last to support this service on Airways Bermuda Ltd, but the war changed all this, and in due course most of the flying boats that could be operated were taken over by the British Government, thereby eliminating British flag schedules, and Pan American was left to continue the service as it was. This summer, however, the Atlantic will again be crossed by British flag planes, as provided by Airways Atlantic, a BOAC subsidiary, using a new type Kestrel, but the Short S-50 C Class, with a gross weight of 10,000 lb., that could fly high was maintained last summer. In addition, the three Boeing 314's obtained from Pan American by the British may be moved over to BOAC and used on the Empire routes, thereby making some planes available for the other services, especially the bottleneck between England and Lisbon, at present served by Douglas DC-3s and Short boats.

A rather remarkable story came to New York from Buenos Aires, which would indicate that some sort of an interchange of military status between the Axis and Britain, at least as far as commercial airlines are concerned. It has always been more or less striking that none of the transporters still flying in Europe, or from Africa to South America, have ever been shot at, and except for the mistake of a mobile mine, who took out shot at a KLM DC-3 marker in the war, nothing was done to prevent the plane, no attempts have been made to bring an Axis plane down. There have been many stories about enemy airplanes carrying transporters with notable passengers on board, but no one has ever given a hard explanation of these facts. The story goes that as long as the Axis planes British "last look" to Lisbon airport, England will not interfere with the Italian Lufthansa from Portugal to Latin America. This somewhat curious but understandable attitude has the Axis to keep to non-military lines to South America, and the other hand does not force Britain to maintain a North Atlantic airline in winter, impossible with present equipment due to the frozen-over harbors at Bordeaux and Marseilles.

By "Viate"



Five military councils in coordination and control. George A. Brown, vice president of Imperial Airways, is shown with the following changes: CARL A. COOPER, left, is now executive vice president; GEORGE A. BROWN, right, becomes director of Imperial Airways for all pilots; R. G. Houghton is named ferry pilot of new Long Beach plant, George R. Tuley, chief pilot; H. E. McGowan, right; G. A. Houghton, director of testing for all pilots.



Flight to London was made by EDWARD WARRIOR, vice president of CAB in house of absence to assist Mr. Houghton, personal representative of Dept. Research in England, as technical advisor.



H. E. McGowan, right, is shown with the following changes: CARL A. COOPER, left, is now executive vice president; GEORGE A. BROWN, right, becomes director of Imperial Airways for all pilots; R. G. Houghton is named ferry pilot of new Long Beach plant, George R. Tuley, chief pilot; H. E. McGowan, right; G. A. Houghton, director of testing for all pilots.



Ray Tommy Tomlinson, assistant chief of Manufacturing Unit, Aircraft Section, OPM, is shown with the following changes: CARL A. COOPER, left, is now executive vice president; GEORGE A. BROWN, right, becomes director of Imperial Airways for all pilots; R. G. Houghton is named ferry pilot of new Long Beach plant, George R. Tuley, chief pilot; H. E. McGowan, right; G. A. Houghton, director of testing for all pilots.



The new firm of Air Corps, vice president of the Air Corps, is shown with the following changes: CARL A. COOPER, left, is now executive vice president; GEORGE A. BROWN, right, becomes director of Imperial Airways for all pilots; R. G. Houghton is named ferry pilot of new Long Beach plant, George R. Tuley, chief pilot; H. E. McGowan, right; G. A. Houghton, director of testing for all pilots.



E. B. CAMERON, vice president of the Air Corps, is shown with the following changes: CARL A. COOPER, left, is now executive vice president; GEORGE A. BROWN, right, becomes director of Imperial Airways for all pilots; R. G. Houghton is named ferry pilot of new Long Beach plant, George R. Tuley, chief pilot; H. E. McGowan, right; G. A. Houghton, director of testing for all pilots.



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POCKE-WULF 100 is one of Germany's new planes to go into service this spring. It is a ship-like aircraft ship, with a general similarity to the Lockheed P-38.



JOHN RANSDORN, vice president and director of Sperry Corp., takes an additional change of duties for Sperry Corporation, as has been vice president, treasurer and director of Sperry Corp. since 1935. At the same time JAMES E. WEBB, executive personnel director and assistant to the president of Sperry Corp., moves as to treasurer of Sperry Corporation.



The Reynolds Metal Co. has appointed HOWARD E. BRIDGES as superintendent of its new mill located at Lister, Ala. Among past jobs he has served as general consultant for OPM.



New material development of Republic Aviation is headed by LT. COL. HARRISON W. FLORES, who will coordinate purchase and handling of all materials and continue his former post of director of Republic, OCM MARKET, in personnel director of Republic, because executive assistant of the Department of Defense, with Curran and Gensler, Republic, in the Republic in 1935, as made chief of Republic Department.



New material development of Republic Aviation is headed by LT. COL. HARRISON W. FLORES, who will coordinate purchase and handling of all materials and continue his former post of director of Republic, OCM MARKET, in personnel director of Republic, because executive assistant of the Department of Defense, with Curran and Gensler, Republic, in the Republic in 1935, as made chief of Republic Department.

Steam Engines for German Warplanes?

By Dr. Leo Heisel

Reports reaching the United States that Germany is about to launch one of her "secret" projects in the form of an aircraft engine made of composite plastic material, were covered here last Saturday with utmost reserve, even though the news was indicated by the *Aviation*. It is possible this was merely the first time that the news engine was used in aviation. One of the first airplanes ever designed, the "Aurore" by General Ader (which is exhibited in a museum in Paris) was equipped with a steam engine. Its more recent evolution is now being the engine which tried to enter the steam engine's domain.

The steam engine for this particular purpose was presented in Germany for economic reasons, and not for military ones. These ones are completely cheap but all which is much easier for Germany to obtain than the expensive high speed engine for airplane engines.

One very important feature of the steam engine from the military angle is its silent performance, even though the noise created by the pistons could hardly be considered as that of a conventional engine. Other advantages of the steam engine in aviation are that the reduced fire danger due to the heat of fuel oil.

These ultra-modern steam engines on which Heinkel experimented have very little in common with the standard machines. The fuel used, as mentioned above, is a special, high grade oil. It is reduced under high pressure into the steam generator, a special kind of carburetor-like air as a certain valve in this air-valve mixture is used by a plug. The steam is gradually generated in a spiral pipe of very small diameter which is somewhat more than 500 feet long and should reach its bottom section exactly at the end of the spiral just before it enters the cylinders. The exhaust from the cylinders leads to a condenser and the same cycle starts all over again. The loss of steam and water is extremely low in these new models. It might be mentioned here that fluids other than water were also used in some experiments and that the water has first to undergo a special treatment. The pro-



be installed. The basic part of the boiler (1) where the steam is produced is generally shaped in a manner very similar to a turbine (4) which has a very small diameter and often is more than 500 ft long. The steam is directed into the highest pressure at the entrance end of this spiral just before it enters the cylinders (2) through the expansion valve (4). Its temperature rises from 430 deg. Centigrade and the pressure reaches 120 atmospheres or more. The exhaust from the cylinders is conducted to the condenser (2) where the steam is cooled and converted into water which is fed to the condenser (1). The condensed steam drives the small turbines (3) and (7). Turbine (3) operates the air pump (3) and the electric generator (12). The first one provides the heated air for the carburetor, the second one drives the propeller, which operates the pump (12) in the first line. Turbine (7) drives the fan of the condenser.

The condenser (10) provides for water replacement, acting as a reservoir. The water is pumped (13) into the boiler and the same cycle is repeated. The fuel is reduced under high pressure (11) in the first line. All these constructions have a special, automatic flowway, safety device, because an overheating of the steam beyond a certain degree must be prevented as it would lead to an explosion.

The steam engines of the latest construction of this kind have 2 or 4 cylinders. The water has first to undergo a special treatment. The pro-

cedure is not as simple as it seems. The wind-motor is the largest induction motor ever built, requiring 125 tons and standing 30 ft high. A steel shaft 12 in. in diameter will transmit the 10,000 hp, and when operating at top speed the motor will consume more than 30,000 kw of electricity. Despite its size, the speed of this plant induction motor can be varied from 21 to 200 rpm.

Combined weight of the 35 blades from 12 to 15 tons, each fan measuring 40 ft high. At top speed a restraining force of over 100 tons per blade is required to prevent each blade from tearing away.

Pre-doped Aircraft Fabric

Defense manufacturers of aircraft may be speeded by use of a new "pre-doped" silk fabric developed recently by a Cleveland patent concern. Fast dope used prior production methods drove out of the aircraft industry. With the new product, made by Stevens Williams Co., complete elimination of the present dyeing process is said to be possible.

Pre-doped in only of various weights, material has a heavy dye coating on one side of the cloth. This is dry enough to work with, but becomes hard-coded after being removed from its soaked original condition. Applied to the framework of the aircraft, the silk fabric, fabric lighter than it handles.

In the March 1942 issue, page 44 could not be immediately located for the sketches of the proposed attachment of the fuselage to the wing. The sketches of the proposed attachment of the fuselage to the wing. The sketches of the proposed attachment of the fuselage to the wing.



CALLED THE "SLOW TESTER," this device was developed by R. P. Davidson Co. to test aircraft fuel tanks. Devise tests the fuel tank in 30 to 40 minutes as to measure the effect of the force of the shaking fuel on the tank.



From husky aircraft engine bearings to jewel-like midgets for sensitive instruments, every New Departure is the product of more than half a century of experience in precision manufacturing.

NEW DEPARTURE BALL BEARINGS

Committee on Ice

Washington (AVIATION Business)—A special committee to study the ice problem in air operation has been set up by National Advisory Committee for Aeronautics, with director Tammie Tamm, chief engineer for TWA, as chairman, and 15 other experts. One of two meetings have been held already. In formation to come, proposals and in the conference, will be taken up. The committee will be charged with the task of recommending a CAB report on the IALC. The committee will be charged with the task of recommending a CAB report on the IALC. The committee will be charged with the task of recommending a CAB report on the IALC.



TRAINING PILOTS is one of the biggest jobs on any airline today. This is one of TWA's simulators equipped for instrument flying. Inset: A. T. Vance has no trouble visibility from his back seat. Instructor is Hugh Davis.



PILOT'S INSTRUMENT PANEL on new Boeing 714A. Instruments layout and control arrangement look like those re-designed in light of past experience. Pilot now has a control for the pitot heat under the main—up of the lower instrument main. Engine passenger deck capacity has been decreased from 74 to 66 passengers, but allowing capacity is now 58 as against 34. Gross weight 84,000 lb.

GAB Reports 3 Crashes. EAL is Forced Landing

Washington (AVIATION Business)—Civil Aeronautics Board recently reported the findings that the IALC. The committee will be charged with the task of recommending a CAB report on the IALC. The committee will be charged with the task of recommending a CAB report on the IALC.

short errors. The report stated that the ship had been on the ice for 10 minutes on the landing gear. The Board found that the aircraft was on the ice for 10 minutes on the landing gear. The Board found that the aircraft was on the ice for 10 minutes on the landing gear.

long below the 400-foot ceiling. The report stated that the ship had been on the ice for 10 minutes on the landing gear. The Board found that the aircraft was on the ice for 10 minutes on the landing gear.

to hold altitude to a combination of man, fuel, and engine. In addition, the CAB plans to study 15 more airports with IALC, and within 4 months, the entire 35,000-airway system. No deadline is being set for dissemination of low frequency.

Pick-up Picking Up

Pending before the Civil Aeronautics Board is this morning's application for a certificate of airworthiness for a 10,000-mile "pick-up" system to serve night routes from Illinois north to the coast. The system is a 10,000-mile "pick-up" system to serve night routes from Illinois north to the coast. The system is a 10,000-mile "pick-up" system to serve night routes from Illinois north to the coast.

GAB Plans UNF and Instrument Landing

Concert has appropriated money for 37 new ultra-high frequency radio range stations, and for 100 instrument landing systems. The UNF money is reportedly from the Treasury to split in multiple. For four years the Radio Technical Commission for Aeronautics has been working on this development.

Chicago office. The CAB plans to study 15 more airports with IALC, and within 4 months, the entire 35,000-airway system. No deadline is being set for dissemination of low frequency.

Rates for the Masses

Seaward Foster Airline's application for a certificate of airworthiness for a 10,000-mile "pick-up" system to serve night routes from Illinois north to the coast. The system is a 10,000-mile "pick-up" system to serve night routes from Illinois north to the coast.

AIR TRANSPORT INDICATOR

April 1, 1941

121.18

Which is the rate of passenger miles reported by the Air Transport Association for all domestic routes for March, 1940? The figure for March, 1940, is 121.18 million passenger miles. For the same month of 1939, the figure was 118.18 million passenger miles.

Chicago office. The CAB plans to study 15 more airports with IALC, and within 4 months, the entire 35,000-airway system. No deadline is being set for dissemination of low frequency.

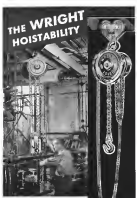
Tras-Globe gives foundation

Tras-Globe gives foundation for the Wright Hoistability. The system is a 10,000-mile "pick-up" system to serve night routes from Illinois north to the coast.

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WRIGHT Improved High Speed HOISTS HAVE "HOISTABILITY"

which gives vertical transportation with speed, economy and safety. EASY WRIGHT Improved High Speed Hoist is fast, smooth and positive in action—because of Wright's unique design and construction. It is economical, being highly resistant to corrosion because of its full steel construction. It is safe with a load chain that snaps in the point, regardless of operating position—because of the Wright safety guard. Furthermore, the load chain has a safety factor of 7 to 1, and is made of a special process steel which permits it to disengage under overload. It is the best hoisting device. Likewise the bottom hook will always open so that anyone can see the danger when loaded, beyond the danger limit of the chain. WRIGHT TROLLEYS are as dependable as Wright Hoists—made to give long, trouble-free service. Write for the WRIGHT CATALOG describing modern hoisting equipment in operation from 15 to 50 tons. WRIGHT MANUFACTURING DIVISION YORK, PENNSYLVANIA In Business for Your Safety AMERICAN CHAIN & CABLE COMPANY, Inc.

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But the Job's in Production Now!

TESTED by twisting—four times 'round, this both-weld of copper to aluminum proved that these two metals could be joined together in a satisfactory manner—with today's improved resistance welding doing the job. The finished weld showed a perfect bond even under a magnification of 250 times!

Because of improvements made in modern resistance-welding machines and modern controls, especially electronic control, resistance welding is first making yesterday's impossibilities a part of today's regular production.

If you now use, or can use resistance welding, you'll be assured of maximum production speed—which means profit, too. If you equip your welders with G-E electronic controls, connectors, circuit breakers, and cable General Electric equipment is readily applied to existing installations or to new-machine installations. In every case, it will pay you to specify G-E equipment.

Full details from your G-E office, or write General Electric, Schenectady, New York.

In EVERY case, it will pay you to specify G-E equipment

GENERAL ELECTRIC

Lockhead's first quarter deliveries approximated \$223,900,000, nearly as large as they were at the end of the entire 1949 year. March was the largest month in the company's history. Lockhead has earned \$21,441,000 in its affiliate company, Vert-A-Plane Co.

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Other sales included 708 Rock Aircraft by T. A. Wells

United Aircraft plans a \$14,000,000 expansion which will bring the firm's production capacity to \$65,000,000 during the next two years. In answer to a stockholder's inquiry regarding current earnings, President Wagon stated at the annual meeting that "there is no reason why we should be operating any differently this year from last year."

Industry Backlog

Confirmed orders of the aircraft industry have crossed the four-billion-dollar mark. For the first time and approximately \$4,000,000,000—four times as large as a year ago. In theory ordered by the large military orders now on the books of the industry, twenty-two years ago. In the first year of the World War, Congress appropriated \$900,000 for military orders. In 1917 the war program was \$700,000,000. A score or more of additional aircraft plants will be built in preparation for the new orders that will flow from the increased and supplementary defense appropriations. These, too, their immediate predecessors will be government, financed for operation by the aircraft and automotive industries. When the new facilities are completed, it is anticipated that 1949 orders for the Army and 1950 for the American air force will be supplemented on the present \$7,000,000,000 program. U.S. Steel estimates call for production of 10,000 military planes in 1949. The following table gives the latest up-to-date confirmed orders of inventory primary products.

Company	1949	1950
Boeing	1,000,000	1,000,000
Lockheed	1,000,000	1,000,000
North American	1,000,000	1,000,000
Republic	1,000,000	1,000,000
Grumman	1,000,000	1,000,000
Waco	1,000,000	1,000,000
Stearns	1,000,000	1,000,000
Boeing	1,000,000	1,000,000
Lockheed	1,000,000	1,000,000
North American	1,000,000	1,000,000
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Grumman	1,000,000	1,000,000
Waco	1,000,000	1,000,000
Stearns	1,000,000	1,000,000

\$4,000,000,000

AMPCO over England



American-Built Bombers and Fighters use **AMPCO METAL** at Points of Exceptional Stress and Wear

FLYING FORTRESSES and fast pursuit ships—**AMERICAN** built, but now in the service of His Majesty's government—fly daily over England—and the Continent, too.

Practically all these ships of American make employ **AMPCO METAL** at numerous vital points—wherever stress is severe and requirements exacting. In controllable price, propensities, resistance, bending, heat, rust, and guides, both new and a host of other points, you'll find this remarkable metal giving trouble-free service.

Takes Grueling Punishment

AMPCO METAL is an aluminum known without equal in its many qualities, unsurpassed hardness, high tensile strength, impact strength and resistance to corrosion. It withstands tremendous impact without warping, dent or peeling—ideal for most grueling punishment—and like all of them also takes **AMPCO** heat, made good where all other metals have failed. When you need a superior metal for some highly stressed part—look to **AMPCO**. Write for complete data.

AMPCO METAL, INC., Dept. A-5-6, Milwaukee, Wis.



SALES TO THE AIRCRAFT—**AMPCO** has secured 60 percent of Western Aircraft's landing steel as of 62 percent in passenger traffic in the first quarter of 1941 while Northwest Airlines showed a 70 percent gain in the same period. United Air Lines' West Coast line was up 20 percent. The Wright engine division of Curtiss-Wright Corp. accounts for 55 percent of the parent company's business. Pratt & Whitney has \$20,000,000 in military orders.

COMPANY—**AMPCO** officials state that orders on hand and in prospect indicate better profits this year despite higher costs. Shareholders of Aircraft Manufacturers Corp. have approved a new issue of \$30,000,000 of preferred stock to finance the company's \$2,000,000 building. United States of America Corp. has grown to \$13,500,000, with advances scheduled to increase each month this year. Trans-Canada Air Lines earned \$225,000 in 1940, an upward trend of \$411,000 in 1941. Shipments of planes by Piper Aircraft have averaged 1300 since the 1940 1500 deliveries with 437 C-47s manufactured in January and February compared with 200 in the same months last year. American Airlines increased its fleet to 10,000 by adding 10 B-24s and 10 B-26s and 10 B-29s to the fleet.

REPUBLIC AIRCRAFT has an upward movement with 1000 ships. Republic Aircraft has an upward movement with 1000 ships. Republic Aircraft has an upward movement with 1000 ships.

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COMMERCIAL AIR LINES have been relieved of one of their big worries as it becomes evident that they will get sufficient new transport equipment this year to handle a 25 to 30 percent increase in traffic. While the airlines have not received all the equipment they wanted, most of them agree that A. D. White, commercial aircraft division chief in the defense commission, has created their faith. Mr. White made such a good record in aircraft production that he has been given a specially high assignment or new priorities.

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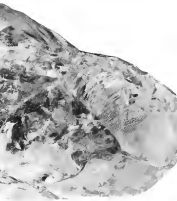
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comes strength to stand up under counter pressures, filing and rough handling. **ALBANITE** has all the working qualities you've always wanted—and it will retain all these characteristics indefinitely.

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TO MAKE MORE VALVES for aircraft engines, Thompson Aircraft Products will soon have 5000 more men at work in the Cleveland plant. Albert Kahn is doing the factory design.

AVIATION MAY 1941

K&E Albanene
THE STABILIZED TRACING PAPER

5 Bills for Gliders

Interest in gliders, for sport and especially for primary training, is continuing to grow. The government threatens to discontinue all the manufacture of small airplanes and would for defense. There are at least five bills before Congress which provide for \$500,000 to \$1,000,000 subsidies for glider promotion. The question whether jetless flights on a large scale is desirable, economically sound and efficient is being debated again. Years ago everybody went glider-happy but nothing much came of it. This time, important development is a possibility. One of the new airplanes is so far in that the cost is so low that boys and adult fans could afford to pay for their own training without much federal expense. Charles C. Malone, president of the American Glider Association, is working up a large following in California for the pending glider subsidy bills.

More Airport Classes

Expansion of classes for airport ground service men on 14 new airports throughout North Carolina, South Carolina, Georgia, Florida, Mississippi, Alabama and Tennessee were introduced recently by Col. Don H. Cassidy, CAA Administrator. Similar classes have been in operation since the first of the year at airports at Maryland, Virginia, New Jersey, and New York. The new program of a 90-day training period. Trainees are capable of performing all airport ground duties except those of certified mechanics. Airplane ages are 35 to 35.

WPA announces establishment of 17 similar service schools at Bendin, Dakota; Ridge and West Tennesse, N. J.; Roosevelt Field, N. Y.; Rye, N. Y.; Wright Field, Ohio; and Wright Field, California; and Elkhart, Ind.; Byrd Army, Richmond, Va.; and Conley Field, Atlanta. Courses are being set up at Raleigh, N. C.; Columbia, S. C.; Jacksonville, Fla.; and Tampa, Fla., and Memphis and Nashville, Tenn.

The aviation school of the Wichita Engineering Co., Wichita Falls, Tex., has been approved by CAA. Organized last June, the school has 100 students. Complete mechanical courses are given.

New Advanced School for Air Corps

The Air Corps' newest advanced school, located at Selma, Alabama, began operations on May 1 with a cadre of 180 students. This is the advanced school of the much better known Kelly Field. Cadets will work to Selma for the third two-year period of training.

The field is the largest airport in the United States. It is as large as 140,000 and 100,000 ft. The field area is 2,000 by 2,000 feet, and an additional 2,500 feet will later be added to the length of the field. It is estimated that the Army's largest plane could take-off and land, and take-off in a single line.



Advanced Air Corps school at Selma, Alabama showing a view of the barracks.

Foremanship Training at Republic

Republic Aviation Corp. of Farmingdale, N. Y. has inaugurated a similar course for foremen and would be foreman groups of about 30 men attend class at one time. All airport training and other duties will now be started by the company.

Students of Los Angeles are to be added to approve a \$75,000,000 loan over the next municipal election to expand and improve the Los Angeles Municipal Airport. Under the proposed expansion program, an additional 200 acres of land east of the present field would be purchased and improved. It is also proposed to erect a new administration and office terminal building, and to provide additional runway drainage, parking, and other improvements.

Airline Credit Corp. of St. Louis has opened a new branch office in the Airport Terminal Building, Long Beach, under the management of Dan N. Bishop.

When all the construction is completed, there will be more than 100 buildings on the field. A great total of more than 3,000 officers, cadets, and enlisted men will be stationed here. A new plan for quartering, flying, and training of the Western Method to be used in the Air Corps. Six men authorized to one room in barracks and between each two small fields. There is also being built a number of low rental houses for the families of enlisted men, non-commissioned men, and civilian employees. Commanding officer of the field is Commander Vincent B. Dunn.

A survey job completed by the American School of Aeronautics, Cincinnati, Ohio, reveals that hundreds of trained and experienced technicians are currently working in Army and Navy aircraft maintenance and repair positions. Civil Service and private flying clubs, which employ many Civil Service men, apply and receive pay to \$10 per month.

So scarce is the present shortage of trained instrument men, according to Robert W. Harris, president of the American School of Aeronautics, Cincinnati, that the Civil Service Commission has amended its requirements to recognize the school's six-month training course as a complete substitution for the two years actual experience formerly required for eligibility to these instrument positions.

The Reliability Institute now has about 500 in its aviation classes in New York City. The school gives one day a day and on coming over of 200 hours which can be given in a single day. The school, for most work, enthusiasm, and fast learning.

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Scoring Course

A course in scoring in gliding and soaring will be offered by the Eastern Area Scoring Course, which starts May 1 and will close June 21. Three kinds of courses are given: a seven month of two months, a seven month of one week, and weekly or bi-weekly training for the non-competitive glider pilot, or power plane pilot. John Zalkow, 3940 National Scoring Champion, will be the instructor.

Next public glider program is that announced by the State of Tennessee. Glider pilots from Tennessee will be available to support a substantial glider movement. Eight thousand boys and girls between 14 and 18 years of age have already registered for free courses. Much physical instruction will be given, as well as ground and satisfactory school grades and payment of \$20 per session for a year's membership. J. G. Taylor, Director of Aeronautics for Tennessee, is men behind the state.

A Tennessee chapter chapter has already been organized for primary instruction. Student groups of 100 will be started in Memphis, then Knoxville and Murfreesboro. Sub work in non-glider classes is to be added to curriculum. State also plans to start for private glider groups of five to six gliders.

Spartan School

Air Corps school at Spartan School, Tulsa, Okla., has been more than 70,000 hours since last July 1, according to Capt. Max Hickey, Director of the school. The school has been working steadily for many months. Most recent addition to the building program is a large cafeteria and ultra-modern kitchen equipment.

The aeronautical engineering course at Texas A. & M. is rapidly expanding and more than 800 students are already enrolled. A student branch of the Institute of the Aeronautical Sciences has been established. The local airport is being improved substantially and much new work is expected. The school, for most work, enthusiasm, and fast learning.



Every village and city in the Western Hemisphere has within the long range protection of the mighty B-19—newest Douglas Army bomber. The world's largest airplane, it has the world's most powerful engines—four Wright Cyclone 18's of more than 2000 HP each. It forebushes America's rapidly approaching supremacy in the air.

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Their speed in performance—savings in costs and ease of handling, plus the fact that they've got what it takes for long punishing service—makes SIOUX Quality TOOLS ideal for manufacturers on production lines, as well as for Transport Lines and airports on maintenance.

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Wet Grinding Built In



SIoux VALVE FACE GRINDING MACHINE WET GRINDER

Designed and built from the base up for wet grinding—most compact, complete and efficient machine of its kind. Assures factory production in wet grinding valves. Grinding especially 1/4" to 3/4" diameter. No. 835 comes completely equipped.



SIoux AIRCRAFT DUAL ACTION VALVE SEAT GRINDER

Accuracy with speed in refacing aircraft valve seats—either cast, hardened steel, bronze or alloys. The device grinds in adjustable. Universal motor operates on AC or DC. Net weight 95½ lbs.

SAE Papers

At the recent National Aeronautics Meeting of the Society of Automotive Engineers held in Washington, D. C., some twelve or more technical papers were presented. Although a complete review of all papers may not be printed here, the following abstracts are representative of areas of the subjects discussed.

Coasting Characteristics Of Submerged Light Aircraft Engines

E. H. Glendrock

Aviation Mechanical Engineer
National Advisory Committee for Aeronautics

An investigation has been started by the NACA to determine the performance of a small airplane with two Continental A-75 air-cooled engines installed in the wings. One of the main problems with engine installations in the wing is the cooling engine. The cylinders must be completely baffled and the cooling air quantity reduced to as low a value as possible. A low quantity of cooling air is necessary in order that the wing ducts may be small so as to cause little interference with the aerodynamic characteristics of the wing. In current installations the crankcase and oil pump of the Continental A-75 engine are cooled by air flowing over them but when in the wing no such cooling is possible. This causes the oil to heat up and increases wear and the possibility of preventing adequate oil coolers.

The Continental A-75 engine used in these tests is a four-cylinder, horizontally opposed piston, air-cooled engine, rated 75 brake horsepower at 2600 r.p.m. with 29.9 in. of air velocity pressure (dry) and 60 deg. F. Temperature is the intake standard. The base and super are 31 and 31 mm., compression ratio is 6.3. The displacement is 171 cu. in. The engine in the present airplane are to be used as pistons and for this reason the air is introduced to the cylinders from the end of the engine duct is normally the rear.

One factor that enters into the cooling of an engine that is submerged in the wing that is not encountered in present-day installations is that the crankcase is not cooled by air, as mentioned previously. This causes the oil to heat up and the bearing temperatures to be too high. An external oil cooler of adequate size must be used. The present tests it was necessary to hold the oil temperature to 160 deg. F. by means of a cooler to obtain satisfactory bearing temperatures. The oil cooler

could be submerged in the wing, in the same manner as the engine, and wing ducts used for cooling.

The shape of the duct system leads directly to the efficiency of a blower. If the better is needed to obtain satisfactory cooling. With the installation arranged as a pusher, a blower can be placed behind the engine in the propeller support. A blower of the axial-flow type could be easily designed to obtain efficiencies of at least 75 percent.

The propeller support would be larger than if just the propeller shaft passed through it but could be banded into the wing. In order to use the blower system efficiently, the cooling air after passing over the cylinders could be used to cool the exhaust pipes before discharging to the atmosphere. Tests on the engine showed that the air power, which was the brake power minus the power required by 180 percent efficient blower, remained practically constant as the pressure differential between the air intake and the exhaust could be varied satisfactorily, with 30 deg. F. between base temperature, without a blower it would probably be better to use a blower even at 10 in. efficiency a 75 percent and obtain the same cooling which would prolong the life of the engine. With a blower with 75 percent efficiency, the air power would be almost as good as without the blower.

A major disadvantage of placing the engine in the wing is the required engine accessibility. The relative in operating cost, however, will probably be sufficient to give incentive to efforts to overcome this disadvantage. One point worth noting at this time is that in the first blower installation used on the Continental engine in the present tests, the quick plug-in type, joined under the baffles and no provision was made for removing the plugs without first removing the baffles. In subsequent modifications of the baffles a hole was cut in each baffle. The engine which the plug projected, the hole being sealed with a locking pin. A few tests were made on the engine with a standard conventional duct of 76 inches diameter with the

original intake standards to compare the performance of the engine with the performance with 100 inches duct. With wide-open throttle the power decreased approximately 2.5 percent and with part throttle the power remained the same with 76 inches duct as compared with power with 100 inches duct. The decrease with wide-open throttle was possibly due to a lower velocity and lower limiting value of the 76 inches duct as compared to the 100 inches duct. At part throttle the superposition of the 76 inches duct may have been greater, due to lower multiple pressure, than it was with wide-open throttle, which would increase the velocity and thus cause the power output to be the same as with 100 inches duct. The use of the lower intake duct is substantiated in the axial airflow duct as to lower cost. Although there are many problems to be solved in connection with submerged engine installations the ultimate improvement to be placed here that warrants research on these problems.

Design of Aircooler For Aircraft Carburetor

H. L. Effer

Chief Engineer, Aircraft Division
Military Commission Co.

Some people concerned with aircraft carburetor problems feel that aircraft carburetors should be designed to work currently regardless of any installation circumstances, and the design of the carburetor should have an effect on the functioning of the carburetor. This would certainly be an ideal condition, but unfortunately, it cannot be immediately realized.

The carburetor entering circumstances for variations in rate of airflow, for changes in air density, and for changes in air temperature, are determined by the response of the carburetor mechanism to variations in air velocity pressure, and temperature. Anytime you have a problem involving the measurement of air flow in ducts has reduced the difficulty of obtaining true average values of air velocity, pressure, and temperature. . . . In this connection, it is interesting to note that similar to the motor it seems there is no air meter available on the market which will handle the quantities of air handled by a large-size aircraft carburetor . . . and read sensibly in the limits of an accuracy equal to a modern carburetor. It would therefore seem reasonable that the carburetor should be supplied with air that is flowing at fairly uniform velocities of velocity, pressure, and temperature terms the area of the carburetor entrance. This

(Turn to page 142)

Recent Books

THE STRONG FLYING TRAINING
POWERS, by Colonel Frank J. Kover.
 Published by D. Van Nostrand, 280
 Fourth Ave., N. Y. City. 172 pages.
 \$2.

This is the best book on pilot training that has thus far been published. It is written by one of the greatest experts on the flying business. Col. Kover has been flying for about 25 years. He was Chief of Staff of the G-10 Air Force in the period of its organization. He has had a large experience in pilot training for many years.

In his preface he states: "... Flying ability is no more inherent than the ability to walk. Any pilot with a normal physical endowment can learn to fly an airplane with safety." He also emphasizes the value of studying flying methods on the ground. He explains in detail the "Many methods of instruction make the initial error of teaching one to fly by flying. As a consequence students frequently are confronted with emergency requiring coordinated action, which they are totally unable to perform."

An outstanding feature of the book is the simple manner in which the use of airplane controls is discussed. A student who masters this material thoroughly will have gone a long way in his instruction. The book is filled with advice and wisdom which this reviewer has found is no other book is giving. The excellent illustrations make a considerable contribution to the text. In short, here is a book that will help you develop pilot, as well as air flight instructor with his job.

AIRCRAFT DEVELOPMENTS their description and use by Bruce Edmund Powers. Published by D. Van Nostrand Co., New York. 258 pages. \$2.75.

The author is an aircraft development technician and instructor at the Air Corps, as well as chief instructor at the Harnsberry Evening School and the Pittsburgh Institute of Aeronautics. He has written an excellent book for students and instructors who wish to understand the fundamentals of aircraft operation.

Chapters of the book are as follows: components, altimeters, rate of climb and descent rate indicators, air and ground speed indicators, turn points, electrical fueling systems, temperature gauges, oxygen regulating systems, pressure and vacuum gauges, instruments, instrument malfunctions and their causes, fuel quantity and fuel flow gauges, pumps. The book is illustrated with over 200 excellent photographs.

PILOTS IN AFRICA, by Richard Upshall Light. Published by the American Geographical Society, New York. 238 pages. \$5.

Here is a book that will make both pilots and amateur photographers green with envy. In 1937-38 Dr. Light and his wife flew their Bellanca from Michigan down through Canada and South America and then took a steamer to Cape Town. This book is the story of their flight from the southern tip of Africa up to Cairo and then across to Geneva. If we were giving a prize for the best travel book of the year this would surely win first place.

The book will appeal to pilots because it describes the kind of trip that all pilots dream about. To photograph the world will be in the least of aerial shots made by Mary Light, who could certainly help to sell air travel in America if she turned out even half as interesting a group of photographs taken along our coasts.

Dr. Light who is well known among private pilots for his flights to many parts of the world, was not only an other pilot on a journey. This was a flight with interesting observations and packed with interesting information on the countries of Africa through which he flew.

NOTES ON AVIATION, an Air Youth Manual for London. Published for Air Youth of America by D. Appleton-Century, N. Y. 286 pp. \$2.50.

Germany has received considerable praise for her aviation efforts in making Air Youth air-minded. It is generally recognized that Nazi strength in the air today is partially due to its long-range program at aviation training, as boys and young men in that country Air Youth of America has the only comprehensive plan for putting to good use the instruction received in youth air instruction. The program is not only wise education but considerable for national defense.

Air Youth's new book is the best volume for leaders that has yet appeared. It is 28 pages and is a book that presents advice for men who are in lead groups of boys in model building and in soaring. There are many handsome photographs and clear drawings. The book is a valuable contribution to aviation literature.

ALL AVIATION MECHANICS by Donald K. Green. Published by Thomas Y. Crowell Co., New York. 122 pages. \$1.

The author begins by saying: "These are the champions. Each airplane in this book has won its place because it

is loved by all Americans. Each type is the outstanding one of its class and type." There may be some disagreement about this but there won't be on the fact that this is a collection of beautiful airplane photographs. The accompanying text is fresh and to the point. Over 30 airplanes of all types are pictured and described.

WINGS OF VICTORY, by Ivor Holstead. Published by S. P. Dutton, New York. 221 pages. \$2.50.

This is a story of the R.A.F. from the beginning of the war through the events of last November. It is an exciting, interesting story of R.A.F. pilots and their heroic defense of Britain. It contains many new stories and episodes that have not been before and many fine illustrations. The famous "Dorchester" is repeated, as well as Charles Gardner's BNC broadcast of an attack on enemy which he saw. J. R. Planché has written an episode which is a moving tribute to R.A.F. pilots.

TOWN FROM HEAVEN, first-hand narrative of the war in England, edited by Atlas A. Hinkle and Walter Graeber. Harcourt Brace & Co., N. Y. 226 pp. \$2.50.

Ed's London correspondents, Alan McKee and Walter Graeber, present a collection of personal experiences of individuals actually exposed to the Battle of Britain. This collection of first-hand account by men and women who have participated in the battle of Dunkirk's R.A.F. side, submarine attacks, fire-fighting in London, bombings, hospital service during raids, cover battles, is a tape of courage, skill, and endurance. It gives a graphic picture of the nightmare war in England as it is being fought by courageous and resilient. No official communication, "approved" accounts, or newspaper reports can bring home as vivid a realization of the struggle being waged to do these tales—simply told men's words, heightened only by the "I" of their telling and the excellent photographs. If this is a cross-section of the spirit of the British people, it is not difficult to believe that there will be the victory.

TESTIMONY New and Patriotic Frontiers, published by the Elliott Service Co., 319 East 46th Street, New York City.

Elliott Service was founded in 1951 and since that time has made many important contributions to industrial education and training. It provided

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Simonds on Plastics

(Continued from page 33)

classified as mechanical. This, in the blade, shock absorbing rings and in the Hamilton hydraulic propellers. These rings are molded from phenolic plastic material and mounted between the main bearing race and the blade hub as illustrated in Fig. 4. The purpose of the rings is to absorb stress concentrations set up in the blade shock due to contact with the blade thrust race. They also prevent the points used in making part of the hub from chafing the shock of the blade root, and in addition provide improved sealing surfaces. The factors influencing the choice of this type of plastic for fire applications were its impermeability to lubricants, cushioning effect and its non-directional.

Plastic Invents

Getting back to aircraft inventors, what is believed to be the first all plastic instrument panel, Fig. 5, is now used as standard on Avian planes of the Aero-Mechanical Corp. of America. The primary considerations influencing this application were light weight and dimensional stability. The material selected to meet these, and other considerations was the polystyrene type from the DuPont Chemical Co. This material meets number one on the long list of plastics for low specific gravity, high electrical values and low moisture absorption. Other considerations, according to the manufacturer, are its lack of loss of color and moldability, to permit good structures for lamination of additional materials.

The use of transparent plastics for pilot cockpit instruments, instrument housings and other points for observation, has become universal, with the choice of materials used ranging from cellulose acetate, through cellulose acetate, and vinyl to methyl methacrylate. For military and transport planes the methyl methacrylate type of plastic material is generally specified because of certain inherent optical properties of light rays which this composite properties found in other types of transparent plastics. Though not possessing the hardness and resistance to abrasion of glass, methyl methacrylate exceeds glass optical properties equivalent to the finest quartz crystal and in addition has a low specific gravity and relatively high resistance to impact (6). A typical thickness of methyl methacrylate is the benchmark comparison, the barrel and the tailgun compartment of the Martin bomber, shown in Fig. 6. In the Boeing Flying Fortress, extensive use is made

of this type of plastic in the gun position in rotor, tip and location of the fuselage. See page 48 for additional photographs of the plastic.

Plastics for Structural Members

From time to time one reads or hears of a new "plastic" plane design. The squawking sort of term "plastic" in describing these planes has, in some instances, regrettably developed some wrong impressions. Right here the writer wishes to reiterate his previously expressed, and oft repeated statement that a more accurate designation of such planes would describe them as reinforced plastic planes (17), (18), (19), (20). The more recent developments in reinforced plastics as materials for aircraft structural members, such as fuselage, wings, and tail units, are attracting considerable interest and attention. Collectively, these pioneering developments are adding impetus to the trend toward molded reinforced planes and already several plane manufacturers are either in active production on such planes, or intensifying their experimental development work in this direction.

The Glenn L. Martin Co. has contributed a fellowship at the MAA Institute for research on the problem involved. Another group engaged in similar work is the Plastics Section of the U. S. Bureau of Standards, under the sponsorship of the NACA. It is believed that the technical data gathered by these two groups will be made available to the industry in complete.

In practically all of the developments in reinforced plastics, the plastic is subjected to structures, then wood veneers are used as the reinforcing agent for the synthetic resin, the latter being usually of the term and phenolic type. The reinforcing plastic is usually a resin, actually, are to be found in the resin and techniques employed in molding the members, this being caused by a matter of individual preference, and yet the fundamentals embodied in the reinforced processes are common to all. Fundamentally, the procedure embodies the coating of the veneers with a liquid resin or impregnating a resin film between the veneer strips, placing the resin treated veneers in a mold as form until the required thickness of lamination is obtained, means to hold the veneers in place under pressure, and subjecting the entire built-up assembly to heat until the resin forms an inflexible bond between the veneers.

The Damsell bomber illustrated in Fig. 7 is a good example of a completely molded structure without metal reinforcements of any nature. This fuselage is 19 ft long, 24 ft in diameter at the cabin end, and is molded in two

halves from wood veneers ranging from 1/8 in. to 1/2 in. in thickness with 800 in. phenolic type film as the veneer bonding media. The specific gravity of this structure is 0.52. In the Damsell process, the mold or form with the veneers and resin film is placed in a rapidly heated tank (Fig. 8 shows a 30x31 in. tank being installed) of special construction where the curing heat is applied and, though unique and expensive method, a uniformly distributed pressure is applied and maintained on all sections of the fuselage.

The Trim plane shown on page 40 is completely molded from resin coated wood veneers, both the nose and the plastic type being used as the bonding media. At the writing no details are available on the fabricating process or techniques employed by this manufacturer.

Molded aircraft structures present many advantages (11), (12), (13), (14), (15), (16), and present, one of no little importance being the possibility of laminating, without weight penalty, the designer's dream of a strong, durable and almost flawless aerodynamically perfect surface as opposed to the conventional riveted metal plane surface (14).

Two very new applications of reinforced plastics have recently been announced by the Glenn L. Martin Co., one being applicable to any type of plane, the other designed especially for combat planes. The laminated fuselage was developed for use on aerobics, elevators and radars. The primary object of this is to reduce the pilot of conventional aircraft effort in controlling them in their proper position, relative to the surface to which they are attached, during normal flight conditions. These tabs are adjustable from the pilot cockpit in any fixed position required to balance the plane for various flight conditions, such as climbing, cruising and gliding. The tabs shown in fabricated from laminated fabric impregnated with liquid phenolic resin. These tabs are structural lighter than their metal counterparts and, unlike metal tabs, there are no protruding rivets or lap joints. The entire laminated construction embodies both impregnated fiber which adds to the rigidity of the outer surface and provides the means for the attachment of the control mechanisms. Incidentally, this tab was awarded top honors in the Transportation Group of the 1940 Modern Plastics competition.

The bomb-bay doors of Martin bombers are unusually hollow shell structure resin bonded plywood covers and popular framing. They are 14 ft. long and weigh 41 lb. less the hardware.

(Turns to page 120)

It's the person with an indefinable "extra something" that wins admiration. It's the motor built to deliver extra hours of "life" that wins favor. Dumore motors are chosen to power business machines, household appliances, electric tools, etc., because, anywhere, are dynamically balanced to eliminate vibration; commutators are ground concentric with bearings for longer life; windings are expanded at high speed, then sealed to prevent "breathing"; leads are swept by special process to insure 100% electrical contact; each motor is inspected 5 times during manufacture. If you seek extra power hours for your product, write for Dumore Engineering help. There is no obligation.

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Decat on Plastics

(Continued from page 42)

in your telephone or your car panel. It would be so simple to build a press which in each molding would make a complete wing or landing. The answer is in the fact that, incident, it is molded in powdered form, requires a very high pressure, and a press to handle any such mold has to be very large. Models for any such press have been made, using heating coils for thousands of dollars, a prohibitive cost for the present small production level.

Present Developments

The Gerni L. Martin Co. set the pace in fabric laminated plastic with a table made by Taylor Fibre Co., composed of different laminations of fabric impregnated with phenol formaldehyde and formed in a hardened steel mold heated at about 300 deg. F. with a pressure of at least 2,000 lb. per sq. in. It is cured under pressure and heated for several minutes, depending on the thickness of the part to be polymerized. No finishing is required when it is removed from the mold. Some metal parts are also be molded while the press is in the mold.

The Gerni other company have not said that laminated plastic plastic were frequently but in the initial cost of the die and the fairly high cost of the laminating itself.

The outstanding properties of laminated plastic are its flame, sound resistance, its acid, chemical and water resistance. Although it has a great electrical insulating quality, this quality is not often advantage of the die construction.

Martin also met interested photo-laminated wood in a long bomb die cast made by Bendix Aviation. Though not much information is available as to their process, it seems to be in the same trend as those of other companies.

One manufacturer says an article mold which be achieved in a small press. The molding a wing, a jig is made for all the ribs. On this jig two or more laminations are laid with resin formaldehyde or polyamide in between. The ribs are placed in a rubber form which all at a mold. This bag acts as a clamp on top of every wood lamination and curves it to any shape the mold might have. The molds and ribs are then laid into after another on the same process. These are again placed in the bag for gluing and finally the stressed skin, which is made of

31/45 in. monolayer lamination, is laid top of the ribs and spars, which are placed in the rubber bag. Air is withdrawn and the bag is placed in an autoclave in which a pressure of 80 to 100 lb. is applied while the mold is raised to a temperature of about 200 deg. F., depending on the type of resin used.

Another process is based on an outside mold made of iron, the counter-mold being also a rubber bag. The glass-mold is placed formaldehyde which at first runs the only one that enclosed being case it was set. The mold is self-heating in the sense that pipes have been laid inside so as to be able to apply the heat, being, and the cure, only the placed which has set at around 200 deg. F. Though the iron mold was used for the Ford 48, in cost was so great that now it has been replaced by a stronger mold made either of wood or of a higher metallic material. The advantage of this process is that as the mold is outside, the press comes out with a perfect finish, whereas in the process in which the outer surface is in contact with the rubber, the inside comes out of the mold smooth but the outer surface has to be smoothed down in the latter process, thus in the advantage that once the mold is established, say rubber bag larger than the mold can take care of it, the only restriction being the pressure tank. In the former process, each mold requires its special rubber bag.

Texas Aircraft Corp. has released very little information regarding its plastic process. Texas apparently uses a made mold of the desired shape, lay on top of the different layers of glass and finally die have been cut in the desired shape and impregnated with resin. These are superimposed on top of another, covered and then surrounded in a frame made aluminum. The curing is done under dry heat in a big oven (Ford 48) which can hold a wing fuselage and all in the same time. The only drawback is the unsuitability of formaldehyde in that, contrary to other processes which have—depending on the kind of glue they use—a margin of several hours for the setting of their glass bag. The cure in the oven, at least an hour after applying the glue. The resin must be to about halfway between a self setting and a hot-setting resin, allowing the final curing to be done at a temperature under 200 deg. F. See Fig. 3, page 43.

Still another method used in the country consists of two layers of plywood with a layer of cork sandwiched between. The advantage of this process over the others is the total absence of spars at ribs allowing an entire stressed skin for wing or fuselage. To reinforce the cork a fine wire mesh is laid in the middle. The mold is made

of a thin metal skin stretched on the airplane principle, with heating coils, coming in contact. The counter-mold is made of a rubber bag of the shape of the inside. The whole wing is made at one time of a single piece including a finished edge resembling a finished de-lyde. The wing comes out of the mold highly polished and needs only to be dished at the trailing edge. A three-quarter wing is used on both sides. So far, only a small wing part 3 ft. long has been produced. It was able to withstand the weight of the men.

The Airtel part, of which little has been heard, is to show a better tool, though the plastic used is of the cellulose nitrate type. The glass is of good concentration and the setting is done at room temperature. A frame rather than a mold is used, and the wings of wood are interwoven.

The same kind of construction is used as the Messerschmitt made by the Piccolet Aircraft Corp.

Finally, the manufacturers of sail planes, in which construction, taking advantage of a Lockheed development made some years ago, fuselage starts with a concrete mold made in the shape of half a fuselage. For the sub-plate, all layers are deposited in proper order and will consist with canvas glass. A rubber bag is placed on top, the mold closed with a wooden frame and the resin superimposed into shape.

Other experiments have been conducted in the mold-setting type of resin. The molds primarily were made of wood, but Catalin Corp.'s plastic resin mold is not. The different layers of cork and finally die have been made. Between the layers of wood there is spread a self-setting phenol glue, and in accordance with setting the mold and glue is raised tightly up to about 180 deg. F. in a kind of box in which the moisture content is thoroughly controlled so that the wood does not "lose its life", i.e., lose its moisture and crack.

All these processes deal with wood superimposed or not set under heat. However, various developments have been suggested for the use of superimposed, compressed, improved wood, compressed, etc. In other words, the wood is given not only a chemical treatment but is mechanically treated as well. The process varies from 200 lb. in 3,000 lb. per sq. in., which pressure is governed so as to avoid splitting the wood under pressure or springback once the pressure is relieved. The wood which has been thus compressed and chemically treated results in a material that is less or than water with a density over 1.00.

Under the "Solignum process" a (Turn to page 128)

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North Atlantic Ferry

(Continued from page 12)

interviewing. In Montreal I was greeted by Capt. A. S. Woodcock, well known for his work with the Mayo composite plane, who was in charge of the Air Division of the Canadian Pacific Ry. Co. He said that it was the second American to volunteer, the first having been a young pilot who had gained most of his flying experience in Central America. Others to join our little group included: Hal Rogers, Tommy Sawyer, Earl Gorman, and many more.

Although I had joined up in August, it was late September before we got our preliminary flying instruction assigned. Two Lockheed Hudsons arrived, all lined up in close places to test the instrument flying abilities of the various volunteers. These tests were under the supervision of D. C. T. Bennett, flying expert of the Air Division, who later led the first formation flight across. Those of us who passed the tests were given cockpits. Pilots received \$1,000 per month, co-pilots or first officers, \$800 per month. The crew was completed by a radio operator who was paid \$600 a month. I was appointed a captain and assigned a crew. Part of the duties of such crew was to conduct a thorough test flight program with each bomber prior to the actual ferry flight, carefully checking engine power and fuel consumption.

The Lockheed Hudson, a military version of the Lockheed Model 14 transport, is well known. As detailed in Canada they are already marked with RCAF insignia, and are outfitted for service over Europe. Powered each two Wright CR165-5-24A Cyclones the Hudsons have a normal gross weight rating of 18,900 lb. However, with additional gas load, extra equipment, and spare parts carried across they were overloaded to a gross of 21,000 lb. Total fuel capacity was about 1,250 gal. and 36 gal. of oil. Radio equipment was Bendix, one transceiver and two receivers, with long direction finder instrument equipment a standard and includes an automatic pilot. Coldest navigation equipment was provided for the last phase of the spectrum only. Actual navigation was done by dead reckoning, determining known and estimated winds along the course as determined from weather maps based on such information as could be obtained prior to each flight. Pan-American Airways flies an entire lot 36 mi. at 1,000 mi. when flying by dead reckoning. On my first ferry trip

I was off just about 45 mi. in 2,300.

In addition to standard equipment, the Hudson is fitted with a collapsible rubber lifeboat built into the cabin door in such a way that it is released automatically if the plane is forced down on the ocean. These lifeboats are inflated from air bottles and are equipped with emergency water and food rations, and with signal flares.

Goodrich de-icers are mounted on the leading edges of wings and control surfaces of the Hudson. In addition we received compass known as Kilmont was mounted over the drivers and all leading edges. Kilmont has a very definite time limitation, but up to five or six hours it will cause us to break away after burning. The combination of de-icers and Kilmont worked perfectly under all very cautious which I experienced, but I do not think this could be covered in under half hour conditions for longer than 5 or 6 hr.

Since only sufficient fuel capacity is provided in the Hudson to allow a comfortable margin for the flight, it is quite important that consumption be closely controlled. Use of power was limited to the minimum efficient horsepower per engine on the actual ferry trip. This was calculated in advance to fractions of a horsepower, varying with diminishing fuel load, to conserve engine fuel consumption. Each plane was put through a thorough flight test program in Canada, under simulated trans-Atlantic fuel conditions, to determine actual fuel consumption accurately. I made one test run of about 900 mi., at an indicated air speed of 130 m.p.h., at altitudes of 21,000 ft., and established a fuel consumption of 47 gal. per hr. Our second test run I climbed to 9,000 ft. with a crew of three in the plane and established an average consumption of 46 gal. per hour for a flight lasting 4 hr. and 2 mts. Indicated air speed of level flight cruising was 137.5 m.p.h. and true air speed 137 m.p.h. (actual) mph. Later I found that the actual fuel consumption of the first ferry trip was at the rate of 73.5 gal. per hr., with a land-to-land ground speed of 201 statistical mph. The high speed of the trip, considering the low power at which the engines were operating, proved partially from diverging winds and also due to the altitude of 16-18,000 ft. at which most of the trip was flown. Kilmont-blow ground speed was only 145 statistical m.p.h., due to a deflag of about an hour after making my last land in the British Isles. This deflag was chiefly caused by local storm conditions which made it difficult for us to orient ourselves by the dead approach to our assigned terminal field.

After completion of flight tests in the vicinity of Montreal, I was ordered to proceed to Newfoundland and meet

start of the first Atlantic delivery flight. The trip from Montreal to Newfoundland, about 900 mi., was made on-stop and by dead reckoning navigation to simulate trans-oceanic conditions. We carried a full load of fuel and made the trip without incident at night. These planes made the flight in formation, arriving in good order at a harbor but well equipped base near the Newfoundland coast.

About two weeks of waiting we received word from Montreal on the morning of Nov. 10 that Bennett, our flight superintendent, would arrive in the early afternoon to lead our formation across to England that next evening. We were told our entire crew of seven planes on the base and ready for the trip. The rest of the day was spent in a casual check of all equipment. Instruments and life preservers were carefully studied. Emergency return were inspected and stowed away, consisting of several gallons of water, half a dozen cans of food and half a dozen cans of chocolate. The emergency lifeboats previously described were examined, along with Very pistols and signal flares. But no fire arms were given out as we were not supposed to fight, even if we ran into trouble. Since we were obvious pilots we were no sailors at any time.

Bennett landed in the early afternoon and called a conference of the flight captains. He had complete weather reports from stations in Canada, Ireland and England, and the weather bureau in our Newfoundland base prepared a detailed weather map and forecast at conditions along the proposed route. Each captain was provided with a small folder carrying a short copy of this weather map and the report of known and estimated wind conditions. From this information Bennett prepared a flight plan which covered every conceivable contingency. We were to fly a ground-to-land course. If we entered unfavorable flying conditions we were to break formation according to a pre-arranged plan and continue on our own. Up to a pre-determined point, about 44 hr. out, each Captain was to decide whether to continue the trip or turn back, depending on conditions. After passing that point we were required to continue in any case. When he had completed his explanation of our flight procedure none of the captains had other suggestions or questions to offer.

We were ready for the take-off. We were a pre-occupied group at dinner, 21 of us scheduled to undertake the experiment which would determine whether or not American bombers were to be delivered thereafter by moon formation flights. There were four American crews, and four British. The British

(Continued on page 12)

STREAMLINED POWER



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North Atlantic Ferry

(Continued from page 121)

when around noon at dusk, but I think most of the Americans shared my own feeling of tension. I felt very much "on the spot." These British people had been in to do a job at delivering their much needed supplies for them. We were the "fun-boys" boys on whom they counted to deliver the goods. There wasn't much time or inclination to think of ourselves as "surrendering the north of Britain," but I do know that I felt a responsibility to uphold the honor and traditions of American pilots in general. Already there had been quite a few American volunteer pilots (before after a flight test punctured their sanguineous dream of flying experience. And there had been some grumbling among the Canadians about these American pilots with poorly honed flying skills.

At a half before 7 p.m. I collected my crew and climbed into the plane. Mine was number six in the seven-ship formation, last on the right leg of the Vee formation. On the dot a 730 Bonnat moved down the runway in the lead plane, his lights reflecting back from the snow piled along the runway. His plane skirted slowly after me, probably for length of a wing, following runway, and disappeared into the night. The other four planes ahead of me took off in rapid succession and soon it was we four. There was nothing unusual about the takeoff and, as visibility was good, we easily located the running lights of the other planes and were soon in position in this formation.

We reached an altitude of 9,000 ft., which we held for the first leg of the flight in order to gain the benefit of favorable winds, and also to top a thick cloud layer which hid the Atlantic as far as we could see. The moon rose and shimmered across the sea of clouds. The air was calm and clear and our plane behaved perfectly. We all followed the leader and let him do the navigating. It was an anxious wait for the flight.

Although we had interplane radio communications and were permitted to use it at this stage of the flight, there was little occasion for its use. We had agreed in advance on a plan to proceed if weather conditions forced us to break formation. Each captain was to turn off course a prescribed distance and then proceed to his destination on his own. A radio plan had been established in advance, at about 45 hr out, beyond which all planes were

to proceed in England. As mentioned earlier, if trouble developed prior to reaching that point the plan to trouble was to return to the island field.

After a few hours of comparatively good flying weather we found the cloud formations extending up into our path and we were soon in an active storm area. This storm had been charted on our flight map and we climbed to 8,000 ft. in an attempt to get over it, but without success. Shortly we all plunged into rain and black storm clouds. We began bouncing around in lively fashion and I decided at once to turn off course. We flew due South for 20 min., then turned back on course, following a line of the storm for the distance of the original flight plan. Within a few minutes after breaking formation we began to pick up a lead of us. Already I was losing the effect of the altitude and my first officer, as mentioned earlier, was completely incapacitated by now. To get out of the ice I had climbed rapidly to 10,000 ft. and we soon were in the worst of it, though the storm contained scattered it was unusually more inside the clouds, but the outside temperature had dropped to -23 deg. C. Our oxygen equipment was not reliable and, what with the storm and darkness, I did not try going any higher, but held level at 10,000 ft. for the remainder of the flight until time to let down. Our weather map had indicated that we would be in the storm area for about an hour but it actually extended about all the rest of the way across. Several low pressure areas had apparently crossed by together into one big disturbance and we drifted along through the "meat" at 10,000 ft. for hour after hour. Presently one of the engines began to cough up and drop a few revs. Probably it was using too much as it spun itself down to work again, and did not bother us further.

At this stage of the flight my mind, like the engine, was dropping a few notches. I supplemented my brain by drawing on the oxygen bottle and managed to keep the engine from being. Worst scored me most was the behavior of the automatic pilot. The wing would dip in the rough air and then come up very slowly. I studied the instruments for some clue to the performance, tried to trust my own senses against the automatic pilot, and noted that the oil pressure for the automatic pilot was low. I reached over and managed to keep the oil pressure normally for about an hour. As I diagnosed the pilot the hydraulic pressure gauge, which should have read 100, was hovering around 55. Since the servo motor is operated by the hydraulic pressure it was obvious that the low pressure was responsible for the sluggishness of the controls. Doubt

less due to faulty maintenance some hydraulic matter in the hydraulic system had been causing valve leakage and consequent reduction of pressure. Permitting the hydraulic system to idle for a period seemed to solve the difficulty for after I re-engaged the control there was no more trouble from that source. After carefully rechecking my sensors and allowing for estimated wind I reset my time of arrival, moving it up by 30 min., and when I let from my destination I began to let down. It was still nasty weather and Mark as fast but at about 11,000 ft. we broke out between two layers of clouds and could see faint, dimly lighted islands ahead of us. We continued down and at 8,000 ft. the scene was visible below us. I continued at this altitude for half an hour, keeping close to the cloud layer above me just in case those imaginary Munsterbros should materialize. At 0800 Greenwich Mean Time I spotted a dark shadow on the water which soon turned out to be an island. We had made landfall 15 min. ahead of schedule. We viewed the island in an attempt to identify it and fix our position on the map. At this point a plane came out of the storm and after a few anxious moments I identified it as the ship's ship, which later proved correct, as he reported making his landfall at about the same time and place as I had, but before I could intercept how he had plunged into another cloud bank and was lost.

The area we had at land was equally making visibility bad. I finally dropped down to an altitude of 6,000 ft. but it was still impossible to identify our position. At this point I decided to go on wing tanks run dry, and, as I switched over, I noticed that the master fuel gauge for all wing tanks was not functioning at all. There was another 100 gal in the landing tanks, but here once again I could not be sure. With that beginning to run low I decided to use emergency procedure and ask for a radio bearing, although we were under strict orders not to use the frequency.

My radio operator soon got a bearing but it didn't fit my idea of our location and I asked for a confirmation. But when my operator tried to reach the station again they refused to answer. After circling the island for 45 min., with fuel running dangerously low, I authorized the sending of a "priority" message, which demands immediate attention. This got on a paper bearing in about 10 min. and we flew on to our destination without further delay. However, I heard a great deal about this priority message later to a friend as he spent a portion of his time in the air. Apparently they differed considerably from my own estimate of the

(Turn to page 126)

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SAE Papers

(Continued from page 120)

short, is the principal job that the air-scoop has to do.

It is recommended that in the design of a practical air-scoop, the first line laid should be straight lines from the top of the collector flange and at right angles to it.

The elbow or turn in the scoop is usually the one feature which causes the maximum amount of difficulty. The most of the trouble is caused by the efforts which are put into providing a smooth turn. In such cases the air entering the scoop simply cannot pass through the scoop in high velocity without piling up along the outer radius.

The simplest turn is definitely a desirable feature, provided that the cross-sectional area through the turn is not restricted by the use of a large radius for the outside of the turn.

In a smooth radius elbow the air is continuously thrown against the outer radius by virtue of its centrifugal force, with the result that there is no upper limit for the flow to become equalized across the area of the passage.

In the case of most under-cowl types of scoops, the normal space limitations call for a design of passage which is rather low in the vertical dimension and rather broad in the horizontal dimension.

A duct with a high aspect ratio is definitely recommended, and it can be continued in this form as close to the actual elbow as possible. In the case of the high aspect ratio elbow, the velocity along the inner and outer walls of the scoop differs by a relatively smaller amount than it does in the case of the low aspect ratio.

In most scoop designs there is a desirable limitation in overall height, and here again the high aspect ratio design is advantageous, in that it permits a greater straight length of duct at the collector for a given area and given total height.

A third advantage of high aspect ratio is that the bends on the lower velocity are reduced.

The function of the lower valve is to provide for the admission of hot air to the scoop in quantities ranging from 10 to 100 percent of the total. In many designs of scoop where the lower valve is immediately above the collector entrance, it is a justified impossibility to obtain good temperature and pressure distribution at the collector, due to the extremely short distance allowed for the air to move toward and its pressure equalized.

By putting the valve around the collector

and ahead of the elbow, the technique preferred in the valve is used to assist in thorough mixing of the hot and cold air streams. Since the valve is at a maximum distance from the collector entrance, the pressure disturbance caused by variation in the valve position becomes a minimum.

In some designs of scoop that have been tested, it has been found that the upper portion of the power loss experienced when going from full-cowl to half-cowl was due to restricted hot air passages in the scoop, and that the major changes in fuel-air ratio in going from full-cowl to half-cowl were due to pressure disturbances caused by the lower valve itself.

In the case of scoops which project above the cowl line, the shape of the scoop is usually pretty well established after the foregoing design considerations are followed. A certain amount of area is required to provide proper burning capacity for the engine and

the area of the mouth of the scoop should normally be less within 75 to 100 percent of the area of the collector entrance.

In the case of under-cowl types of scoops, the attention is somewhat more complicated due to the restricted length of scoop and the necessity for turning several corners in going over the cylinders and coming out under the lip of the cowl. Here again it is felt that the maximum cross-sectional area of the horizontal passage of the scoop should be from 75 to 100 percent of the area of the collector entrance. In most types of under-cowl scoops a fair amount of area is available at the mouth of the scoop to provide a good, smooth approach. Full advantages should be taken of this space in this point. The final vertical portion of the scoop leading to the collector entrance should be especially watched to prevent any rapidly diverging area at this point.

Most of the material in this paper may be summarized in the following five rules:

RULE I—Provide the maximum length of straight section of scoop immediately ahead of the collector.

RULE II—Use a sharp right angle elbow—not a smooth turn.

RULE III—Maintain a high aspect ratio in the wing passages and elbow, wherever possible.

RULE IV—Keep the lower valve as far away from the collector as possible and locate it on the far side of the scoop elbow.

RULE V—Maintain at least 75 percent of the area of the collector entrance throughout the scoop and also at the lower valve. Avoid rapidly diverging sections, particularly just ahead of the collector entrance.

Using Problems Attached To The Operation Of Transport Aircraft R. L. MILLER United Air Lines Transport Corp.

The ice formations adhering to the component parts of an airplane may be grouped into major classes: (I) those producing a loss of flight performance, and (II) those which serve as an annoyance to the operator or passenger without seriously affecting the flight characteristics inherent to the design of the aircraft or its engine.

Removal of ice forming on wings . . . is not always assured by merely starting de-icing operation in the long range is arrived and stopping that operation after emergency. Experience has shown that such a procedure often aggravates ice roughness along the wing without removing any appreciable amount. The procedure adopted . . . has been to purposely avoid operating the deicers until an ice thickness of approximately $\frac{1}{8}$ in. is shown. They are then operated for a short period to sufficiently crack loose this thicker formation so that it is blown from the wing in turbid clouds. The deicers are then turned off until the ice thickness again approaches $\frac{1}{8}$ in. Treatment is repeated in this manner as required.

Rubber inflation deicers bring up other points that bear mentioning. After they rubber bags are taken on a slight pressure . . . they lose some of the rubber qualities originally designed. Eventually "ballooning" develops. Ballooning is a lifting of the deicer from the wing surface which makes the part unstable so the deicer inflates, though some air pressure is not present in the tubes. Ballooning of test tubes in low pressure areas has only partially relieved the likelihood of ballooning.

The obstruction of the pilot's forward vision by ice accumulations on airframe windshields is probably the most frequently encountered emergency situation. The most common method now in use for anti-icing windshields is the use of a heated air blast on the inside of the glass. The system is fairly effective in light and moderate icing conditions. In glass . . . and for rain freezing at low temperatures, however, it is of no benefit.

Deicing fluids (alcohol, glycerine, deicing glycol, etc.) have been sprayed against the outside of the glass. More have proven satisfactory in themselves.

In conclusion it can be fairly stated that much progress has been achieved during the past few years in the improvement of all aircraft anti-icing equipment. However, much still needs to be accomplished before air transport operations will be able to operate without regard to possible icing conditions.

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Ready-Made Airports

(Continued from page 23)

hours in locating proper fuel, since the problem was a usually lengthy process. In spite of these obstacles it is a noticeable fact that airplane pilots represent the most ardent group of enthusiasts in the flying fraternity.

The whole subject of seaplane bases has, in the past, been rudimentarily regulated by most municipal, state, and federal airport construction agencies. As a result, the Seaplane Flying Association and individuals have for several years lobbied the Civil Aeronautics Administration with requests to alleviate the deplorable conditions. It was their contention that the federal government should participate in the construction, improvement, development and maintenance of a national system of small seaplane bases, which by definition of the Civil Aeronautics Act (Section 1, Paragraphs 6 and 22) are included in the designation "Airports." It was their further opinion that the mismanagement by the federal government of such activities would be more, per dollar expended, than any other comparable program for the general development of private flying and would be of the greatest importance to economic and the national defense.

It remained then for the director and owner of Major A. H. McElroy, Chief of the Airport Division of the C.A.A., to take the initial step. A seaplane unit was formed two years ago, with the writer assigned to the task of "surveying the watersheds" in all the states where such existed, which subsequently proved to be 31 of 48 of them.



Flying a seaplane from a seaplane base. The idea appeals to any pilot who loves the water. BTA camp at the left.

with Alaska is the offer for good reasons.

Together with the very able assistance of Major McElroy and his assistant W. G. Stewart, a comprehensive coast-wide program for the construction and maintenance of seaplane bases was drawn up. The first phase of this program was to conduct a comprehensive field, as well as office, survey and analysis of the entire subject of private and commercial seaplane flying in the United States with respect to the need for seaplane facilities. It was decided that the survey should cover among other points the following:

1. A study of the U. S. by means of large scale maps, supplemented by a field survey and the personal knowledge of experienced seaplane pilots, in order to determine and outline all suitable water flying areas.

2. An analysis of these areas in terms of their relation to adjacent water areas, their population and buying power, and their present landplane and seaplane activity.

3. A field survey of all areas where private seaplane flying is now being practiced or has been shown to be practical and desirable in order to compile a list of such seaplane bases as are now in existence, note their limitations and the reasons for their existence and determine the need for added facilities.

From the very beginning the survey revealed a tremendous amount of interest in water flying, the interest apparently having remained latent due mainly to a general lack of understanding of its basic requirements. The need of an educational program became apparent and reformative, but it is now being disseminated by means of magazine articles, bulletins, radio

broadcasts, addresses before civic organizations, service clubs and personal interviews with operators, airport managers, inspectors, city officials and others. Complete working drawings and specifications were drawn up covering hangars, ramps, mooring railways, docks and other accessories for the efficient handling of all types of seaplane craft and made available to anyone interested at no cost. Engineering and advisory service are also available to cities as well as individuals by appointment to the Airport Division of the C.A.A.

A Directory of "Seaplane Bases and Anchorages" has been issued, a sort of "Blue Book" compiling cross-country seaplane data in its plan and chart their location in relation.

Comprehensive written reports, together with aerial photographs of all water areas surveyed, are being filed for reference purposes and a great deal of valuable data has already been compiled which will enable the Seaplane Unit to render assistance and advice as well as routes and available landing areas throughout the country.

The most important and needed step taken was for the immediate construction and installation of especially designed landing fairs in which airplanes could make land for the discharge of passengers and refueling.

Rapidly passages have been made over a period of years to interest various communities in the promotion of these highly desirable facilities. Due to lack of interest and funds the results were disappointing. The Civil Aeronautics Administration was however fortunate in securing the whole-hearted and enthusiastic cooperation of the National Youth Administration for the aerial construction and maintenance of small seaplane bases on a nation wide scale. Here was an extremely useful and constructive project for teaching our youth carpentry, machine work, the building of blueprints etc., and for service that it was connected with aviation gave it a real and appeal which immediately captured the workers' enthusiasm.

So much has already appeared in print about this program that I will touch upon it but briefly. The success of the program has been quite gratifying and to date, I believe, to its simplicity and economic features. Suffice it to say that the National Youth Administration will construct and install landing fairs, moorings, hangars, etc., for any community which will provide the necessary materials. The landing fairs measuring 10 by 22 feet are supported by steel beams, usually of burlap, the cost averaging about \$100 to \$200 per unit. Several of these units

(Turn to page 144)

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5 Gun Turret

(Continued from page 47)

rise of a second of time available, and that it is all over. Either the target will have been hit by a few bullets, or he will have finished off the target, as touched, all within a few seconds of time.

The five-gun synchronously adjustable turret shown in Figure 1 and 2 is designed to provide accurate division of loads between the five guns constituting the battery with instantaneous adjustment of each gun within a predetermined sector of operation.

All five guns of the battery, which are light 30 caliber machine guns, are mounted upon a common cradle plate or pivot point. This plate or point is part of a mobile mounting, in this case the entire turret, which can be rotated through 360 degrees. The gun mount within the turret may be raised or lowered and the guns themselves elevated or depressed as well, all of these operations being accomplished by hydraulic means.

The individual guns are mounted on gun cradles on special adjusters upon which they can recoil freely. These cradles or adjusters are fixed to the common base of the battery on a universal pivot point.

The entire gun is fixed to the cradle plate or pivot point, and it is with this gun or rather, with the

aligned portion of the gun, that the rotating and aiming is done. First A, B, C and D, however, are flexible, being actually five guns within the limitations imposed upon them by the cable control system designed for use with five battery.

By synchronously we mean the mechanical coupling of the diagonally opposed guns A-C and B-D (See Fig. 2) that bear control the movement of guns A-C and the other lever those of guns B-D. These two units are therefore interdependent of each other. The gunner controls both levers and thereby the complete movement of the four main guns, with respect to sector gun.

The control system has been so designed that at a range of 1000 yards which is actually beyond the range at which combat usually takes place, each of the four main guns has a maximum radial movement of 50 yards. (See Figure 3.) When operating the battery, the controls are left as manual, all five guns will fire at one point. When under adjustment, however, each of the four outer guns A, B, C and D can be controlled through, and extended within a 50 deg. sector of a circle, the size of which sector or circle being determined by the range at which the target is located, and the degree of manual adjustment required.

It is evident, that under many conditions of aerial gunnery, this adjustable turret will be necessary, what industry has been impossible with a fixed group of guns firing at one point. According to the movements of the target, the

entire turret must, and it will display the same three guns of the battery in the same favorable position for answering some hits. These positions will vary accordingly as the target is moving at an angle away from the defending plane, or flying at some particular angle. An accurate range of positions can be taken up by the four guns under adjustment. The absolute ideal would be to have each gun individually adjustable by the gunner, so that every possible sort of curve along the five guns at a burst could be placed within the circle which comprises the maximum spread of the guns. At present when gun A is moved anywhere within its sector of operation, gun C follows the same movement inversely in its own sector.

The particular point of interest in this system of fire-guns control is that under any condition of adjustment required by the gunner, the division of loads between the guns would be accurately spaced from center. We see that the guns comprising the battery can be deployed diagonally horizontally or vertically, as that five lines of the gun can be swept across or along a target, depending upon conditions. And furthermore, the main guns can be placed at will at any point of the sector in which they operate, with the exception that diagonally opposed guns follow each other inversely.

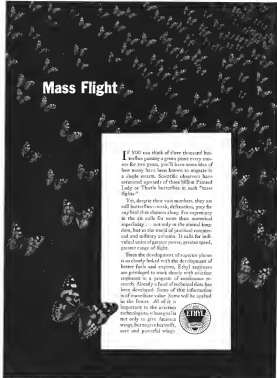
Figure 4 shows the original concept of this multiple turret, which allows only for spreading the four outer guns at a 45 deg. angle to the fixed inner gun E. This was a controlled dispersion system. When we add two-dimensional adjustment to the battery, it can no longer be called a dispersion device, as dispersion does not enter into the picture at all. In fact, dispersion as such is undesirable, as we have a heightened spacing of the bodies of each gun. With the system just described the pattern of fire from each gun can be accurately controlled from a single construction of fire of all five guns to a variable pattern, which is restricted only by the design of the controls.

The control system must be designed to build certain conditions. Paramount among these are:

- 1 Control by one gunner
- 2 Instantaneous control
- 3 Smooth control throughout the field of adjustment

A cable control system such as is applied to the present battery appears to meet closely comply with these conditions, and at the same time offer a single mechanical apparatus. An alternative would be a system of links and levers, which might be cumbersome and slower in action.

To obtain the maximum radial movement of 50 yards at a range of 1000



Mass Flight

IF YOU can think of three thousand bats in the passing a given point every minute for two years, you'll have some idea of how many have been known to migrate in a single season. Scientific observers have estimated upwards of three billion Pileated Lady or Thistle butterflies in such "mass flights."

Yes, despite their vast numbers, they are still butterflies—weak, defenseless, prey for any bird that chances along. For supremacy in the air calls for more than numerical superiority—... not only in the animal kingdom, but in the world of practical commercial and military aviation. It calls for individual units of greater power, greater speed, greater range of flight.

Then the development of supersonic planes so closely linked with the development of better fuels and engines. Ethyl engineers are privileged to work closely with aviation engineers in a program of continuous research. Already a fund of technical data has been developed. Some of this information is of immediate value. Some will be applied in the future. All of it is important to the aviation technologists whose goal is not only to give America wings, but to give her swift, sure and powerful wings.



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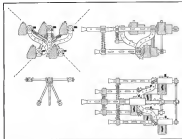


Fig. 4. The five-gun turret shown based on radial-controlled dispersion system. Adjustments made by means of a single control lever, with manual help. ETHYL.

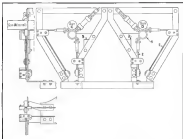


Fig. 6 Detail of entire control mechanism for lower guns "C" and "D". 1, Return or centering spring; 2, control contact; 3, switch for changing guns ahead; 4, gun act at gun ready in which control enables gun attached. The gun lock is here in ready on the ready or release.

yard, each of the four outer guns has a radial extension of 2 deg. 32 min at the gun muzzle. The actual movement at the muzzle is therefore less than an inch. The angle through which the central cables of each gun moves is of course much greater, so that the leading effect of the return or centering spring is not too much at the previous extension of the gun. Vibration of the gun while firing and the jarring action which holds up and causes the muzzle to have a tendency to elevate might prove to have too much counter effect to the leading action of the return spring, and the spring may be replaced with a third cable, which would definitely provide steady control at the gun throughout their spreading range, but would not provide for the automatic return of the gun in normal or emergency firing position. The former would then have to do this manually, which may not be a serious objection.

The mounting arrangement of the five guns with respect to each other may be in any manner compatible with the small space available in a turret. The newest choice in the figure is a mechanically fixed, inclined, provides level view for the gunner and the control system, and allows sufficient room for quick replacement of ammunition magazines, or cartridge belt feeds. For fire and hit target positions other arrangements of the gun will prove desirable.

The effects of a variable centrifugal force pattern out of a multi-gun turret mount can only be shown in actual mod-

els. Besides spreading a number of lines of fire assembly, it will undoubtedly discover attacking planes, the paths of which will soon become aware of the fact they find themselves in a constantly changing cone or nose of fire, instead of in one fixed cone, out of which they can easily outmaneuver. Coupled with increased aiming effectiveness at the multi-gun turret, the psychological advantages involved are of the importance of an arsenal.

It is necessary to note that the variable fire pattern control theory does not require five guns as a basis, four or even three guns may be mounted for adjustment under the same system.

About the Author

Mr. Braden's connection with aviation began in 1920 when he was a radio engineer with Universal Air Lines, before he was taken over by United. He has spent the past decade in Europe as a correspondent and aviation writer. In 1930, he became interested in aircraft and anti-aircraft armament, and currently he is engaged in armament engineering for one of the large eastern aircraft manufacturers.

Aviation in Transition

(Continued from page 191)

as far away as the aircraft builders. This was demonstrated in the case of the Glenn L. Martin Company. The

company reported preliminary earnings for 1940 of \$3.92 a share. This was prior to the enactment of the tax amendments. The revised report now shows that the company was able to secure additional earnings of \$1,115,000 in addition last year's results to \$4.94 a share. Revised earnings for the major companies are shown in Table I.

While the aircraft group was held hit by income taxes last year, they nevertheless showed strong progress over 1939. For example, about 60 percent of United Aircraft's net earnings went to the Treasury for normal and excess profits taxes. Sperry paid more than 50 percent of its earnings for tax.

Yet despite these heavy capital, both companies showed veritable profit increases over the previous year.

Among the most important measures of managerial efficiency is the profit margin reported on the year's business. In view of the severe nature of taxes, Table I presents 1940 profit margins for the major aircraft builders, before and after taxes. An examination of the final results reveals some very interesting facts pertaining to the relative position of these companies. The last change, before taxes, was made by Martin, North American and United Aircraft. Earnings after taxes, however, were best received by North American (Douglas and Martin). In these and other cases, the relative high profit margins may be attributed to the preponderance of deliveries on foreign orders. While an important amount of foreign business is yet to be delivered—these high profit margin orders are rapidly being expanded. The bulk of billings for this year promise to be of a high profit nature as they are for the second of the government.

In reading summarized results for the year, it is interesting to note that the major aircraft manufacturers are to be found in the list of the top 100 industrial corporations in the United States. For example, North American Aviation, Inc. reported net earnings of \$2.06 per share for 1940—the same as for 1939. On the contrary, this can be interpreted as reflecting lack of progress on the part of the company. However, upon closer analysis, we find the company, after showing 36 cents a share for the first nine months of 1940, started \$1.05 a share in the final quarter. This would indicate earnings rising at the annual rate of at least \$4.46 per share. We note further that in December 1940, the company delivered approximately twice the number of planes shipped during the first months of 1939. Should any doubt remain as to North American's sharply mounting rate of plant delivery, we refer to President Knudsen's statement commencing on December production schedule, as "even this record will not take into account."



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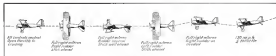


Fig. 4 The stall roll.

ards of inverted flight (though the first one is about twice longer) to pick up the maneuver again, the rule of thumb is to wait until the ship is inverted. When 90 mph has been attained, you are ready to roll right side up. Let's assume that the first roll is to be made to the right. Keeping the stick in the same fore and aft position, move it as far to the right as it will go at the same time applying enough right rudder to maintain direction. As the ship rolls through the position of a vertical bank, additional right rudder will be needed to keep the nose from dropping; then as the wings approach the level position, the rudder is gradually eased off and the stick brought back to the center but still kept somewhat ahead of neutral as to maintain the dive.

At this point, the engine, which has probably stopped, will start running again and the winged should be about 120 mph. Maintain the dive until the proper speed for the loop has been acquired and correct the maneuver due to any drift to the left. As many more loops may be performed as desired but it will be noted that with the engine running there some altitude is lost in each loop. Don't let your reluctance to stop with the engine to the extent that you'll consume gas too before a safe altitude. This maneuver should be practiced long and continuously until it can be performed perfectly, for it will make things much simpler in other maneuvers further on.

There are several series which you are likely to make at first. One of these is attempting to roll too soon due to your natural reluctance to push the stick ahead when you are going down. If you give way to the tendency you will make every attempt to roll and will probably cease on pointing in some direction other than straight down the road. In fact you may have difficulty in rolling at all, instead of making a free turn you will simply follow from a right-side up position. Another error is making too long to start the roll. This will bring you out in a probably a vertical climb, from which you will have trouble making the second half of the loop. Suppose one of the rudder will prevent your maintaining direction; using the side-

roll, too delicate, is to swing back with the stick when the wings are inverted will have the same effect, but since you have learned the proper coordination and timing, it'll guarantee that you will like the latter style.

The Slow Roll

This maneuver, illustrated in Fig. 4, is considered by many to be extremely difficult. Actually it is not difficult at all. If you become proficient in the Cobra roll, the slow roll will give you little trouble in any airplane of reasonable power and controllability. When performed properly, with the nose held dead on the horizon, it is extremely graceful and spectacular. In it, probably more than in any other maneuver—smoothness, delicacy in control operation, and relaxation of the pilot should be emphasized. If there is a comment here or said elsewhere rather with much it is recommended that you get a firm grip on it before the roll is begun, particularly if the ship is one which requires extreme forward movement of the stick to hold the nose up while inverted. This is so far corresponds to "pulling bottom," or raising the hold of the saddle while riding a horse, or it is an indication of nervousness. It simply serves to keep you from falling to one side of the cockpit when the wings are inverted, and to enable you to pull yourself ahead if this is necessary to get the stick far enough forward.

The slow roll is begun by setting the stabilizer to a slightly nose-high position and during the ship in a rather shallow climb at a speed of about 120 mph has been attained. The throttle is eased back during the dive so as to keep the wings within the proper limits. This means that eventually the throttle will be only about one-quarter open. When the winged indicator shows the proper reading, let go the throttle and a you think a desirable, get hold of the brace previously mentioned. Then the stick back slowly and the nose comes just about the horizon. Then move the stick smoothly off the map to one side. For the sake of simplicity of explanation, let's make it the right side. At the same time apply enough right rudder to offset the adverse yaw and maintain a straight course. The course should be determined by some object

on the horizon, preferably a cloud, which will remain visible throughout the maneuver. As the bank approaches 45 deg. it will be necessary to move the stick somewhat ahead of neutral, so as to hold the ship on the course at the same time releasing the pressure on the right rudder pedal and applying slight pressure on the left pedal. The control act to push the stick too far ahead at this point or if you do not will push the nose off the mark. As the ship rolls into the position of the vertical bank, more and more left rudder will be needed to hold the nose up. Meanwhile, and throughout the maneuver, full throttle is maintained. As the ship rolls past the 90 deg. (or vertical bank) position, the stick must be moved farther ahead, replacing the rudder action in holding the nose up until when fully inverted the rudder is in neutral and the stick still ahead—in some cases all the way ahead. Once more however you are cautioned about too much forward stick while in the 90-deg. position, either while approaching the inverted position or during the second half of the roll.

When the ship is on its back the nose should be slightly above the horizon. As the full continuous right rudder should be moved on and the forward pressure on the stick gradually lessened. When the wings are again vertical, approximately full right rudder is being applied, full right aileron stick about neutral. As the wings approach level the right rudder is moved off, likewise the left and forward pressure of the stick so that all controls are in neutral just as the ship crosses neutral flight.

The throttle has remained in its closed position throughout the maneuver. The engine stopped (thereafter will be easy about one-quarter open again at level flight is resumed. However, since the throttle was partly closed it will not start with a bang but will take hold smoothly and evenly, once the engine has started in the roll is begun. In this case, the engine stops just the same while it is inverted and when the wings become level starts at full power and spins the smoothness of recovery, and the ease with which the throttle is closed at the completion of the roll.

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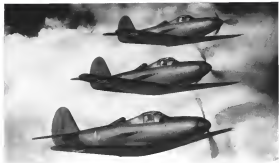
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AVIATION Dept. 342

AVIATION May 1944

Planning

(Continued from page 11)

A study which was made in 1937 for the procurement planning section of the Air Corps on how, in the event of an emergency, a rapid rate of expansion could be accomplished, led Hordern Standard to recommend, among other things, the installation of a new system of manufacturing and inspection, controlled as the system then in effect would not accomplish the purpose since it lacked the flexibility for utilizing available material and machine hours available through adaptable plants. The rapid rate of increase at the beginning of such an expansion period, the first of the four basic differences just mentioned, made this new system almost mandatory under such conditions. The change required in the new system was the replacement of Army inspection material, Navy inspection material, and commercial inspection material, together with three sets of records and controls, by one group of inspection material which would be acceptable for any use. (Points were cut from 60 to 20). The simplification on paper work allowed by this new system almost eliminated the necessity of any increase in personnel to handle adequately the increase in the ordering department as the principal factor in increasing the load as the increase in the number of designs and parts. It is obvious that this change greatly accelerated the rate of increasing capacity. With the same old machineage composition of the Army and Navy the change in the system was effected in 1938. (Fig. 2—Standard drawing specification altered by new control system) This change made possible —

- 1 Quicker initial delivery of government contracts
- 2 Flexibility of schedule
- 3 Availability of materials for Army and Navy requirements for all types in stock
- 4 Simplification in procurement procedure
- 5 Greater control of inspection stand and
- 6 Increased efficiency in the manufacture of properties

The second phase of planning resulting from the rapid rate of increase was that of eliminating, wherever possible, those matters and divisions which depended upon opinion. For example, the acceptance of major parts which are highly stressed, and made from alloy steel, is based on the magnetics method can

certainly allow personal opinion as to the usefulness, bearing, and use of the substance to destroy all endeavor at acceptance or rejection. The engineering department among the scientists of this particular division, recommended this problem and brought it down to departmental matters and plans determined the position and study of acceptance and rejection.

Prior to this time there were reports of from 25 to 60 percent, which made it impossible, after having accepted men and machine hours, still to maintain schedules. Today, under first new specifications, the rejection has been reduced to approximately 1 to 3 percent without impairment of quality.

Referring again to its procurement planning study made in 1937, it is stated that the three present personnel



Fig. 2



Fig. 4



Fig. 5

would provide the nucleus of the supervisory force in the event of an expansion. The career choice there is a comparatively short period for the peak. As a matter of fact, production has never yet reached that peak (Fig. 3—short showing, however, scheduled steps of accelerated production). This short clearly shows that no action is as obvious reached than it is well-known to a new endeavor. In fact, three times the new schedule took action before the previous one could be completed. Therefore, the peak is expected to be of short duration so that whatever additional men or equipment are to be used will be used only a comparatively short time.

A further result of the apparently short period of the peak, is the use, in certain instances, of second hand equipment. It is sometimes possible to obtain, at comparatively low cost and quick delivery, second hand equipment which will be adequate for short time use. Such equipment, of course, is not used for process work, or in the case of office equipment, for permanent installation.

The third difference is the change between the of expansion curves, so that the volume in the accelerated production studies a peak which was extremely high compared to the normal base. This difference makes it almost mandatory that new methods be found. Wherever possible, the method used, whether it was manufacturing, direct, or supervisory, was simplified. The simplification of manufacturing methods was accomplished, wherever possible, by the executive of order purposes, high production, accurate machine tools. By this means they reduced the requirement for flow space, less skilled labor, and for the number of machine tools. The success in volume allowed them to salvage their equipment in functional loss. Further development at the same time, led to transformation of emergency basis which again reduced the requirement for flow space, reduced the amount of materials handling labor, and simplified the job as supervision. This policy of simplification is not limited to manufacturing operations. It is extended to all the related departments. A request for an addition to overhead personnel is frequently resolved with a simplification of the function for which additional help was requested. For example, this expansion has obviously increased the load on the personnel test engineering, and ordering departments (instead of increasing these departments in the same ratio as the load, advantage was taken as the increase in volume to simplify the systems, and to process machine-made equipment. The result of this philosophy is shown on Fig. 4. (Chart



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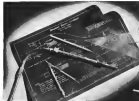
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The Seed and the Timber...

FEBRUARY 1906. Two men, fresh come from the frozen lake which has served them as a flying field, are thawing out by a roaring cabin stove. Their faces are flushed, partly by exertion, partly by the warmth of the room after the icy bite of the winter air outside. The younger of the two is scribbling excitedly in a diary, the words bubbling swiftly, unimpededly from his pen.

"Today we flew—**FLEW!** Yes, after all those weary months of work, we have done it. 825 feet in one straight flight! Even now I can hardly believe it....."

February 1941. Thirty-three years on, and the seed so laboriously planted on this icy February day out on the frozen lake is now broad, strong timber. On the tarmac outside the airport buildings

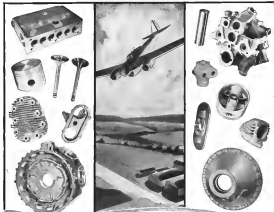
a great streamlined sky-rocket is waiting to take off. Time marches on. . . . Thirty-three years it is a short time in which to progress from perilous three-hundred yard hops to regular transcontinental flights, yet even now who can tell how immeasurably far we are from full efficiency in the air? In a hundred years time, maybe, they will be laughing at our efforts. . . . Yet in one thing [1941] sees us already possessed of maximum efficiency. **That thing is Filtration.** After long months of research, the VOKES laboratories have evolved filters, incorporating an entirely different principle of filtration

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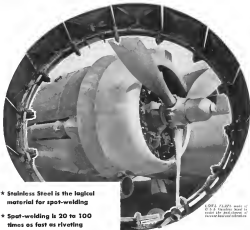
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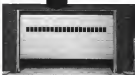
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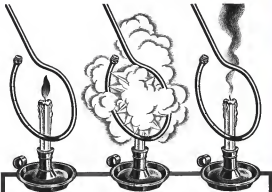
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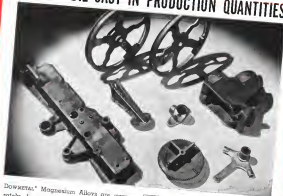


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